

UNIVERSITY OF CALGARY

Comparison of Assisted GPS and High Sensitivity GPS under Weak Signal
Conditions

by

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A THESIS

SUBMITTED TO THE FACULTY OF GRADUATE STUDIES
IN PARTIAL FULFILMENT OF THE REQUIREMENTS FOR THE
DEGREE OF MASTER OF SCIENCE

DEPARTMENT OF GEOMATICS ENGINEERING

CALGARY, ALBERTA

October, 2006

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ISBN: 978-0-494-26381-5

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ISBN: 978-0-494-26381-5

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**Comparison of Assisted GPS and High Sensitivity GPS
under Weak Signal Conditions**

(URL: <http://www.geomatics.ucalgary.ca/research/publications/GradTheses.html>)

by

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October 2006

SCHULICH
School of Engineering



Abstract

The Federal Communications Commission (FCC) E-911 mandate, Location-Based Services, as well as personal and vehicular navigation applications are among the main forces driving the need for a navigation capability in degraded signal environments such as in urban areas and indoors. Since the positioning accuracy produced by GPS methods is superior to that of other positioning technologies, most wireless carriers are exploring Assisted GPS (AGPS) as a potential strategy for meeting the FCC criteria. In this thesis the performance of AGPS was investigated under various field test conditions as measured through a range of acquisition and tracking tests. Tests were performed in several settings, including: a suburban environment, residential garage, speed-skating track and a steel-reinforced concrete basement. Distinct aiding scenarios were explored to investigate the effects of aiding data on AGPS signal acquisition. Limited simulation tests were also carried out to determine acquisition sensitivity and the effects of different aiding data on AGPS. Simulation tests showed that AGPS had higher sensitivity as compared to high sensitivity GPS (HSGPS) and a standard receiver; it also demonstrated the importance of satellite ephemeris data in terms of factors such as higher sensitivity (11 dB) and lower Time-To-First-Fix (TTFF). The field tests illustrated the limitations of HSGPS; its inability to acquire in weak signal conditions and longer TTFFs in comparison to AGPS. Acquisition tests using AGPS demonstrated the importance of accurate time and position assistance in obtaining a shorter TTFF under weak signal conditions which refer to test sites such as the speed-skating track and the concrete basement. Tracking tests showed similar results for AGPS and HSGPS receivers in terms of factors such as positioning accuracy and solution availability.

Acknowledgements

I would not be here if it not for the following people who have encouraged, supported and provided motivation enabling me to complete my thesis. I would like to thank:

- Dr Elizabeth Cannon and Dr. Richard Klukas for their continued financial support, encouragement and guidance which have been valuable and has enriched my knowledge in the field of GPS.
- My parents in Lautoka Fiji, my Kaka, Kaki (uncle and aunt in Hindi) from Red Deer AB, my cousins, Nikita, Krishen and Neelam, and my brother, Somjeet for their love, care, moral and financial support.
- SiRF Technology Inc. (www.sirf.com) for proving us with the AGPS receiver. I would like to thank Geoffery Cox and Lionel Garin (now at Nemerix) for answering many questions. Finally Greg Turetzky at SiRF is also acknowledged.
- Gerard Lachepelle is also acknowledged for his encouragement throughout the course of my program.
- Mr Ajay Varma, Binita Devi, Alvin, Amanda, Shyna for their moral support and continued encouragement.
- Dharshaka Karunanayake for his continued loyalty, friendship and encouragement throughout my research.
- Mr Merlin Keillor, Judy Smith and Pat Pardo from the Disability Resource Centre at the University of Calgary for providing assistive devices and the 42” plasma screen.

- My friends and colleagues who were or are in the PLAN group; Anastasia Salycheva, Chaminda Basnayike, Cecile, Bao Zhang, Gao, Glenn MacGougan, Oleg , Olivier Jullien, Rob Watson, Sameet Deshpande, Salman Syed, Saurabh, Seema Phalke, Shahin and Tao Hu, for their assistance and friendship
- My online friends either from www.indofiji.com or msn messenger for keeping me company and sane when I was stressed; or during those long lonely nights; Ranjit R. Raju, Shalini Chand (Miss Princess), Sanjana, Jotishna (Tishy), Agnes, Sangeeta, Vandana, Uma (Sugalicious), Anuresh Chandra (Legend), Gine and Urmila (Priya).
- Sean Mow and the staff at the Olympic Oval for giving us exclusive access to the facilities to enable us to carry out numerous field tests.
- My soccer buddies of Nitin Nand, Rohit Kumar, Ravi Padurath and Rajnay Dayal.
- Mr Emmeron Glennon from Signav Navigation, Australia for his willingness to help and give me much needed advice in the field of AGPS.
- Dr Paul McBurney from Eride Inc for a very nice presentation and advice in the exciting field of AGPS

Dedication



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List of Abbreviations

A/D	Analog-to-Digital
AFC	Automatic Frequency Control
AGC	Automatic Gain Control
AGPS	Assisted GPS
AM	Amplitude Modulation
AoD	Age of Data
ASIC	Application Specific Integrated Circuit
BiCMOS	Bipolar Complementary Metal Oxide semiconductor
C/A	Coarse/Acquisition
CCIT	Calgary Centre for Innovative Technologies
CDMA	Code Division Multiple Access
C/N ₀	Carrier to Noise Ratio
CTL	Carrier Tracking Loop
C/W	Continues Wave
DGPS	Differential GPS
DFT	Discrete Fourier Transform
DLL	Delay Lock Loop
DOP	Dilution of Precision
DSP	Digital Signal Processing
E-911	Enhanced 911
E-OTD	Enhanced Observed Time Difference
FCC	Federal Communications Commission
FFT	Fast Fourier Transform
FLL	Frequency Lock Loop
FM	Frequency Modulation
GPS	Global Positioning System
GNSS	Global Navigation Satellite System
GSM	Global System for Mobile Communications
HDOP	Horizontal DOP
HSGPS	High Sensitivity GPS
I	In-phase
IE	Information Element
IM	Instant Messaging
ION	Institute of Navigation
LBS	Location Based Services
LMU	Location Measurement Unit
LNA	Low Noise Amplifier
LOS	Line of Sight
LSQ	Least Squares
NCO	Numerical Controlled Oscillator
NLOS	Non-Line of Sight

PLAN	Position Location and Navigation
PLL	Phase Lock Loop
PRN	Pseudorandom
PVT	Position Velocity Time
Q	Quadrature- phase
RF	Radio Frequency
RFI	Radio Frequency Interference
SNR	Signal to Noise Ratio
SoC	Silicon on Chip
TEC	Total Electron Content
TOA	Time Of Arrival
TTB	Time Transfer Board
TTFF	Time To First Fix

CHAPTER 1: INTRODUCTION

The US Federal Communication Commission (FCC) has set regulations (Phase I and Phase II) for accurate cellular positioning, i.e. Enhanced E-911 [FCC, 2001]. There has also been an increase in the demand for Location-Based Services (LBS). These services would include mobile applications such as personal navigation using digital maps. These two applications have been pushing for accurate positioning solutions. The Global Positioning System (GPS) has so far proven to be a more accurate positioning solution when compared with existing cellular positioning technologies. GPS can give the end user an accuracy of within 30 m, unlike cellular positioning techniques such as Enhanced Observed Time Difference (E-OTD) which can deliver accuracies in the range of over 100 m [Syrjärinne, 2001]. However, just like any satellite-based technology, GPS suffers from signal obscuration or blockage in urban or indoor environments which would reduce satellite availability and multipath effects from materials such as wood concrete or glass would degrade the positioning accuracy. Furthermore, shadowing or signal attenuation produce weak signal conditions (< -150 dBm) which tend to make it difficult or impossible to acquire or effectively track GPS signals. A conventional receiver's inability to acquire or track GPS signals under weak or degraded signal conditions has prompted the development of High Sensitivity GPS (HSGPS) and Assisted GPS (AGPS) technologies. AGPS and HSGPS carry out longer coherent and non-coherent integration enabling them to acquire and or track weaker GPS signals (-160 dBm).

The Position Location and Navigation (PLAN) Group of the Schulich School of Engineering, University of Calgary has been active in assessing the performance of HSGPS receivers in degraded signal environments using field and Radio Frequency (RF) simulation experiments [MacGougan, 2003]. In indoor environments, with the availability of AGPS, it is now possible to directly acquire signals in weak signal environments using assistance data from a network server or a reference receiver [van Diggelen, 2001]. Recently, various simulation tests have been carried out using an RF simulator to determine the effects of various aiding parameters on the acquisition performance of the AGPS receiver [Karunanayake, 2005b]; however, further research is required to investigate effects of different types of aiding data under various field signal conditions. Tracking tests were carried out to illustrate the similarities between AGPS and HSGPS receivers. While simulation tests provide results based on a limited range of controlled environments, the variety of actual end-user signal reception situations requires a wider array of field test sites that would more realistically indicate actual and distinct challenges to the AGPS receiver in terms of the factors discussed above, based on different aiding scenarios.

1.1 Motivation

The requirement set out by the FCC-E911 phase II mandate requires cell phone service providers to locate mobile users with an accuracy of 50 m for 67 % of the time, and with an accuracy of 150 m for 95 % of the time for handset based technologies [FCC, 2001]. There has been a steady increase in the demand for LBS due to the rise in the number and variety of mobile devices which has led to many exciting applications. Some LBS

applications include mobile-gaming; vehicular or personal navigation; locating restaurants or hotels within a specified range; coordinating the location of groups of friends; and chatting services similar to traditional Instant Messaging (IM) applications such as Yahoo™ and MSN™ Messenger. Across this potential range, factors such as quality of service, limited storage capacity or battery power supply require positioning technologies such as AGPS or HSGPS to implement power-saving implantation strategies and faster signal acquisition schemes for more rapid position fixing. The various LBS application and E-911 mandate requires cell phones to work in many different signal conditions which was major driving force for conducting this research. HSGPS and AGPS receivers were tested in particularly weak signal conditions, where conventional receivers are unable to acquire or track signals.

1.2 Literature Review

Field and simulation tests have been carried out using the SiRF StarII HSGPS and SiRF standard receivers to demonstrate the effects of longer integration time [Shewfelt *et al.*, 2001]. The HSGPS receiver carried out a coherent integration for 1 ms followed by non-coherent integration episodes of 4 ms, 12 ms and 16 ms durations, and was able to detect GPS signal strengths with carrier to noise ratio (C/N_0) levels of 39 dB-Hz, 35 dB-Hz and 30 dB-Hz, respectively. Simulation tests in a weak signal environment (30 to 35 dB-Hz) have shown that the HSGPS receiver had a shorter Time-To-First-Fix (TTFF), as compared to a standard receiver, while field tests that were carried out in a San Francisco road tunnel established that the HSGPS receiver had better solution availability compared to a standard receiver.

To compare the tracking performance of HSGPS (SiRF StarII) and conventional (NovAtel OEM4 and SiRF standard) GPS receivers under different weak/degraded signal conditions, field and simulation tests were carried out by the University Calgary's PLAN Group [MacGougan *et al.*, 2002]. Static field tests were also carried out in a residential wood and concrete garage. The results of the various tests demonstrated the ability of an HSGPS receiver to give position fixes in an indoor environment; the HSGPS receiver was able to deliver a position with an accuracy of 50 m (RMS), while a conventional receiver was unable to provide a position fix indoors. The HSGPS also performed better in terms of positioning accuracy and availability under weak or degraded signal conditions. Simulation tests also demonstrated the HSGPS receiver's superior acquisition sensitivity (10 dB) in comparison to conventional GPS receivers.

Field tests have been carried out by SiRF Technology Inc. using the SiRFLocTM client AGPS receiver under various field test situations [Garin *et al.*, 1999]. Test conditions used were a parking lot, a narrow walkway between tall buildings, a shopping mall with a glass roof, and inside a two-storey building close to a window. Results indicate that the AGPS receiver was able to obtain a position fix with an accuracy of 100 m under most of these field test conditions; the exception, however, was the achievement of a positioning accuracy of only 184 m for the narrow walkway due to poor satellite availability. Further simulation tests have been performed to determine the effects of different power levels, specifically in terms of the TTFF and positioning accuracy [Garin *et al.*, 2002]. The simulation tests revealed longer TTFFs and degraded horizontal positioning accuracy

levels, with decreasing power levels for the AGPS receiver. The SiRFLoc™ is a multimode receiver, which can operate in either an assisted or standalone HSGPS mode.

Field tests were conducted by Moeglein and Krasner [1998] with the aid of SnapTrack™ (a product that was later acquired by Qualcomm Inc.) in an outdoor, urban environment, inside a sport utility vehicle located in a concrete parking garage, as well as in the basement of a two-storey building; in a two-storey office building in the urban centre of Denver, CO; and on the 21st floor of a 50-storey glass building, also in Denver. Test results demonstrate a positioning accuracy, in most cases, of within 30 m, a figure that was progressively degraded with increasing hostility of the testing conditions. The worst results were obtained in the setting of the 50-storey glass-clad office building (84 m for 68.3 % of the best results); the yield or percentage of successful position fixes also decreased, with the receiver delivering a yield of 89% for the 50-storey test.

Data obtained from the deployment of Qualcomm's gpsOne™ solution, which employs a hybrid methodology of AGPS (functioning in the mobile-assisted mode) and cellular positioning technologies, showed that AGPS was used for 84% of the time to obtain a position solution, where the test sites ranged from subways to urban canyon environments, characterised by low satellite availability and highly attenuated GPS signals [Biacs *et al.*, 2002]. The hybrid cellular and AGPS solution can be used to increase the solution availability, employing either the AGPS or cellular individually or in conjunction, to obtain the position solution.

Field tests have been conducted by van Diggelen and Abraham [2001] using Global Locate's GL-16000TM AGPS receiver under various field test conditions such as downtown San Francisco; inside a shirt pocket; in the cab of a steel truck traveling at 112 km/h; within a four-story building; and inside a shopping mall. The maximum TTFF in these trials was obtained at the bottom floor of a four-storey building. Field tests were also carried out in a concrete garage; downtown San Francisco; in a two-storey office building; and within a drawer inside the two-storey office building. Results showed a mean accuracy of within 25 m. The GL-16000TM chipset is a multimode receiver, which uses 16000 hardware correlators, where the aiding data was provided from their worldwide reference network.

Sigtech Navigation's subATTOTM technology demonstrated an acquisition sensitivity of -155 dBm [Bryant *et al.*, 2001]. The receivers were assisted by satellite ephemeris and almanac data, as well as approximate time and position, to hasten the acquisition process. The field tests were carried out in a parking garage on the uppermost level of a three-storey building; and two floors below the top. A signal strength of less than 30 dB-Hz was observed at two floors below the top, with a positioning accuracy of within 50 m.

Numerous simulation tests have been carried out under static conditions to explore the effects of various aiding parameters on GPS receivers [Karunanayake *et al.*, 2004]. Such tests confirm the importance of accurate time or position aiding under weak signal conditions (-140 dBm), while tracking tests have shown that AGPS and HSGPS had the same tracking threshold which was 15 dB better than that of conventional receivers.

1.3 Thesis Objectives

There are no documented attempts within the existing literature detailed above to compare the tracking performance of HSGPS and AGPS receivers, or to determine the effects of different types of aiding data on AGPS signal acquisition under various field test conditions. To address this gap in the literature, the following thesis objectives are proposed:

- 1) Compare the acquisition performance of AGPS under different aiding scenarios using the hardware simulator;
- 2) Investigate the effects of variations in timing uncertainty on AGPS signal acquisition at different field test sites;
- 3) Determine the effects of varying the horizontal position uncertainty on AGPS signal acquisition at different field test sites;
- 4) Determine the effects of satellite ephemeris or almanac aiding on AGPS signal acquisition at different field test sites;
- 5) Compare acquisition performance of an AGPS receiver under different field test conditions; and
- 6) Investigate the tracking performance of HSGPS and AGPS receivers under different field test conditions.

1.4 Thesis Overview

Chapter 2 provides the theoretical background of concepts such as the GPS signal structure; receiver architecture; signal power; signal acquisition or tracking schemes; along with HSGPS and AGPS implementation details. Chapter 3 presents a discussion of simulation tests conducted to determine the effects of various types of aiding data on AGPS acquisition sensitivity. Chapter 4 discusses acquisition tests carried out under various field conditions and aiding scenarios, while Chapter 5 presents the results and analysis from tracking tests carried out in distinct field test conditions using HSGPS and AGPS receivers. Finally, Chapter 6 presents conclusions and recommendations for future work.

CHAPTER 2: AGPS AND HSGPS THEORY

This chapter provides the theoretical background on AGPS and HSGPS. Section 2.1 discusses the basics of the GPS signal structure, measurement attributes, and possible measurement error sources. Section 2.2 then gives a discussion of aspects of the receiver architecture including thermal noise, acquisition schemes, and possible ways of integrating tracking loops through the use of code or carrier tracking loops. In Section 2.3, a discussion of HSGPS challenges and possible implementation schemes is given, followed by a discussion of AGPS concepts and implementation strategies currently employed by certain companies in production model receivers.

2.1 GPS Signal Structure

GPS is a satellite-based positioning system capable of providing a user position anywhere in the world. This system was developed by the U.S. Department of Defense (DoD) to support the military forces of the United States of America by delivering world-wide, real-time positions [Parkinson *et al.*, 1995]. GPS can be used for civilian applications even though it was originally developed for military applications [Spilker and Parkinson, 1996]. The system currently consists of a constellation of 27 (nominally 24) satellites and an associated network of ground stations, which transmit, through the satellites, continuous information for the user to compute position, velocity and time (PVT).

GPS transmits on two carrier frequencies referred to as L1 (1575.42 MHz, the primary frequency) and L2 (1226.7 MHz, the secondary frequency) as illustrated in Figure 2.1.

These two frequencies are modulated by a pseudorandom noise (PRN) code, which is in turn modulated by the 50 Hz navigation data message. Two spreading codes are used to modulate these carriers. The precision P(Y) code is present on both L1 and L2, and has a chipping rate of 10.23 MHz which repeats after a period of 38 weeks.

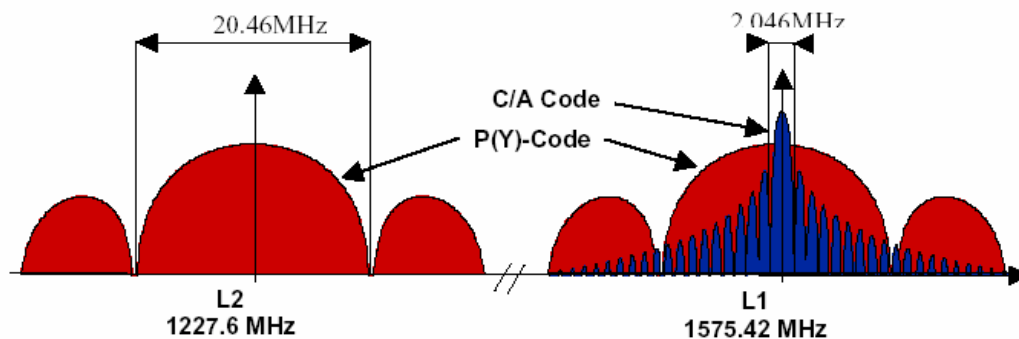


Figure 2.1: GPS Signal Structure [Deshpande, 2004]

Range can be measured by differencing the time of transmission from the time of reception for the GPS signals; however, since the clocks contained in the GPS satellite and the receiver are not synchronized, the measured range is characterised at this point as a *pseudorange* [Kaplan and Hegarty, 2006]. Civilian GPS receivers rely on L1 Coarse/Acquisition (C/A) code measurements, which are modulated on the L1 carrier. The C/A code is replicated in the GPS receiver and can be correlated with the incoming signal to output pseudorange information. Pseudorange measurements from four or more satellites are required to compute three unknowns in the position domain (x, y and z) and the receiver clock bias.

The relative velocity between the transmitter and receiver results in a physical phenomenon known as a *Doppler shift* [Tsui, 2000]. Doppler would cause a change in the frequency that is observed by the receiver due to the relative motion between a receiver and transmitter - in this case the GPS satellite. An analysis of the Doppler effect can be used to compute the user's velocity. At least four measurements are required to compute three velocity components (v_x , v_y and v_z) and the receiver clock drift. The maximum Doppler shift would be 5 kHz for a static user, and reaching up to 10 kHz for a high-speed flying aircraft.

2.1.1 GPS Measurement Error Sources

A GPS measurement is corrupted by errors such as control segment errors; furthermore, satellite clock or ephemeris errors, and uncertainties in the propagation medium may affect the signal's travel time from the satellite to the receiver [Misra and Enge, 2001]. These errors can be categorised into either ionospheric or tropospheric delay components. Noise observed at the receiver is typically caused by interference from surrounding sources. Reflected or multipath signals, also affect the accuracy of the measurement.

The ephemeris and satellite clock parameter values broadcast by the satellite are computed by the control segment with the use of measurements from the GPS monitoring stations. There are errors associated with the prediction of the current and/or future values of the parameters; the prediction errors grow with the Age-of-Data (AoD), which is

defined relative to the time when the parameters were last uploaded. Satellite clock error is due to satellite clock drift with respect to the GPS time reference.

The ionosphere is a region consisting of ionized gas (free electrons and ions) that extends from 50 km to 1000 km above the Earth. The ionization is the result of solar radiation; the speed of propagation of the GPS signal depends on the number and density of free electrons, which is referred to as the Total Electron Content (TEC). The TEC may vary, depending on such factors as solar radiation or geometric distance and is at least one or two orders of magnitude greater during the day than at night. The ionosphere is dispersive; that is, because the velocity is dependent on frequency, ionospheric errors can be eliminated with the use of dual frequency L1/L2 GPS receivers.

The troposphere describes an oblate region consisting of water vapour (found below an altitude of 12 km), and dry gas which can be found 16 km above the equator and 9 km above the poles. The components of the tropospheric error which result from dry gas or water vapour are known as dry and wet delays, respectively. The troposphere is non-dispersive and, thus, its effects cannot be isolated by dual frequency measurements.

Receiver noise is caused by factors such as amplifiers, cables and interference from other sources such as wireless networks or GPS-like broadcast sources which may be augmented with the GPS receiver. Multipath is another source of measurement error, where the Line of Sight (LOS) signal may combine with various reflected components as affected by various reflective surfaces along the path. A further discussion of multipath is

provided in Section 2.3.1. Table 2.1 shows the 1σ values for a range of GPS error sources.

Table 2.1: GPS Error Sources [Source: Lachapelle, 2002]

GPS Error Source	Error Magnitude (1σ) (m)
Satellite clock and orbital errors	2.3
Ionosphere on L1	7.0
Troposphere	0.2
Code multipath	0.01-10
Code noise	0.6
Carrier multipath	50×10^{-3}
Carrier noise	$0.2-2 \times 10^{-3}$

2.2 GPS Receiver Architecture

As illustrated in Figure 2.2, GPS signals are received at the Radio Frequency (RF) front end via a GPS antenna. After performing a series of pre-amplifications, band-pass filtering and down-conversion steps on the GPS signals are conducted to transform them into Intermediate Frequencies (IF) at the RF front end, before the signal is converted into digitized samples using a Analogue to Digital (A/D) converter. The amplification is carried out to set the noise floor, and band-pass filtering is carried out to reject noise, continuous wave (CW) interference or jamming. Meanwhile, the signal is down-converted to enable digitization because signal processing is easier to implement at much

lower frequencies than at the L1 frequency. During signal acquisition, the received signal is correlated with the replica signal that is generated by the GPS receiver, to obtain coarse estimates of the C/A code phase and satellite Doppler. After acquisition of the signal, the satellite can be tracked with the use of tracking loops such as a DLL (code phase), PLL (phase lock loop) or FLL (frequency lock loop). The navigation data bits are demodulated and pseudorange, carrier phase or Doppler measurements obtained from the tracking loops for each satellite are used to compute the user PVT. The following sub-sections provide a detailed explanation of a typical GPS receiver architecture including the received signal power, acquisition and tracking processes.

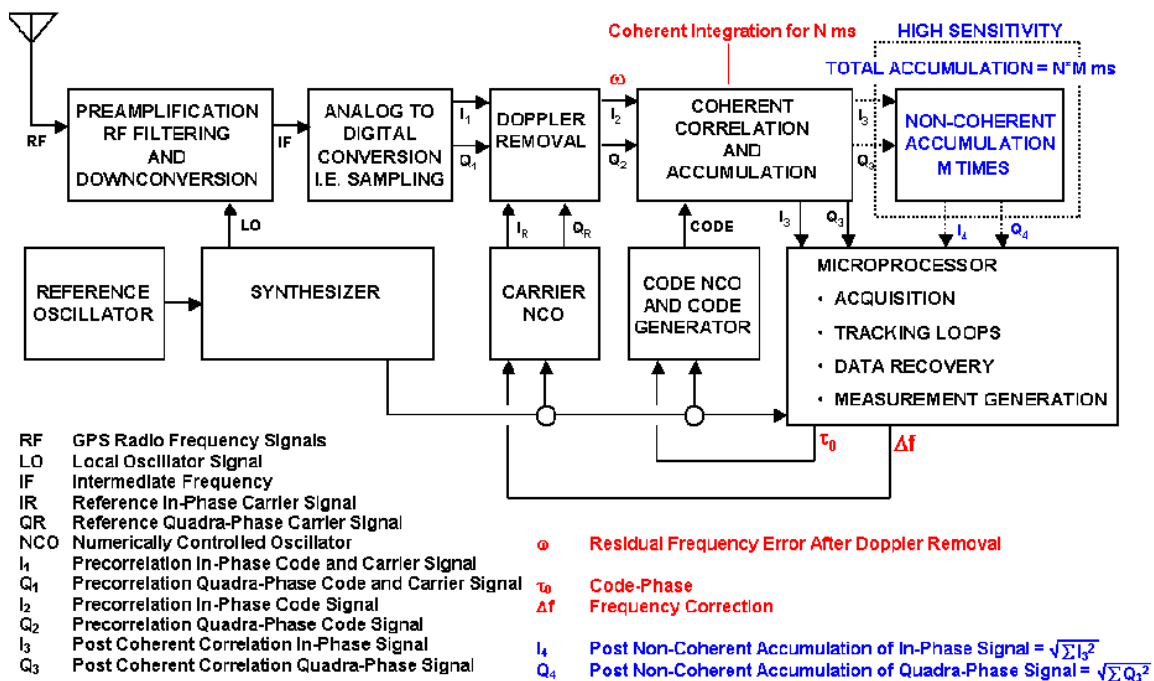


Figure 2.2: GPS Receiver Architecture [Source: MacGougan, 2003]

2.2.1 GPS Signal Power and Signal to Noise Ratio (SNR)

A GPS satellite transmits approximately 27 W of power for the L1 C/A code, which is equivalent to $10\log_{10}(27/10^{-3})$ yielding 44.3 dBm [Misra and Enge, 2001]. The signal encounters free-space path loss that is dependent on the radius from the satellite to the receiver; the transmitted power is increased by redirecting the signal towards the centre and edge of the Earth rather than in all directions, whereby the direction is given by the nadir angle α over the range $\pm 13.9^\circ$. Another important factor is the satellite's elevation angle; a satellite at low elevation has a higher gain of 12.1 dB, while satellites at the zenith provide a gain of 10.2 dB. The gain is determined using the satellite's antenna gain pattern. The properties of the GPS antenna used to capture the signal affect the nature of the received signal; its surface area determines the effective power captured while the gain pattern focuses signal power in certain directions. The user antenna can receive signals only from above the horizon, where the gain is invariant with azimuth; however, the gain does vary with elevation, the particulars of which are captured using the antenna's elevation pattern. There are some antennas that reject interference or multipath from certain directions (from other wireless emitters of GPS-like signals or nearby reflective sources) and can be modeled with the use of the antenna's gain pattern.

Techniques to alleviate the effects of multipath include Rays [2000] study of a multiple antenna array in mitigating carrier phase multipath; and the use of a microstrip antenna array to mitigate interference or jamming [Lin *et al.*, 2002]. The GPS receiver employed in this research uses a microstrip antenna which can either be embedded within the

receiver or connected separately; the advantages of this type of antenna include its small size and low cost.

In comparison to other spread-spectrum communications signals, a GPS signal is very weak, as shown in Table 2.2; however, other signals such as thermal noise generated by the receiver or sources outside the receiver are also weak. For purposes of this study, thermal noise can be modeled as white noise, in which case every frequency component is assumed to have the same power; the power level of this noise is given by 2.1.

$$N_{power} = KTB \quad 2.1$$

where:

- K is Boltzman's constant ($1.38066e-23$ J/K)
- T is the Noise temperature (nominally 273°K), and
- B is the nominal Bandwidth of noise.

Table 2.2: GPS Signal Power [Source: MacGougan, 2003]

SV Antenna Power (dBW)	13.4
S V Antenna Gain (dBW)	13.4
User Antenna Gain (hemispherical) (dB)	3.0
Free Loss L1 for R = 25092 km (dB)	-184.4
Atmospheric Attenuation, (dB)	-2.0
Depolarization Loss (dB)	-3.4
User Received Power (dBW)	-160 or -130 dBm

A GPS receiver uses a *low noise amplifier* (LNA) to achieve the desired high gain and low noise characteristics. It is important to minimize the losses due to components preceding the LNA such as the cable and filters (which reject interference from other frequency bands). The RF front end can be considered as a series of sub-systems - each with a gain, (G_i), an effective temperature (T_i) and noise figure (F_i), The effective operating temperature for the entire receiver system can be expressed with the use of the *Friis* formula (see Equation 2.2). The large gain in the first subsystem would tend to overshadow and, therefore, reduce the effects from the remaining sub-systems. A typical L1 C/A receiver GPS receiver has an RF front end bandwidth of 2.048 MHz, where more than 90% of the GPS signal resides, giving a noise power of -111 dBm (assuming a temperature of 280 K). GPS signals may have a power of -130 dBm, resulting in a Signal-to-Noise Ratio (SNR) of $-130 + 111 = -19$ dB, meaning that the GPS signal is well below the receiver's noise floor. One method of recovering the GPS signal from the noise is to apply correlation until the SNR attains the minimum level of 14 dB required for signal acquisition [van Diggelen 2001].

$$T_R = T_{E,1} + \frac{T_{E,2}}{G_1} + \frac{T_{E,3}}{G_1 G_2} + \dots \quad 2.2$$

2.2.2 GPS Signal Acquisition

Acquisition is a very time-consuming process which can take up to several minutes for a conventional receiver and is carried out to obtain coarse estimates of the Doppler and C/A-code phase before tracking can commence [Misra and Enge, 2001]. The receiver-

generated code is correlated with the incoming code and compared to the acquisition threshold to determine if a useful signal is present. Signal detection is a statistical process; the C/A-code phase or Doppler search bin (described below) may contain either a useful signal or noise; the noise would have a zero mean characterised by a Rayleigh distribution, while a signal with noise has a non-zero mean with a Rician distribution [Kaplan and Hegarty, 2006]. Signal detection is a binary process involving the noise and signal Probability Distribution Function (PDF) in which a useful signal is detected by using parameters such as P_{fd} (probability of detection) and P_{fa} (probability of false alarm). The P_{fd} should be chosen in such a way as to enable signal detection, while P_{fa} must be chosen so as to ensure that noise is not detected as a useful signal. If the receiver does not have *a priori* knowledge of the approximate location, current GPS time or ephemeris, achieving a position fix could take up to several minutes; without initial values for these data, a complete sky search of all PRN codes, all Doppler and code phase bins is carried out and the navigation data needs to be downloaded, each of which could take up to thirty seconds. AGPS receivers (see Section 2.4) rely on aiding data to shorten the acquisition search time. Once the receiver is provided with assistance data such as approximate user position, current GPS time, and satellite ephemeris or almanac, the acquisition search time can be reduced to a few seconds. The receiver can use either hardware or software implementation schemes for signal acquisition [Deshpande, 2003].

The hardware approach is implemented using Application-Specific Integrated Circuits (ASIC) on a chipset. There are two unknowns and the search can be divided into a 2-D search space of Doppler/C/A-code phase as illustrated in Figure 2.3 [Kaplan and Hegarty,

2006]. The 2-D frequency/ C/A-code search space could have a Doppler of ± 4.5 kHz and a 0-1022 chip C/A code phase. Correlation is performed in each cell by using the pre-detection integration and comparing the correlation value with the detection threshold. If the value is less than the threshold, the search goes onto the next cell until a useful signal is found. Two samples per chip are used while searching for the correlation peak in the code space; i.e. there are 2046 samples and, since there are two channels/satellite (In-phase (I) and Quadrature-phase (Q)), there is a total of 4092 samples. The frequency width (f_c) of the bins is calculated by using Equation 2.3. As a rule of thumb, f_c is 667 Hz for 1 ms and f_c is 33 Hz for 20 ms. In order to search for one satellite with a 20 ms integration, there are $4092 \cdot (4500 + 4500) / 67$ bins (given by Equation 2.4), which results in 549672 bins, suggesting that a longer integration time (required to detect weaker signals) results in a longer search time or TTFF.

$$f_c = \frac{2}{3N} \quad 2.3$$

where f_c is the frequency width of the bins and N is the integration time

$$TB = 4092 \cdot (4500 + 4500) / f_c \quad 2.4$$

where TB is the total number of bins, and f_c is the frequency width of the bin.

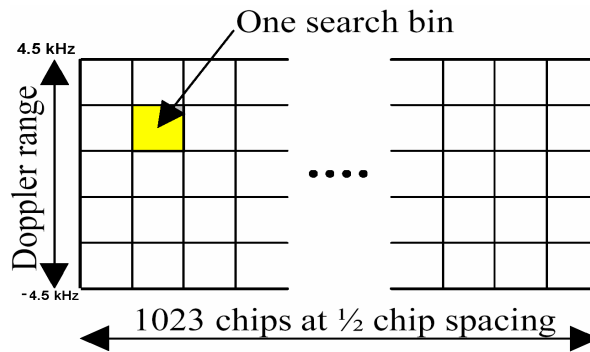


Figure 2.3: 2-D Acquisition Search Space

The software approach is implemented with the use of a Digital Signal Processor (DSP) and offers many advantages, such as easier modification as compared to the hardware approach. Software involves a non real-time process that carries out Fast Fourier transforms (FFT) and multiplies the two signals (receiver generated and incoming) in the frequency domain. The hardware scheme carries out time domain correlation of two signals in real-time. There are many distinct techniques such as circular correlation or multiply-and-delay methods that can be used to conduct signal acquisition. Circular correlation is carried out on periodic signals such as the C/A code (period of 1 ms) using the Discrete time Fourier Transform (DFT) to determine the initial code phase and carrier frequency [Tsui, 2000]. The multiply-and-delay method, which can be used for both the C/A and P(Y) codes, is carried out to determine the initial code phase which is subsequently used to determine the carrier frequency [Lin and Tsui, 2001]. GPS signals as weak as 30 dB-Hz can be acquired by processing longer samples or records of data such as 40 ms [Lin and Tsui, 2001]. Research by Psiaki [2001] demonstrated successful acquisition of signals as low as 21 dB-Hz based on the processing of 4-second blocks of data using methods such as alternate half bit or full bit.

2.2.2.1 Coherent Integration

The L1 C/A code has a length of one millisecond and, so, coherent integration would require successful correlation of at least one millisecond of incoming signal with the locally generated replica. Coherent integration involves a summation of the In-phase components. It is a band-pass process, in which each frequency bin represents a filter and the bandwidth is inversely proportional to the integration time; a longer integration time would filter out more noise and, hence, result in higher sensitivity [Zhengedi, 2000]. When conducting coherent integration, signal power increases by N while the noise power increases by \sqrt{N} resulting in a SNR gain of \sqrt{N} . A longer period of coherent integration results in finer frequency resolution (see 2.3); hence, a larger processing gain can be realized, achieving higher sensitivity at the expense of a longer search time [van Diggelen, 2001].

The 50 Hz navigation data is modulated with the C/A code imposing a limit of 20 ms on coherent integration. The In-phase component incurs a sign inversion when it undergoes a navigation data bit transition. Zhengedi [2000] has successfully achieved optimal gain levels based on 10 ms of coherent integration. Longer integration times resulted in greater loss as the probability of crossing the navigation bit boundary is increased, thus producing errors that result in acquisition loss. Further research has shown the issue of longer integration times, for example coherent accumulation over 20 ms resulted in a loss of 24 dB using signal of 40 dB-Hz [Dafesh and Fan, 2001]. The loss was lower for weaker signals since the thermal noise became a relatively more significant error source -

for example, where there was a loss of 3 dB with a signal strength of 20 dB-Hz. The signal losses were lower for smaller integration times; that is, an integration time of 10 ms led to a loss of 16 dB for 40 dB-Hz signals. Longer integration is possible after bit prediction; a gain of 6 dB was achieved when signals were integrated for 20 ms as compared to a 1 ms integration period. Longer coherent integration is further limited because of residual frequency errors such as receiver or satellite induced motion, or local oscillator clock drift, which would cause the signal power to oscillate between I and Q components [MacGougan, 2003]. Park *et al.* [2004] have shown that a coherent integration of 16 ms causes a frequency error of 31.25 Hz, while an integration time of 64 ms causes a frequency error of 7.82 Hz, when the correlation magnitude was reduced to half of the original value. A more stable clock can be used to extend the coherent integration assuming the navigation data bits are known [Sudhir *et al.*, 2002]. If there is time, non-coherent integration can be carried out to further enhance the sensitivity.

2.2.2.2 Non-Coherent Integration

Non-Coherent integration involves the square root of the summation of squares of the I and Q components [van Diggelen, 2001]. Squaring the amplitude eliminates navigation data bits and, thus, non-coherent integration does not require knowledge of the navigation data bit transitions. However, the gain comes at a price; non-coherent integration modifies the noise behaviour, producing a non-zero mean which causes *squaring loss*. A higher SNR can be obtained by carrying out longer coherent integration which would result in lower squaring loss. If the SNR is positive, the squaring loss is not excessive;

however, a negative SNR results in an inordinately high, and possibly disastrous, squaring loss [Mattos, 2003]. Sensitivity can be enhanced by carrying out coherent integration followed by non-coherent integration. For example, a gain of 20 dB can be achieved by performing 10 ms of coherent integration followed by 19 ms of non-coherent integration [Shewfelt *et al.* 2001]. Equation 2.5 shows the total processing gain that can be obtained by successive stages of coherent and non-coherent integration.

$$G = 10\text{Log}(N) + 10\text{Log}(M) - SQ_{Loss} \quad 2.5$$

where

G is the gain in dB

N coherent integration time in milliseconds

M non-coherent integration time in milliseconds, and

SQ_{Loss} Squaring loss.

2.2.2.3 Comparison of Coherent and Non-Coherent Integration

Following a discussion of non-coherent and coherent integration, it is imperative to consider the respective benefits and drawbacks of these methods. Coherent integration requires a shorter integration time to achieve the same acquisition sensitivity versus a comparable non-coherent integration; for example, an integration time of 100 ms (coherent) will achieve the same acquisition sensitivity as 1000 ms of non-coherent

integration. Non-coherent integration is more tolerant to residual frequency errors and is not affected by the navigation data bits. The frequency resolution is smaller for coherent integration (two times as compared to the non-coherent case), suggesting that coherent integration is able to filter out more noise (that is, has higher sensitivity) at the expense of longer search time for the same integration length. A comparison of the two methods is shown in Figure 2.4.

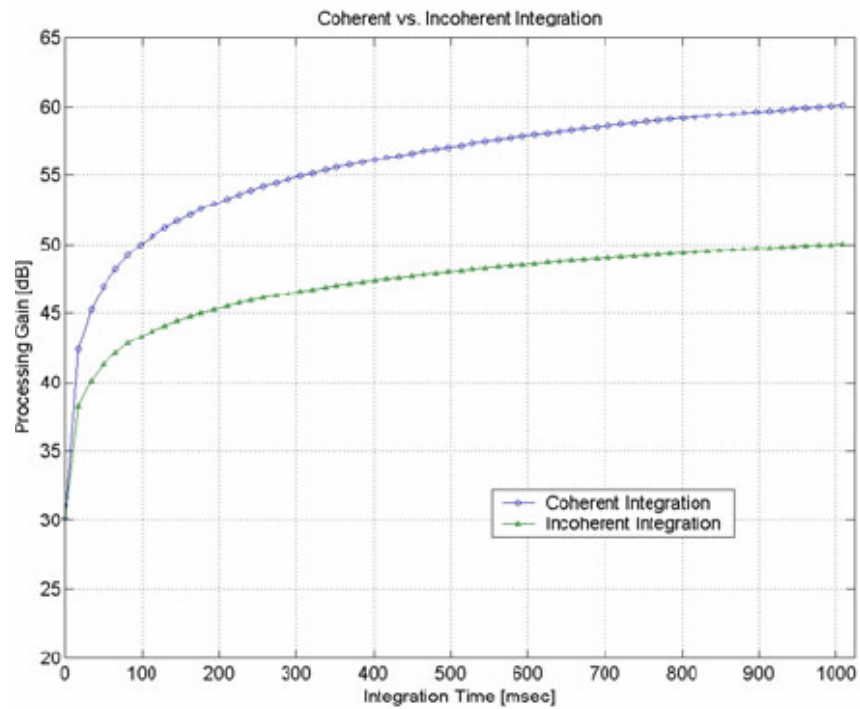


Figure 2.4: Comparison of Coherent and Non-Coherent Integration [Source: Park *et al.*, 2004]

2.2.3 GPS Signal Tracking

Following acquisition of the satellite signal, the associated Doppler and C/A code phase are found. The receiver can be reconfigured in such a manner that a code tracking loop, such as a Delay Lock Loop (DLL), is used to track the C/A code phase, while a carrier tracking loop, such as a PLL or FLL, is used to track the carrier phase. A tracking loop is a feedback control system, which is used to minimise errors such as code phase, carrier phase or frequency errors. The next few sub-sections will explore the various types of code and carrier tracking loops, along with factors such as clock stability, multipath and their effects on the performance of the tracking loops.

2.2.3.1 Code Tracking Loop

The Delay Lock Loop (DLL) measures the C/A code phase of the incoming signal, which is used to estimate the transit time of the satellite, hence, to compute the pseudorange measurements [Misra and Enge, 2001]. Pseudorange measurements are later used to compute the navigation solution. The objective of the DLL is to align the incoming signal with the replica code. The received signal is compared with the replica code to generate the code phase error. The code phase error determines how the code generator must be adjusted so that the replica code and the input signal can be aligned to facilitate subsequent satellite tracking.

The incoming signal contains the navigation data (modulated at 50 Hz), along with the code Doppler and carrier phase [Misra and Enge, 2001]. After Doppler and carrier frequency removal, three correlators (early, prompt and late) are used to track the rising, peak, and falling edge of the signal. If the GPS signal is being tracked (that is, if it is aligned), this implies that the prompt correlator will have ascertained the maximum value for tracking of the correlation peak. The signal is then correlated with the locally generated code for a predefined integration time. The resulting signal is then fed to a discriminator that can be either coherent or non-coherent. A coherent discriminator requires an accurate estimate of the carrier phase; generally, a non-coherent discriminator is used to avoid over-dependence on the carrier tracking loop. There are different types of coherent and non-coherent discriminators, which have been discussed in the GPS literature such as Kaplan [1996] and will thus not be addressed here. The non-coherent discriminator removes the carrier phase and code Doppler. The output from the discriminator constitutes the error between the early and late correlators, which is filtered using the code loop filter, the output of which is fed to the code generator to determine whether to slow down (if the replica signal is late) or speed up (if the replica signal is early) to ensure that the replica code is aligned with the incoming signal.

The spacing between the early and late correlators - known as *correlator spacing* - can be 1, 0.5 or 0.1 chips in magnitude; the first two of these are the basis of “wide correlators,” while the smallest and last interval in this group is fundamental to the “narrow correlator” (developed commercially by NovAtel Inc.). It will be shown later in this thesis that

correlator spacing is an important design parameter that can be used to mitigate the effects of multipath.

The dominant sources of range errors for the DLL include the dynamic stress error and thermal noise jitter [Kaplan and Hegarty, 2006]. Dynamic stress error is due to the filter order and bandwidth, while thermal noise-jitter is due to tracking loop characteristics such as filter bandwidth, correlator spacing and pre-detection integration time. The tracking sensitivity can be enhanced by either increasing the pre-detection integration time (which would lower the squaring loss), or by decreasing the filter bandwidth to filter out more noise; the effectiveness of the latter procedure may be limited by factors such as local oscillator clock drift or user dynamics. Decreasing the correlator spacing would lower the tracking threshold, at the expense of reduced tolerance to dynamic stress.

2.2.3.2 Carrier Tracking Loop

There are two type of tracking loops that can be used to track the carrier phase. These are the PLL, which is usually a Costas loop in GPS receivers, and the FLL [Misra and Enge, 2001]. The FLL is also known as automatic frequency loop control (AFC) since it tries to adjust the frequency to minimise carrier phase error. A carrier tracking loop adjusts the Numerical Controlled Oscillator (NCO) so that the phase error between the input signal and the receiver-generated signal is zero or approximately zero.

The incoming signal is multiplied with the replica generated by the NCO and the resulting signal is multiplied with the in-phase code replica which then undergoes integration [Kaplan and Hegarty, 2006]. The signal is then fed to the discriminator, which can be either an FLL or a Costas PLL. The Costas loop is used in GPS receivers rather than pure PLLs because of its insensitivity to data bit transitions resulting from the navigation message. The signal is then filtered using a loop filter which can be either of first, second, or third order and which is capable of withstanding velocity, acceleration or jerk dynamic stress. The NCO's phase or frequency is adjusted appropriately and the whole process is repeated, until the phase or frequency error is approximately zero.

The dominant sources of range errors for the carrier tracking loop are dynamic stress error and thermal noise jitter [Kaplan and Hegarty, 2006]. Similar to the code tracking loops, the PLL and FLL are subject to tracking errors such as dynamic and thermal noise. The nature of thermal noise depends upon factors such as the C/N_0 , integration time and filter bandwidth. A relatively lower value of C/N_0 , higher loop bandwidth or lower integration time (that is, higher squaring loss), will result in higher thermal noise, resulting in larger carrier phase or velocity errors. Typically, Costas loops have bandwidths of 1 Hz, while FLL's may have a filter bandwidth of 25 Hz; thus, based on this structural difference alone, an FLL is able to accommodate greater receiver dynamics. By comparison, a Costas PLL is profoundly insensitive to dynamic stress but retain the ability to provide the most accurate estimate of user velocity measurements. In practice, GPS receiver design may incorporate both FLL and Costas PLL components, switching to FLL in case the Costas PLL loses lock under higher dynamics.

2.3 High Sensitivity GPS Challenges

This section will focus on the operational challenges inherent in HSGPS receivers and the associated implementation issues. Conventional GPS receivers were designed for outdoor LOS signal conditions. The application reality requires that GPS operate for LBS and E-911 situations. In order to meet these requirements, GPS must work in weak/degraded signal environments where there may be limited or non-LOS (NLOS) signals, significant signal blockages, highly attenuated signals and cross-correlation effects from nearby strong signals. The next few subsections discuss multipath effects, weak or degraded signal conditions and implementation details of the HSGPS receiver.

2.3.1 Multipath

The GPS signal may be reflected from surfaces before entering a receiver's RF front end. This phenomenon known as *multipath*, effectively distorts the TOA of the received signal, which causes a bias in the pseudorange measurement. Multipath is a localised phenomenon, which depends on the distance between the antenna and the reflector, as well as the type of reflecting surfaces involved. Multipath is always delayed with respect to the primary GPS signal of interest because of a longer travel time due to reflection of LOS and reflected signals. The composite signal can be expressed by Equation 2.6 [Braasch, 1996]

$$s(t) = -Ap(t) \sin(\omega_0 t) - \sum \alpha_m A_p(t + \delta_m) \sin(\omega_0 t + \theta_m) \quad 2.6$$

where

$s(t)$ is the composite signal,

A is the amplitude of the direct signal.

$p(t)$ is the pseudorandom noise sequence of the specific C/A code,

ω_0 is the frequency of the direct signal,

α_m is the relative power of the multipath signal,

δ_m is the delay of the multipath signal with respect to the direct signal, and

θ_m is the phase of the multipath signal with respect to the direct signal.

Multipath can consist of either diffuse or specular reflections. If signals are reflected by surfaces such as wood or concrete that are characterised by a texture that is relatively coarse, the result is *diffuse* reflections ($\alpha_m \ll 1$). However, if the signals bounce off relatively smooth surfaces such as metal or glass, *specular* reflection occurs (α_m is close to one). If a receiver is close to a large smooth reflector, the reflected signals may actually be stronger than the LOS signal ($\alpha_m > 1$) which would have a significant effect on the magnitude of the pseudorange error which, in turn, would degrade the position accuracy [MacGougan, 2003]. The magnitude of the multipath depends on the reflector's spacing from the receiver (and will determine the value of δ_m), the strength of the reflected signal, the correlator spacing and the bandwidth of the receiver. Signals inside a building will consist of attenuated LOS signals, complemented by many reflected or echo-only components; thus, HSGPS receivers should ideally be able to track under echo-only or NLOS conditions.

2.3.2 Weak or Degraded GPS Signals

Signal strength degradation can be caused either by *shadowing* or *fading* [MacGougan *et al.*, 2002]. Shadowing is the attenuation of the LOS signal whilst propagating through materials such as wood or concrete. Fading is due to constructive and destructive interference of multipath on the GPS signal. GPS signals are also susceptible to signal blockages from certain directions. The tall glass buildings in urban or suburban environments, for example, would block satellites which would reduce satellite availability (resulting in poor geometry); the glass structures would cause strong specular reflections, thus introducing multipath effects. Both of these factors would inherently degrade the user positioning accuracy. An indoor environment would typically entail highly attenuated signals and / or short delay multipath effects; these two factors would also degrade the user positioning accuracy. Field tests have been carried out using HSGPS receivers in various indoor and forested environments. These tests have demonstrated signal attenuations of up to 15 dB in a wooden house, 20 dB in a concrete building and 15 dB in a forested area where trees were primarily coniferous.

Buildings can be constructed of reinforced concrete, steel or wood, with the windows made of highly reflective glass panes of varying degrees of opacity. The obstructing surfaces would attenuate the incoming GPS signals by as much as 25 dB, as compared to outside LOS signal conditions [van Diggelen, 2001]. Some LOS signals may enter via the glass windows by diffraction or reflections off the surfaces, with the former being highly attenuated and, thus, not much of a concern; however, multipath signals present a major

problem since they are not as highly attenuated as compared to the LOS signals. Furthermore, the dimensions of the building are typically between 15 and 100 m, which are much smaller than the C/A code chip length of 300 m. Short delay multipath is a major issue for indoor positioning research [Peterson *et al.*, 1997].

The other main issue presented by indoor environments is the prospect of satellites with strong signal power (entering from the window) interfering with weaker signals and causing *cross-correlation* effects. Cross-correlation denotes a situation where a satellite is incorrectly identified, possibly resulting in large pseudorange and Doppler errors and, thus, severely degrading position and velocity solutions. The CDMA (Code Division Multiple Access) isolation between two C/A codes or satellites is limited to 21 dB separated by a Doppler of 1 kHz [Kaplan and Hegarty, 2006]. The more difficult situation is the off-frequency resolution where the C/A code repeats every kilohertz; thus, a receiver may find a satellite a kilohertz or an integer multiple of one kilohertz away when it is, in fact, a strong satellite [Mattos, 2001].

2.3.3 HSGPS Implementations

HSGPS receivers have been developed to acquire or track signals in weak or signal-degraded environments. The acquisition sensitivity of an HSGPS receiver can be enhanced by instituting a longer coherent or non-coherent integration time; by comparison, the tracking threshold can be lowered by increasing either the pre-detection or coherent integration time, or by reducing the tracking loop bandwidth.

When the integration time is increased, the result is a longer search time which can be reduced by conducting a parallel C/A code search, implemented using either hardware or software methods. Hardware correlators perform real-time correlation in the time domain while the software approach takes a sample of data which is processed later in the frequency domain using FFTs. SiRF currently has StarII, StarIII, while GlobalLocate has GL16000 and GL20000 chipsets in the market which uses a massive parallel correlation technique. Research has been carried out to show that a parallel search combined with sequential detection can be used to improve signal acquisition for a receiver which uses massive parallel correlators [Rounds and Norman, 2000]. Eerola [2000] has confirmed that acquisition can be reduced to four second duration for a 35 dB-Hz signal, when carrying out parallel correlation using *matched filters*. Matched filters are devices that continually correlate between the replica and the incoming signal, with the output being the maximum correlation value. The troublesome issue of direct cross-correlation, can be resolved by carrying out step by step sensitivity testing at various power levels, while the off-frequency effects could be eliminated by instituting longer integration times. The decision to reject or accept a given correlation peak can be made using logical methods [Mattos, 2003]

Because indoor or signal-degraded environments such as urban canyons constitute, at the same time, limited sources of LOS signals and multiple sources of reflected or multipath signals, the HSGPS should be able to acquire and / or track these reflected signals. Each tracking loop has multiple correlators to track distinct reflected signals – a functionality

which is used to compute the pseudorange further into the data processing. An HSGPS receiver generally uses a combination of an FLL and Costas PLL carrier phase tracking loops. Under high receiver dynamics, the HSGPS would use the FLL tracking loop, but would revert to the Costas PLL with one hertz bandwidth under static conditions [Kaplan and Hegarty, 2006]. Sudhir [2001] gives evidence that the tracking sensitivity of an HSGPS receiver can be enhanced by performing longer coherent integration with the use of a prediction algorithm to determine the navigation data bits. The receivers have been shown to have a tracking sensitivity of -147 dBm when carrying out coherent integration of 80 to 100 ms. Further explorations by MacGougan, [2002] and Karunanayake *et al.*, [2004] have demonstrated tracking sensitivities of -156 dBm with the use of SiRF. SiRF StarII and SiRF XtracTM receivers carry out coherent integration for 80 to 100 ms [Garin, 2005].

GPS signals are very weak in an indoor environment, with the low SNR resulting in high bit error (BER) rates, thus making it very difficult for the receiver to demodulate the navigation data bits. Essentially, this implies that acquisition from a cold start is not possible in an indoor environment for HSGPS. HSGPS receivers may require an initialization of fifteen to twenty minutes under open sky conditions before being brought into an indoor or signal-degraded environment to carry out the tracking tests at various field test sites [MacGougan, 2003]. On the other hand, a GPS receiver could use assistance from a wireless network to carry out signal acquisition under weak or signal-degraded conditions. The following section presents a general discussion of AGPS.

2.4 Assisted GPS (AGPS)

Assisted GPS receivers (shown in Figure 2.5) have been developed to enable GPS operation in weak signal conditions such as indoor environments. AGPS receivers have been embedded in wireless devices such as cell phones or personal digital assistants (PDAs), which are intended to allow constant use across a range of receiving environments – that is, from clear open sky conditions, to urban areas, to indoor office environments. AGPS, as the name implies, requires assistance data to aid in the acquisition process, thus reducing the acquisition search time. Assistance data would include items such as satellite ephemeris or almanac, timing, position and frequency information. Assistance data can be delivered via different wireless networks which use wireless standards such as the Global System for Mobile Communications (GSM) or CDMA, among others. An AGPS receiver can function in either *mobile-based* or *mobile-assisted* mode. In mobile-based mode, the server provides assistance and the position is calculated at the mobile device (e.g., SiRFLoc™) while in mobile-assisted mode, the mobile device acquires the GPS satellites and pseudorange measurements are then sent to the server where the position is computed (for example, gpsOne™ operates in this way). The following subsections describe the assistance message that can be sent via a wireless network, different types of assistance data and their relevance in reducing the acquisition search space.

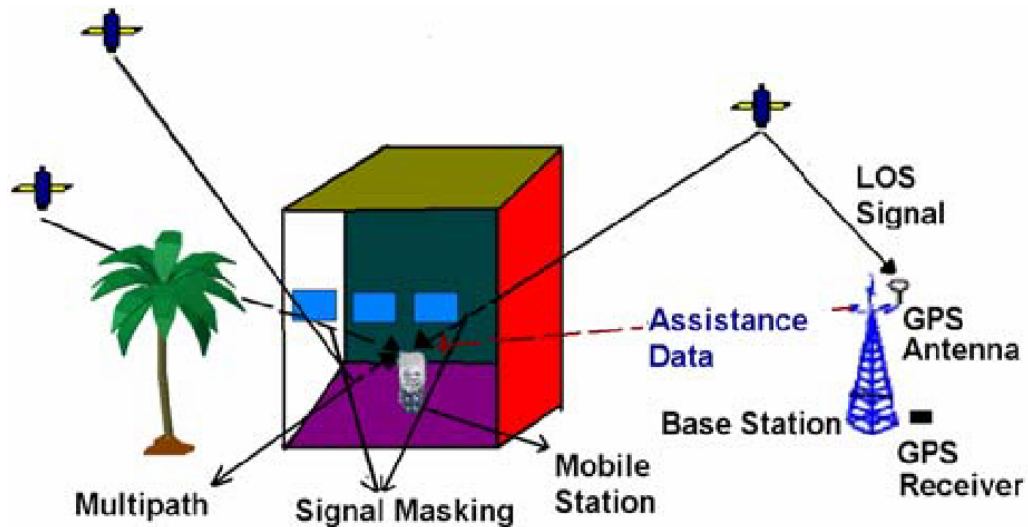


Figure 2.5: AGPS Concept

2.4.1 Assistance Data –Wireless Networks

The method of sending assistance data has been set out in various wireless standards, which are described below. The location assistance data has been part of wireless standards such as the GSM 99 LoCation Services (LCS) protocols (www.etsi.org). The location assistance data can be transmitted either via *point-to-point* or *broadcast* methods. Sending point-to-point messages requires a dedicated two-way connection between the Mobile Station (MS) and the Service Mobile Location Center (SM-LC). The broadcast method would send messages using a one-to-many connection; thus, the SMLC is capable of broadcasting redundant assistance data to many MSs simultaneously.

The messages are delivered via structures known as Information Elements (IE). Table 2.3 shows various IEs which can be transmitted via a point-to-point mechanism for a GSM network; similar messaging structures exist in CDMA networks and, so, will not be

addressed here. The messaging structure includes such things as the reference time, reference location, Differential GPS (DGPS) corrections (which are, more or less, obsolete); the latest ephemeris or navigation components; ionospheric components to correct for the ionospheric delay; UTC time offsets; and acquisition assistance which includes elements such as satellite Doppler estimates for mobile-assisted solution and real-time integrity. The messaging structure contains the base station capabilities, which can be used to transfer the reference frequency. Navigation data bits (also known as sensitivity assistance) can be sent to perform bit cancellation and, thus extend coherent integration beyond 20 ms, while the approximate C/A code phase can be sent to narrow the acquisition search space. Other assistance data, such as C/N_0 , are specified in the standard which aims to reduce the acquisition search time. Once the C/N_0 is known, a certain predefined value of the integration time can be used to carry out the correlation process, which shortens the acquisition search process [McBurney, 2005]. Typically, the value of the integration time is determined after acquiring the strongest satellite which is then fed to the tracking loop.

Table 2.3: Message Structure for Point to Point Method [Source: LCS. 1999]

Information Elements	Description
Control Header	Table of contents of IEs
Reference Time	Contains GPS TOW and week number
Reference Location	3D Location Assistance
DGPS Correction	Corrections used in differential mode
Navigation Model	Latest Satellite Ephemeris
Ionospheric Model	Latest Ionospheric Coefficients
UTC Model	UTC coefficients
Acquisition Assistance	Acquire satellites: mobile-assisted mode
Real-time Integrity	Real-time status of the GPS constellation

2.4.2 Almanac and Ephemeris Aiding

The 50 Hz navigation data bits contain information that is both unique and common to all of the transmitting satellites. The navigation data message contains satellite clock correction data, ephemeris (broadcast orbital parameters) and almanac (coarse orbital parameters), health data for all the satellites, coefficients to compute the ionospheric delay for single frequency users, and coefficients to determine UTC time from GPS time.

The navigation message contains 25 frames of data, where each frame is further divided into five sub-frames. Each sub-frame, which is 300 bits long, is further divided into ten

30-bit words. The first two words contain the telemetry message and the time, while the remaining eight words contain data specific to a sub-frame. The first three sub-frames contain satellite ephemeris data, while the remaining two (four and five) sub-frames contain almanac information for all of the satellites. It takes six seconds to download every sub-frame, thus requiring no more than eighteen seconds to download the complete satellite ephemeris (sub-frames 1, 2 and 3); one full frame can be downloaded in thirty seconds. Downloading the complete satellite almanac (25 frames of data) would take twelve and a half minutes.

A satellite ephemeris contains precise orbital parameters that can be used to compute a given satellite's position, and which are valid for only four hours - after which data quality deteriorates. A satellite almanac contains a subset of ephemeris parameters but is valid for a full week. Sub-frame four contains the almanac and health information for SVs 25 to 32, while sub-frame five contains almanac and health information for SVs 1 to 24 [ICD, 2000].

2.4.3 Time and Approximate User Position Aiding

In one sense, everything in GPS theory can be reduced to its essential timing issues; synchronization of any receiver to GPS timing is critical for satellite positioning and ranging. In order to attain synchronization the exact code phase of a satellite must be known. Timing assistance can be used to predict the code phase; however, on its own, it would not be very helpful. A reasonable estimate of approximate position is also required

to predict the code phase. The positioning assistance can be obtained using E-OTD in a GSM network. The AGPS would receive its timing assistance from a wireless network, which must be synchronized with GPS time. A wireless network such as GSM or UMTS could use the Location Measurement Unit (LMU) to time-stamp the data frames with the GPS time that is being sent to the mobile device. GSM and UMTS are not synchronized with GPS time and, hence, the LMU is required to facilitate synchronization with GPS time. It can be used to provide measurements such as timing or position assistance which are required by an MS for positioning. After accounting for propagation delays due to air interface and signalling, the timing accuracy can be as good as ten microseconds. GSM or UMTS networks have been shown to have a 1σ error of 126 m and 25 μ s [Syrjärinne and Kinnari, 2002]. The CDMA network is synchronized with GPS time and, therefore, only the propagation delay needs to be removed, achieving a timing accuracy in the microsecond range. GPS time for a channel can be given by Equation 2.7 [Syrjärinne, 2001]. It is important to know the approximate location of the code phase so that time is not wasted searching in the wrong place; this also allows the performance of longer integration periods to achieve higher sensitivity.

$$T^J_{GPS}(K) = T^J_{TOW}(k) + T^J_{ms}(k) + T^J_{chip}(k) + 0.078 \quad 2.7$$

The GPS time estimate in the standard positioning service is derived (at an internal time event, k) from a combination of three measured time elements and an average signal time of flight (TOF) of 78 ms. The $T^J_{TOW}(K)$ is the TOW from the most recent sub-frame in seconds, $T^J_{ms}(K)$ is the number of integer C/A-code epochs elapsed since the beginning of

the sub-frame, $T_{chip}^j(K)$ is the combined chip count (integer chips, 0-1022) and chip phase (fractional chips) measurement in seconds and j is the index of the channel. A closer look at Equation 2.7 reveals that the navigation data or the ephemeris would give one $T_{TOW}^j(K)$ and the $T_{ms}^j(K)$, while the accurate timing and position or C/A code phase assistance would enable prediction of the fractional code chip or $T_{chip}^j(K)$.

Accurate position assistance, along with accurate timing assistance, can be used to determine the visible satellite constellation. Timing assistance would directly, while the approximate user position via elevation projected on the LOS component predicts the approximate C/A code phase (see Equation 2.8) [Kaplan and Hegarty, 2006]. The C/A code has a period of one millisecond and, therefore, the approximate user position and timing assistance accuracy should be less than 1 ms. If the time is exact, the position assistance accuracy can be less than approximately 300 km; similarly, when the position is exact, the timing assistance accuracy can be less than one millisecond to enable C/A code prediction. Position assistance with an accuracy of 3 km, or a timing accuracy of 10 μ s can be used to predict code chips to within 10 chips. Timing and position assistance accuracy of more than a millisecond would force a complete sweep of all 1023 C/A code chips, thus lengthening the acquisition search time. Even a combined timing and positioning accuracy of one millisecond can be used to predict data bits, thus increasing acquisition sensitivity [Syrjärinne 2001]. Accurate position, coupled with ephemeris data, can be used to predict the approximate satellite velocity, yielding the approximate satellite Doppler and, therefore, reducing the Doppler search space [Kinnari, 2002].

$$\sigma_{cp}^2 = 4\sigma_{pos}^2 \cos^2(\phi_{el}) + \sigma_{cp_time}^2 \quad 2.8$$

where

σ_{cp} is the C/A code phase chips,

σ_{pos} is the position uncertainty,

ϕ_{el} is the elevation angle,

σ_{cp_time} is the timing uncertainty.

2.4.4 Frequency Aiding

Elements of frequency aiding, such as clock drift and / or clock bias, can be added to the AGPS chipset to reduce the number of search bins. The AGPS chipset embedded in cell phones contains inexpensive un-compensated crystal oscillators; nevertheless, this type of frequency assistance is essential to hasten the signal acquisition process. A CMDA network can provide rather accurate frequency assistance, since the time is synchronised with GPS time, while the same function can be performed by a GSM network equipped with an LMU. If assistance is not available from an external network, a Real Time Clock (RTC) or an internal clock can be used to provide frequency aiding. The GPS receiver will use oscillators such as a Temperature-Controlled Oscillator (TCXO) which has specifications of 0.5 to one part per million (ppm). Having an accurate clock bias or clock drift narrows the Doppler search space, thus reducing the search time.

2.4.5 Interference Effects

AGPS receivers are commonly embedded into mobile devices; which would introduce radio frequency interference (RFI) signals possibly originating in wireless networks such as GSM or CDMA. Generally wireless signals are stronger than GPS signals, which could pose severe problems to AGPS in terms of signal acquisition, tracking and reliability of the positioning solution.

Mobile phones use FM signals for communication and the incorporation of GPS into a cellular handset means that a jammer will be operating nearby at the cellular frequency. For example, GSM phones used in Europe work either on the 900 MHz or 1800 MHz frequency bands while North American GSM phones primarily use the 1900 MHz band. CDMA technology is the basis for Interim Standard 95 (IS-95) and operates in both the 800-MHz and 1900-MHz frequency bands in the US [Paddan *et al.*, 2001]. Frequency allocation and the handset power specifications for some wireless standards are summarised in Table 2.4.

Table 2.4: Mobile Operating Frequencies [Source: Paddan *et al.*, 2001]

Cellular Standard	Transmit Frequency (MHz)	Max. Handset Output Power (dBm)
GSM	880-915 and 1710 –1785	+33
IS-95	824-849	+23
PCS	1850-1910	+24

Any signal that is not modulated is called a continuous wave (CW) signal and may cause interference if in close proximity to the GPS L1 signal. The effects of FM, AM and CW on AGPS performance have been discussed in Karunanayake, [2005]. This research demonstrated that AGPS had better acquisition performance as compared to HSGPS and conventional GPS receivers; however, AGPS and HSGPS had the same tracking thresholds. These results may be explained by the hypothesis that aiding data helps only in the acquisition process; aiding data provides coarse estimates for the Doppler and code phase and, thus, cannot be used by the tracking loops which require finer estimates of the C/A code phase and Doppler. RFI has the same effect on AGPS performance as signal attenuation due to blockage, shadowing or multipath; that is, it lowers the C/N_0 which, in turn, lowers the SNR, thus affecting signal acquisition or tracking of the GPS signals.

2.4.6 AGPS Implementation

The AGPS chipset is embedded in a cell phone or another mobile device, thus affording GPS manufacturers - in addition to achieving high sensitivity - lower TTFF, and greater position solution availability. Exposure to the demands of this segment of the wireless market has allowed GPS manufacturers to adapt to issues such as low power consumption due to limited battery power supply, and the need for miniaturization and, thus, limited system resources (CPU or RAM size). End-users in this market segment also demand a higher quality of service. As an example of this market thrust, this section describes how base-band processing is carried out in an AGPS chipset, and the main differences as compared to conventional HSGPS receivers.

There are three approaches which can be used to embed GPS into a mobile device. The first approach requires a dedicated logic architecture where all of the processing is carried out on the chip (for example, the GL20000 produced by GlobalLocate Inc). The second approach involves an FFT which is implemented using a DSP (NavStreamTM 3000, Parthus Inc) The third approach uses a standard System on Chip (SoC) design which accords with standard practice in GPS receivers (SiRF StarII and SiRF StarIII, SiRF Technology Inc) [van Diggelen, 2001].

Power savings can be realised in a number of ways; for example, faster position fixing using assistance data or different modes such as instant fix or push-to-fix so that the receiver would “wake” up only once every minute or so to obtain a position fix.

Operational modules such as tracking and acquisition can be shut down until required to obtain a position fix, while the clock continues running to maintain an accurate estimate of the time. Research has shown that Silicon Germanium (SiGe) Bipolar Complementary Metal Oxide Semiconductor (BiCMOS) fabrication holds much promise in terms of factors such as cost, smaller size and lower power consumption as compared to traditional silicon fabrication. Thus, BiCMOS would be suitable for embedded handheld applications [Haynes, 2002]. Companies such as SiRF Technology Inc., Nemerix and μ Blox have developed AGPS chipsets with power ratings of less than 50 mW.

Figure 2.6 illustrates a typical architecture of a mobile-assisted AGPS receiver. Usually the AGPS chipset is embedded in an MS. The assistance data such as approximate satellite Doppler shift, GPS timing, or PRN code phase can be sent to the MS from a server via wireless network or downlink. The satellites are acquired; pseudorange measurements are calculated and sent to the server (data uplink) which then computes the user position. The architecture would differ for a mobile-based AGPS receiver that is the user position is computed at the handset. The position can be retrieved by the end-user for many personal navigation applications or used for emergency situations.

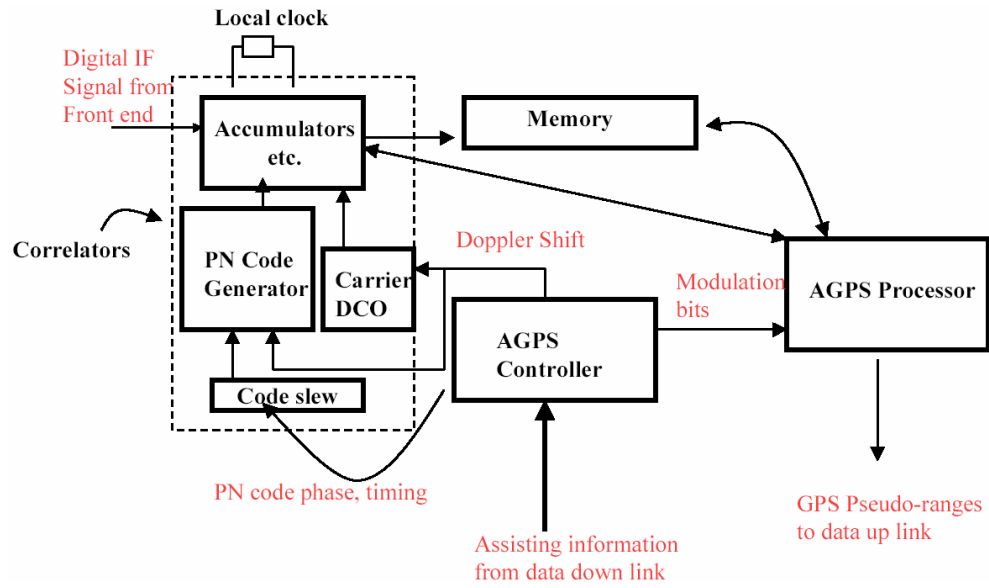


Figure 2.6: AGPS implementation [Source: Chiang, 2005]

CHAPTER 3: SIMULATION TESTS

This chapter discusses various simulation tests that have been carried out using the Spirent SIMGEN GSS 6560 simulator. Tests were carried out to determine the acquisition sensitivities of different test receivers, and to investigate the effects of different aided and unaided scenarios on the AGPS receiver. These receivers were later used to carry out numerous field tests in Chapters 4 and 5.

3.1 Test Measures

Before discussing the results of the simulation tests, the various test measures used, such as accuracy, availability, Time to First Fix (TTFF) and Carrier to Noise Ratio (C/N_0), are discussed.

Accuracy - is a measure that defines how close the location measurements are to the actual location of the mobile station (i.e., the true position). The closer the measured position is to the true position, the higher the accuracy. The Root Mean Squared error (RMS) is widely used for succinctly expressing the accuracy of a measurement.

Availability - is the percentage of the observation time in which successful position fixes are ascertained. Assuming that the measurements are good, availability and accuracy are inversely related; thus, higher availability implies a greater number of measurements and, hence, smaller errors. A successful position fix can be achieved only if sufficient measurements have been made.

Time To First Fix (TTFF) - is defined as the time that it takes for a GPS receiver to obtain a position fix in the context of a cold start, warm start or hot start. Usually the TTFF is measured in seconds; however, it was normalized with respect to the maximum value for all the tests. In the following paragraphs, the concepts of cold start, warm start and hot start are discussed. Acquisition tests were performed on the AGPS receiver using each of these three modes of receiver operation.

- **Cold Start** - occurs when the receiver has no acquisition information available and, therefore, must download the satellite ephemeris and perform a full search which includes a complete sky search of the 32 PRN codes as well as the C/A code/Doppler search space.
- **Warm Start** - occurs when almanac data are present and time and user position are known; however, the ephemeris is not present and, so, must be downloaded.
- **Hot Start** - is similar to conventional reacquisition following a brief outage; i.e., ephemeris, time and approximate position are known quite accurately and, hence, a narrower frequency or C/A-code phase search can be used to acquire very weak GPS signals.

Carrier to Noise (C/N_0) - is the most readily measured value of signal quality present at the input of the receiver. The C/N_0 is an instantaneous measure of the ratio of carrier power present to noise power density measured per Hertz of bandwidth. In theory, C/N_0

should not depend on the receivers used; however, the receiver computes C/N_0 and it depends on factors such as automatic gain control (AGC), SNR, integration time and correlation processes. Thus, the value is computed in various ways, depending on the particular receiver. The nominal value of C/N_0 is 44 dB-Hz; i.e. the nominal noise floor has a spectral density of -174 dBm /Hz and the LOS power is -130 dBm [Misra and Enge, 2001].

3.2 Acquisition Tests

Acquisition tests were carried out by using the Spirent GSS 6560 Simulator. The simulator consists of a PC (software), which is connected to two hardware RF simulators. The hardware simulators will be referred to as vehicles. Each vehicle is capable of simulating twelve channels, with each channel assigned to the satellite (SV) being simulated. A maximum of twelve satellites can be simulated where parameters such as satellite power can be changed. The simulator can also be used to simulate multipath effects, introduce tropospheric or ionospheric errors, interference effects and vehicle dynamics. Because each channel is assigned to a particular satellite, the terms “simulator channel” and “satellite”, refer, essentially, to the same thing and will be used interchangeably herein. This section discusses various acquisition tests that were conducted to meet the objectives discussed below.

3.2.1 Test Objectives

There have been many texts and research articles which have discussed the effects of satellite ephemeris and almanac, and the differences between cold, warm and hot starts.

The factors that are usually used to assess the similarities and differences are classified as either TTFF or acquisition sensitivity. In conducting the various simulation tests, the following objectives are pursued:

- Determine the acquisition sensitivity of each of the three SiRF receivers
- Determine the effects of different aiding scenarios such as satellite ephemeris and/or almanac on AGPS acquisition sensitivity.
- Investigate the effects of hot, warm and cold start on AGPS receiver operation in standalone mode. In this mode, the AGPS receiver receives no aiding data from the reference receiver.

3.2.2 Test Methodology

The following receivers (see Figure 3.1 and Figure 3.2) were used to carry out the test: SiRFLocTM AGPS, SiRFXtracTM HSGPS and SiRF Standard. The SiRF receivers (L1 C/A code 12 channels civilian) are based on the SiRF StarII architecture and will be referred to herein as AGPS, HSGPS and Standard, respectively. The AGPS receiver received aiding data from the reference receiver, known as the Time Transfer Board (TTBTM). Aiding data includes information such as satellite ephemeris, satellite almanac, frequency assistance, timing uncertainties and approximate user position (broken down into both horizontal and vertical uncertainties). The three receivers were later used to carry out various field tests which are discussed in Chapters 4 and 5.



Figure 3.1: Simulator Test Set-up

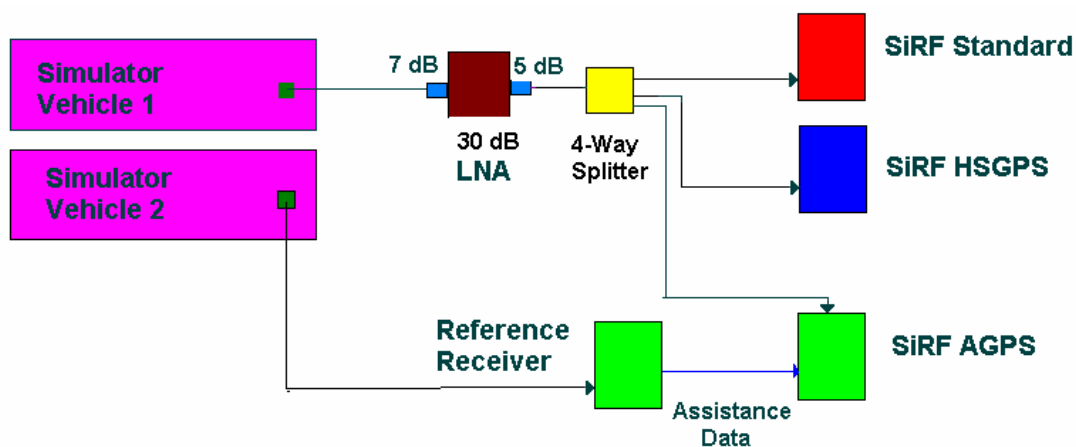


Figure 3.2: Simulator Set-up Schematic

Acquisition tests were conducted by decreasing the simulator power until the receiver stopped acquiring the GPS signals. The test receivers were connected to vehicle one, while the reference receiver was connected to vehicle two which had nominal signals (-130 dBm). The reference receiver provided reference data such as timing with an

uncertainty of 125 μ s; the approximate user position with horizontal and vertical uncertainties of 5 km and 150 m respectively; satellite ephemeris; almanac; and frequency assistance to the AGPS receiver. The simulator power is referenced to -130 dBm and can decrease to a level of -150 dBm. Acquisition tests that required power to decrease beyond -150 dBm were carried out using a 7 dB attenuator. Acquisition tests were started at -130 dBm and the power was decreased by one dB until the receiver stopped acquiring the GPS signals. The test receivers were then set to acquire signals, obtaining thirty position fixes for every acquisition test at each power level.

Further simulation tests have been carried out where the TTFF of the AGPS and HSGPS receivers were compared [Karunanayake, 2005b]. The TTFF was normalized with respect to maximum value during different acquisition tests. While carrying out tests using the AGPS receiver, one satellite was kept at -130 dBm. Simulation tests by Karunanayake *et al.*, [2004] have shown that one strong satellite can aid in acquiring the weaker satellite signals. During acquisition tests, PRN 6 was kept at -130 dBm, while the remaining satellite signals were decreased by a constant power increment until the receivers stopped acquiring; for example, PRN 6 may have a power of -130 dBm, while the remaining satellites may have signal power of -135 dBm.

Acquisition tests on the AGPS receiver in various start-up modes, such as hot or warm start, were carried out by maintaining the satellite power at a nominal level of -130 dBm for fifteen minutes so the complete satellite almanac and the satellite ephemeris could be downloaded. While carrying out acquisition tests in hot, warm and cold start modes, the

AGPS receiver did not receive any aiding data from the reference receiver. Acquisition tests were also carried out in which the AGPS received no satellite ephemeris or almanac data from the reference receiver. Further tests were carried out where the power of PRN 6 was altered, while the remaining satellites were kept at -150 dBm until the AGPS receiver stopped acquiring. Measurement errors such as multipath and ionospheric or tropospheric conditions were not simulated during all acquisition tests; consequently, various tests showed the effects of thermal noise – that is decrease in simulator power, implies that thermal noise will have significant effect on signal acquisition.

3.2.3 Test Results and Analysis

Table 3.1 shows the acquisition sensitivities of the three SiRF receivers that were tested, while

Table 3.2 shows the results obtained by carrying out a series of tests using various start modes and different aiding scenarios on the AGPS receiver. Figure 3.3 gives a comparison of AGPS performance using default aiding parameters, a hot start and no almanac. Figure 3.4 compares AGPS during warm start and without ephemeris. Figure 3.5 shows a comparison between AGPS in cold start mode and an HSGPS receiver. Figure 3.6 compares AGPS, AGPS without ephemeris assistance and AGPS using a cold start. Figure 3.7 shows the results obtained with variation of the signal power of the strong satellite (PRN 6). Finally Table 3.3 and

Table 3.4 give the positioning results obtained from the AGPS receiver using default aiding parameters at various simulator power levels. The position results were obtained using the C³NAV²TM and the SiRF internal solutions. The single-point positioning results (used for all position results in this research) were obtained by processing the raw

pseudorange data from the receiver using the C³NAV^G² software which is based on Least Squares (LSQ) [Petovello *et al.*, 2000].

Table 3.1: Acquisition Sensitivities of different SiRF Receivers

Receiver	Acquisition Sensitivity (dBm)
SiRF AGPS	-153
SiRF HSGPS	-140
SiRF Standard	-133

Table 3.2: AGPS- Acquisition Sensitivities with different Test Scenarios

Type of Start or Aiding	Acquisition Sensitivity (dBm)
Cold Start	-140
Warm Start	-142
Hot Start	-152
No Almanac	-152
No Ephemeris	-142

The simulations tests have shown that AGPS had higher sensitivity as compared to the other two receivers; that is, 13 dB better than HSGPS and 20 dB better than the standard receiver. The higher sensitivity of the AGPS receiver is due to the aiding data, as has been demonstrated by simulation tests using different types of starts or aiding data (without ephemeris or almanac). AGPS receiver with assistance data with hot start or

without almanac had similar sensitivity while AGPS in a cold start is like a HSGPS receiver, which does not have any prior knowledge of such things as GPS time or position. Similar results i.e. acquisition sensitivities for AGPS and HSGPS receivers were obtained during earlier simulations tests [Karunanayake *et al.*, 2004]. The next set of figures compares the TTFF for different test scenarios.

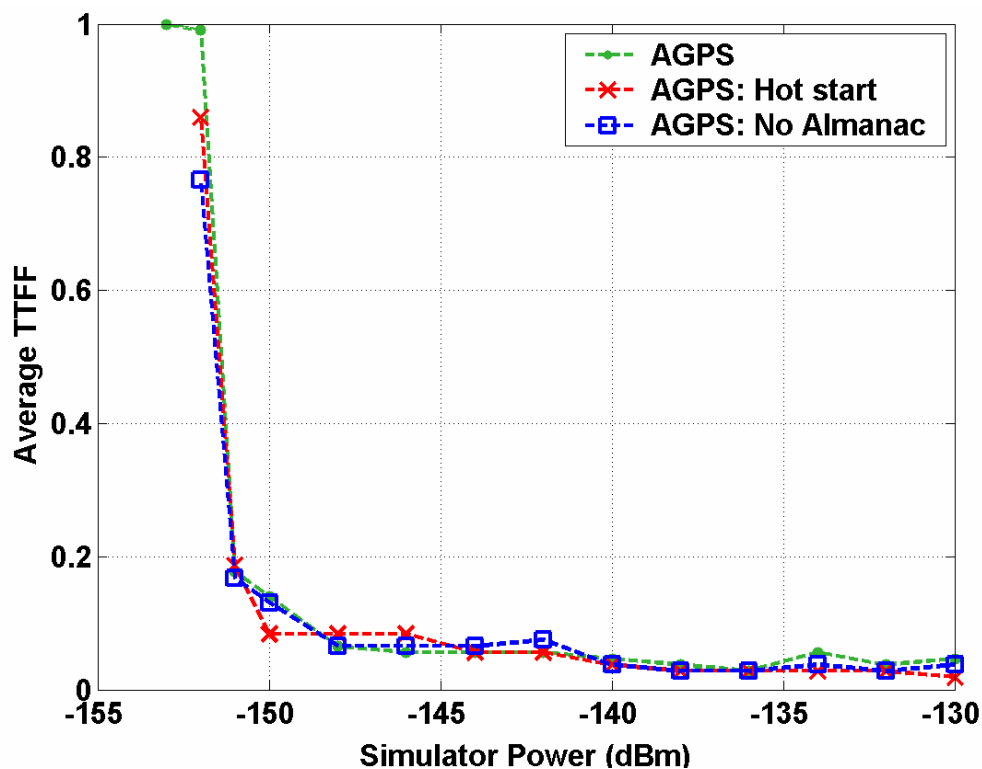


Figure 3.3: Comparison of AGPS test with Default Aiding, Hot start and without Almanac Assistance

The simulation tests of the AGPS unit using default aiding parameters, without almanac or hot start, showed similar performance in terms of TTFF, suggesting that the almanac data are not required since it provides a rather coarse estimate of satellite orbital parameters. By comparison, the test under hot start conditions showed that AGPS is

essentially comparable to an AGPS (without any adding) receiver that functions in the hot start mode.

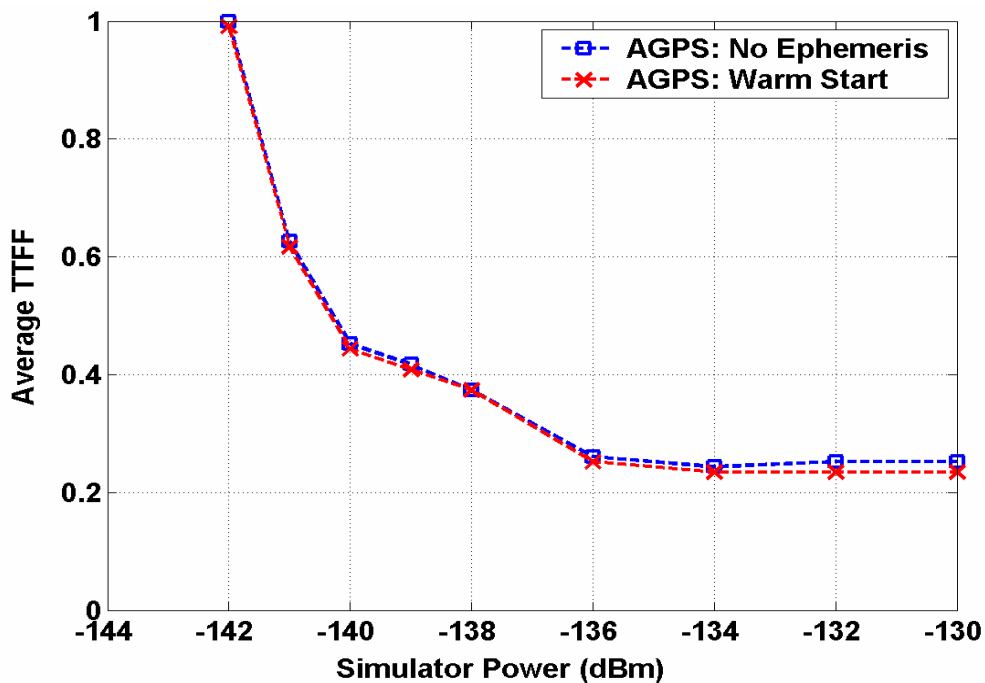


Figure 3.4: Comparison of AGPS Test with Warm Start and without Ephemeris Assistance

Simulation tests carried out using under warm-start and without ephemeris suggest, as discussed earlier, that the warm start is essentially carried out without ephemeris. The test also illustrated that ephemeris data can be used to enhance acquisition sensitivity; that is, an improvement of 11 dB (see Table 3.2) was realized with the incorporation of ephemeris aiding.

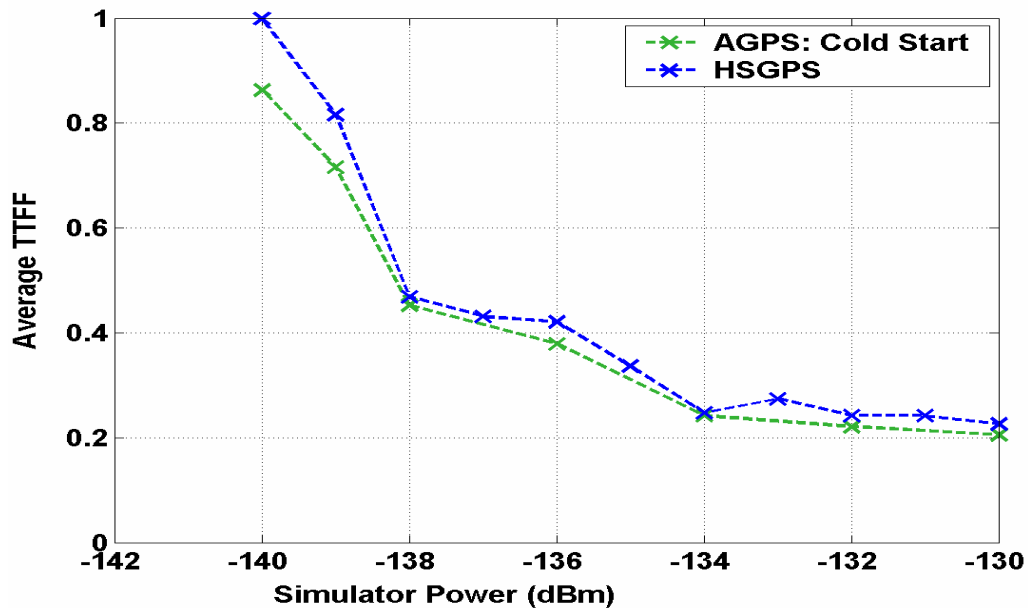


Figure 3.5: Comparison of AGPS with Cold start and HSGPS

The simulation test under cold start and HSGPS had similar results in terms of TTFF. The AGPS receiver without any assistance data (situation could occur when aiding is not available, e.g. when the wireless network is not available) behaves very much like a HSGPS receiver, so the performance is not severely degraded in an unaided state. This test confirms the statements made by SiRF (www.srif.com) where it stated that performance is not severely affect in an unaided situations. The HSGPS is like an unaided AGPS receiver therefore during cold-starts; both receivers are expected to have similar TTFF and acquisition sensitivities.

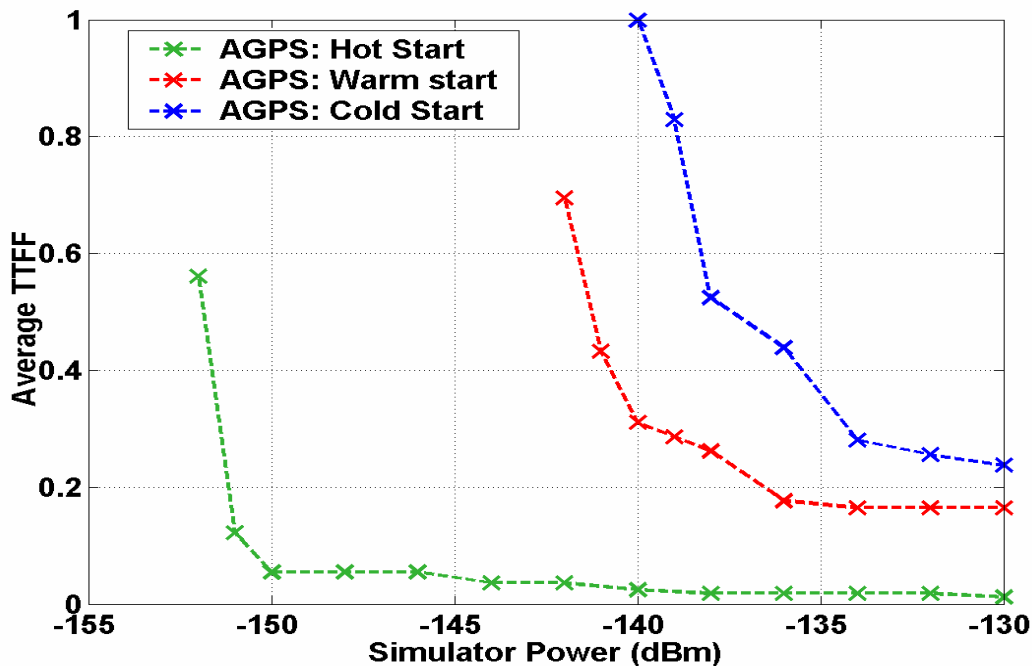


Figure 3.6: Comparison of Hot, Warm and Cold start performance of the AGPS Receiver

In tests of the AGPS in hot, warm and cold start modes, the hot start yielded the best performance in terms of acquisition sensitivity and TTFF since, in this scenario, the satellite ephemeris acquisition search space (C/A code and Doppler) are reduced, and because approximate GPS time and location are known fairly accurately. A receiver that lacks the ephemeris would take up to 30 s to acquire the required data; naturally, this results in longer TTFF – a distinction clearly visible when comparing the warm and hot start scenarios. It is typical for receivers to have non-volatile memory in which the recent ephemeris and or almanac data could be stored, along with the approximate location and GPS time; all of these data could be used to hasten the acquisition process. This shows that it is indeed difficult to demodulate the navigation data bits under very weak signal conditions (less than -142 dBm).

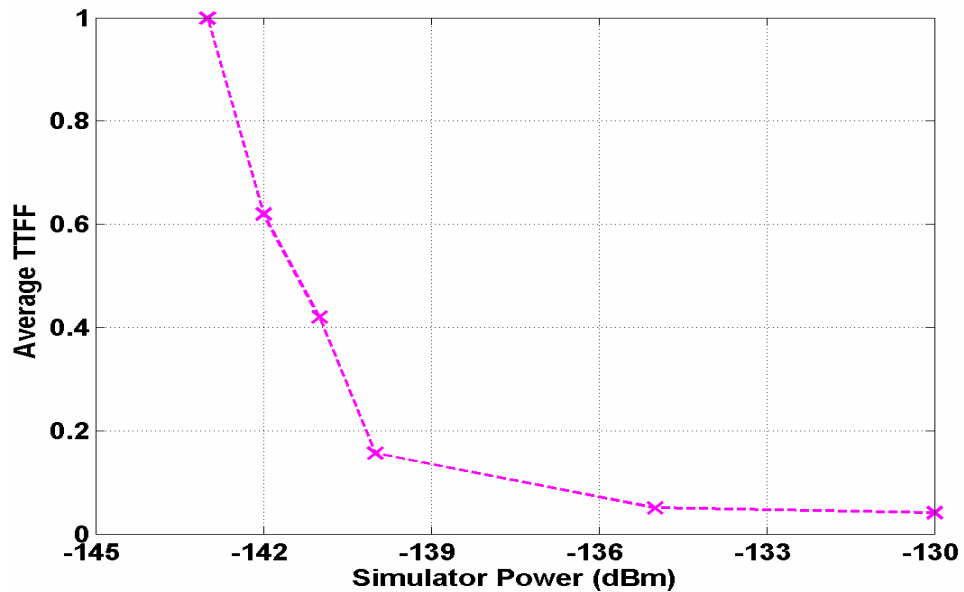


Figure 3.7: Changing the Power of the Strong Satellite Channel (PRN 06) for AGPS

It was noted that, when the power of the strong signal was varied, the power of the strong signal must be at least greater than -142 dBm before it can be used to internally aid the receiver in acquiring the weaker GPS signals which were kept at -150 dBm.

Table 3.3: AGPS Position Results Using Least Squares

Simulator Power (dBm)	Five Position Fixes			Thirty Position Fixes		
	2D RMS Error (m)	Mean # of Satellites	Mean HDOP	2D RMS Error (m)	Mean # of Satellites	Mean HDOP
-130	3.6	8.2	1.2	3.8	7.9	1.4
-136	6.1	8.0	1.0	3.4	8.2	1.0
-142	10.9	8.5	1.0	14.5	7.8	1.1
-148	41.1	4.5	3.9	45.5	4.7	3.0
-152	37.0	4.0	5.1	34.8	4.0	5.3

Table 3.4: AGPS Position Results SiRF Internal Solution

Simulator Power (dBm)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
-130	7.9	9.2	11.7	8.6
-136	4.7	8.0	3.9	8.1
-142	14.7	7.8	15.0	7.8
-148	42.1	4.2	42.3	4.0
-152	38.8	3.3	35.2	3.3

Finally the positioning results showed that horizontal positioning accuracy degraded with lower signal power. Higher thermal noise would result in larger measurement errors. The number of satellites used in a given scenario decreased with lower signal power, resulting in poor geometry (larger HDOP) which also contributed to degradation in horizontal positioning accuracy. Earlier research using the SiRF AGPS receiver has shown similar results where the position accuracy degraded with lower simulated signal power [Garin *et al.*, 2002].

3.3 Chapter Summary

- The AGPS showed higher acquisition sensitivity as compared to the other two receivers, 13 dB better than HSGPS and 20 dB better than the standard receiver making it suitable for many applications where GPS signals can be as low as -150 dBm.
- Satellite ephemeris data is important for signal acquisition in terms of sensitivity (11 dB better with ephemeris aiding) and a shorter TTFF, while almanac data are not required for signal acquisition.

CHAPTER 4: FIELD TESTS: ACQUISITION

Acquisition tests were carried out under static conditions in a suburban environment, a residential concrete garage, a speed-skating track and a concrete basement. Most of the acquisition tests were carried out by using the AGPS receiver to determine the effects of aiding data, which are listed in the test objectives in Section 4.1. Limited field tests were carried out using the HSGPS receiver and are given in Section 4.4. The chapter begins by giving a brief description of the test setup and methodology. A description of different field test conditions is given followed by a discussion of the results that were obtained to meet the goals of this chapter. The results were analyzed using factors such as TTFF, number of satellites tracked, C/N_0 and position accuracy. The different environments are also cross-compared using the above factors. The chapter closes by discussing some conclusions that can be drawn from various acquisition tests that were carried out in different field test sites.

4.1 Test Objectives

Similar to Chapter 3, acquisition tests were conducted using the two receivers (HSGPS, and AGPS). The majority of the tests were carried out using the AGPS receiver where different aiding scenarios were investigated and this is captured in the following objectives:

- Investigate the acquisition performance of the HSGPS receiver

- Determine the effects of satellite ephemeris or almanac on AGPS signal acquisition
- Quantify the effects of different timing uncertainties on AGPS signal acquisition
- Quantify the effects of different horizontal position uncertainties on AGPS signal acquisition
- Compare the acquisition performance across different field test environments

4.2 Test Methodology

The AGPS receiver was used to carry out all the acquisition tests under different field test conditions and the test methodology remained the same regardless of the environment. The test setup is shown in Figure 4.1. It shows the AGPS receiver and the reference receiver, where the reference receiver is connected via a 30 m cable to a reference antenna that has clear LOS signals. The reference receiver provides aiding data such as satellite ephemeris, almanac, approximate GPS time, and approximate user position with horizontal and vertical uncertainties. The AGPS receiver is connected to a micropatch antenna. During each test, more than 20 trials were carried out and 30 position fixes were obtained for each trial. After 30 position fixes logged at 1 Hz, the AGPS receiver would restart from a cold start for the next trial. Each trial was as small as 5 s for a sub-urban test or as large as 150 s for the concrete basement test. In the case that the AGPS receiver was unable to obtain a position fix, it would restart after 300 s. The latter situation can be defined as acquisition failure. The acquisition test is carried out in a similar manner using

the HSGPS receiver with the exception that it does not receive any aiding data. The aiding scenarios for the AGPS receiver are:

- 1) Timing Uncertainties of:
 - Precise time of 10, 50, 125, 250 or 500 μs
 - Coarse time of 1 ms, 1, 2 and 10 s
- 2) Horizontal Position Uncertainties of:
 - 5, 20, 50, 100 or 350 km
- 3) No Ephemeris
- 4) No Almanac

Timing uncertainty better than one millisecond is known as *precise* time aiding, while uncertainty of one millisecond or worse is referred to as *coarse* time aiding. Each aiding scenario is further described in Table 4.1. During the acquisition tests, the approximate user position of the receiver was set to one of the surveyed points at the University of Calgary campus. The position results were obtained using the C³NAV² and compared with the receiver's internal solution (similar to Chapter 3). The receiver would typically track three satellites under weak/degraded signal environments such as the speed-skating track and the concrete basement (significant signal blockage) [Lachapelle *et al.*, 2003]. Height constraints were used to increase the C³NAV² position solution availability. Height fixing requires only three rather than four satellites to solve for the three unknowns, 2-D position and time. Height assistance can be given to the AGPS receiver (obtained from digital maps); therefore, this is a reasonable assumption. This is also similar to altitude aiding that is used in the SiRF receivers to obtain position solutions in

weak signal conditions [Phatak *et al.*, 2004]. The SiRF receivers also carry out Kalman filtering to obtain the position solution. Kalman filtering uses previous measurements to predict the current position estimate, identifies outliers (large pseudorange errors), and is, therefore, able to provide a better position accuracy when compared to least squares estimation [Syed, 2005].

Table 4.1: Description of Different Aiding Scenarios

Scenario	Timing Uncertainty (μs)	Position Uncertainty		Ephemeris	Almanac
		Horizontal (km)	Vertical (m)		
1	Varied*	5	150	Yes	Yes
2	125	Varied*	150	Yes	Yes
3	125	5	150	No	Yes
4	125	5	150	Yes	No

*When timing or position uncertainty was varied, for example in scenario one, the horizontal position uncertainty was kept constant while the timing uncertainty was changed (10 μs to 10 s).

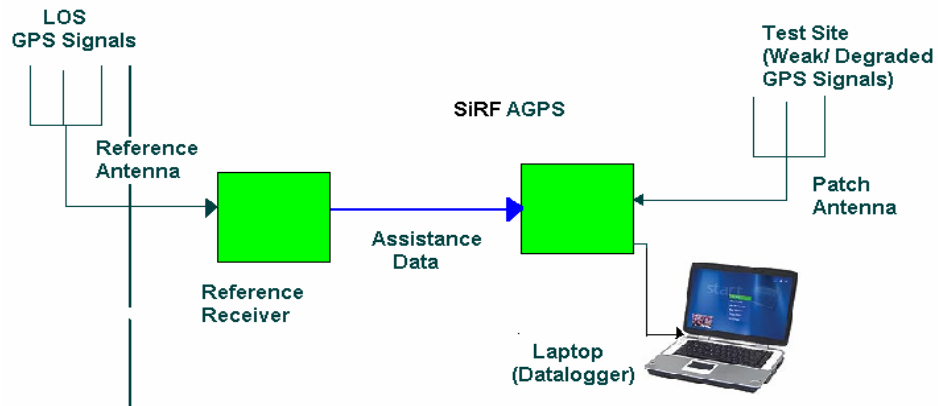


Figure 4.1: Field Test Set-up for the AGPS Receiver

4.3 Field Test Environments

Cell phones are designed to work anywhere all the time. Field tests were carried in different test sites where people may be using the cell phone for different purposes such as emergency E-911 calls, personal navigation or commercial LBS applications. The tests were conducted under many different conditions that would present various challenges such as signal blockage or varying degree of signal attenuation. Acquisition tests were carried out in the suburban environment, residential garage, inside a speed-skating track and a concrete basement. The different test sites were chosen to determine the effects of various factors on AGPS signal acquisition.

4.3.1 Suburban Environment

Acquisition tests were carried out at surveyed points at the University of Calgary campus on October 9, 2004. The test site is shown in Figure 4.2, Figure 4.3 and Figure 4.4. There was a tall glass building on the east side, a smaller concrete building on the west and a

glass walkway to the north. The southern side is unhindered with the exception of some coniferous trees on the southwest side. The AGPS receiver was connected to a microstrip patch antenna, which was placed at the test point as shown in Figure 4.2.



Figure 4.2: Receiver Setup for the Suburban Field Test

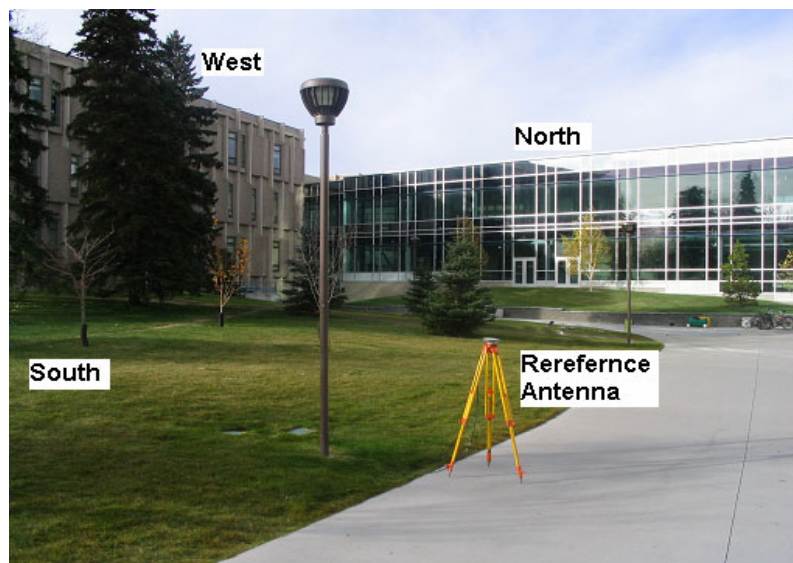


Figure 4.3: Reference Antenna and the Surrounding Site

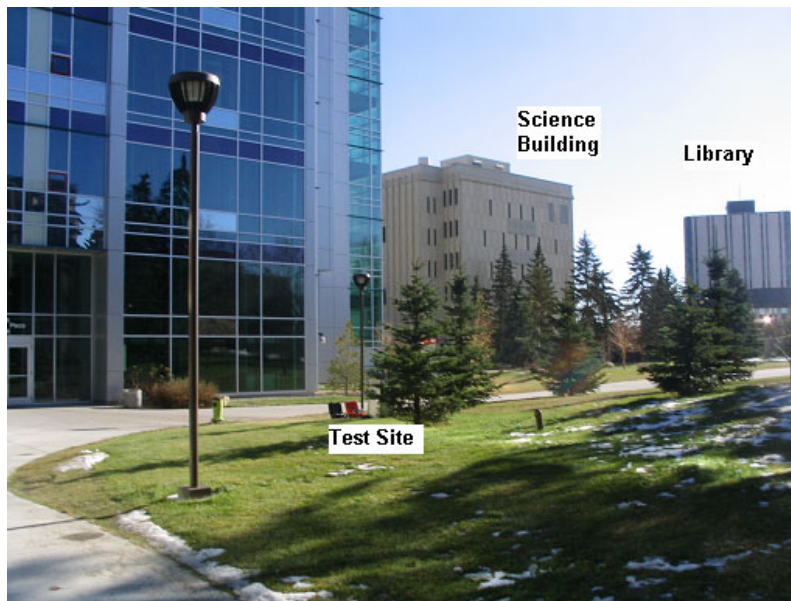


Figure 4.4: Surrounding Area for the Test Site

4.3.2 Residential Garage

Acquisition tests were carried out in a residential garage with dimensions of 9 x 6 m x 2.5 m (located within 5 km from the University of Calgary) on December 9, 2004. The garage setup, as shown in Figure 4.5 and Figure 4.6, is located underneath the living room of a house and its walls are made of wood and concrete. The door facing the east side was made of wood, while the wall facing the south side was partially constructed of wood. The remaining two walls facing west and north were made of concrete. The garage door was closed during all acquisition tests. The reference antenna was connected to the reference receiver and located outside the garage. The AGPS receiver was connected to a microstrip patch antenna, which was placed at a surveyed point inside the garage.

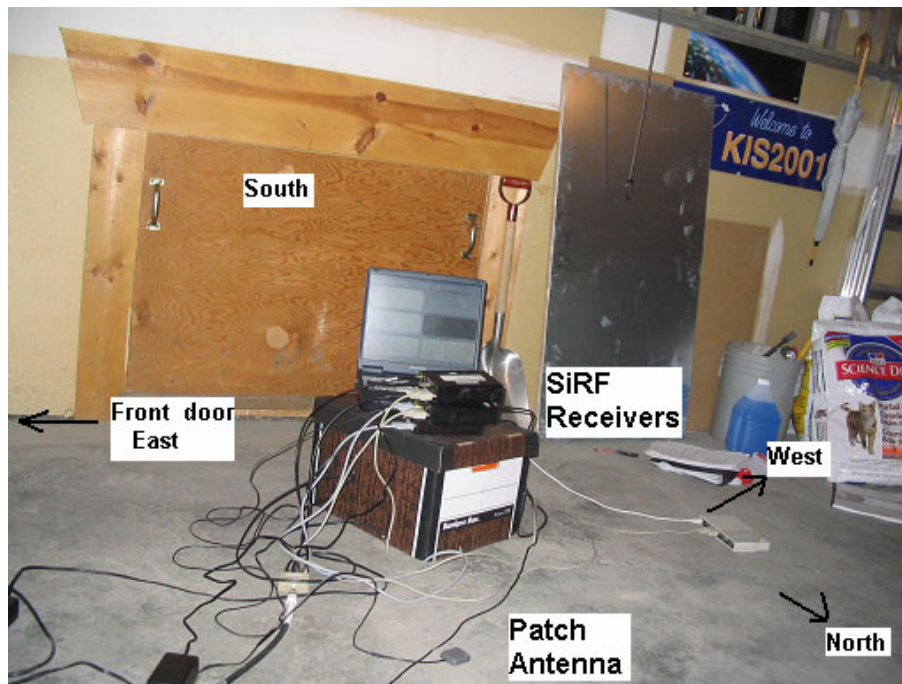


Figure 4.5: Test Setup for the Garage Test



Figure 4.6: Surrounding Area for the Garage Test

4.3.3 Speed-Skating Track

Acquisition tests were carried out on December 13, 2004 in a speed-skating track (see Figure 4.7, Figure 4.8 and Figure 4.9), also known as the Olympic Oval, which is located on the University of Calgary campus. The indoor speed-skating track is made of concrete beams with approximate dimensions of 198 m x 80 m x 20 m. The roof is constructed with corrugate steel (interior) and porcelain panels (exterior). The reference receiver was connected to the reference antenna, which was placed near the window. The AGPS receiver was connected to the microstrip patch antenna that was placed at a surveyed point inside near the window because it was easier to obtain the position fix due to better satellite availability.



Figure 4.7: Outside View of the Speed-skating Track

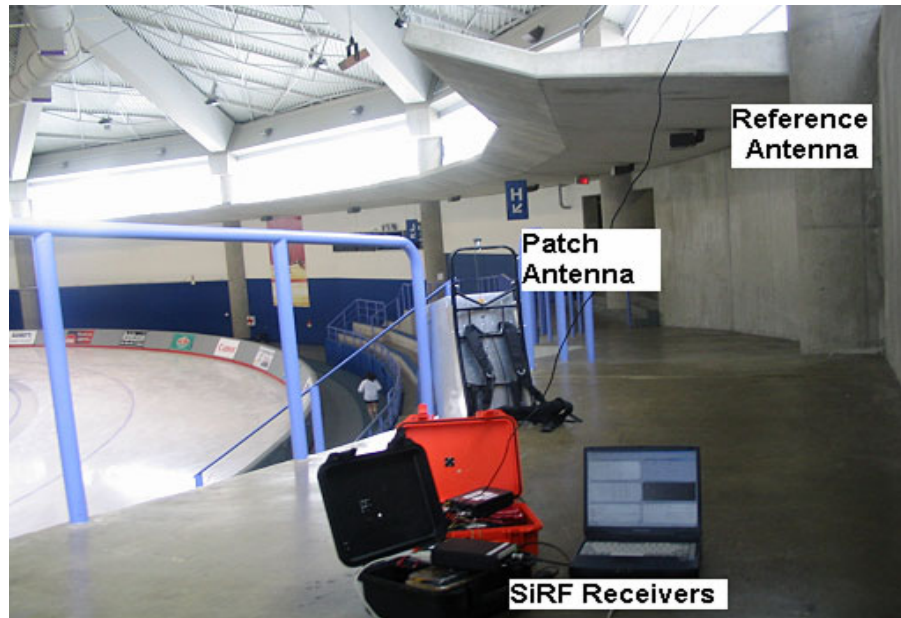


Figure 4.8: Receiver Setup for the Speed-skating Track Test



Figure 4.9: Inside the Speed-skating Track

4.3.4 Concrete Basement

An acquisition test as shown in Figures 4.10 and 4.11 was carried out in a concrete basement with an underground pit inside. The dimensions of the pit are 7 m \times 13 m \times 4.5 m and it is contained within a High Bay of 16 m \times 14 m \times 16 m. The test site is located inside the CCIT building at the University of Calgary campus and the test was conducted on November 30, 2004. The roof consists of a metal deck and steel structure. There is a door located on the north side that is made of wood and on the northwest side there are two small windows. The remaining three sidewalls were made of reinforced steel concrete. Similar to the previous tests, the AGPS receiver was connected to a microstrip patch antenna that was placed at a surveyed point inside the pit, while the reference antenna was kept outside.

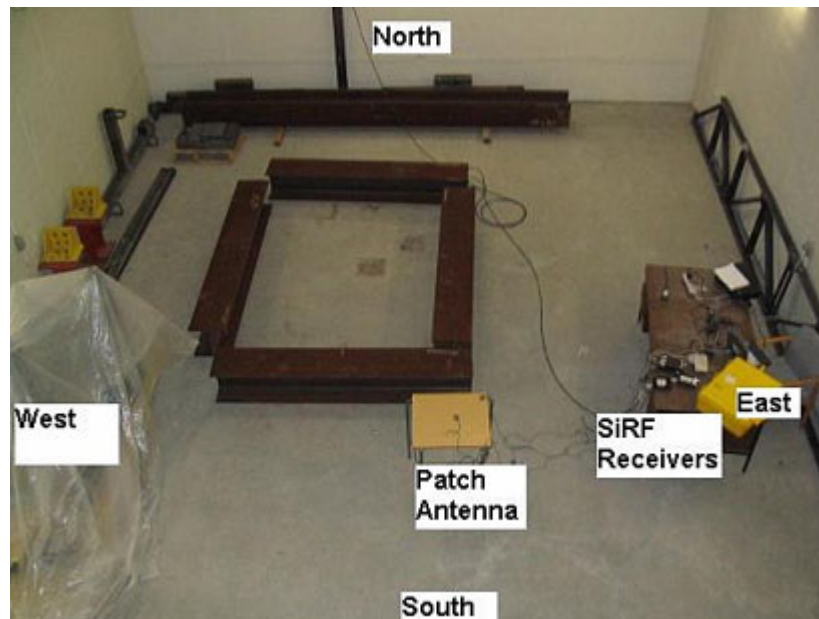


Figure 4.10: Receiver Setup for the Concrete Basement Test



Figure 4.11: Surrounding Area for the Concrete Basement Test

4.4 High Sensitivity GPS Receiver

Limited field tests were carried out using the HSGPS receiver in the residential garage. A HSGPS receiver was also used with the reference antenna on the CCIT roof. The results obtained from the roof were similar to the suburban environment in terms of the TTTF because of nominal GPS signals being present (see Section 4.8). The position results are shown in Table 4.2 and Table 4.3. The position accuracy using thirty fixes was better for the roof test (5.3 m) compared to the garage test (18.3 m). The roof had better satellite availability and stronger LOS signals which resulted in better accuracy. The HSGPS receiver could not acquire and/or obtain a position fix inside the speed-skating track and the concrete basement because of very weak signals (< 25 dB-Hz). The TTFF for the HSGPS receiver was ten times more than that of the AGPS receiver on the roof

and six times more in the residential garage. The TTFF in the residential garage test was seven times longer when compared to the roof test. Simulation tests in Chapter 3 and Karunanayake *et al.*, [2004] have shown that HSGPS was unable to acquire below -140 dBm which was further confirmed during field tests. The TTFF increased when the simulator power decreased. Research using HSGPS has shown longer TTFF with weaker signals during field tests [Shewfelt *at al.*, 2001]. The position accuracy also degraded with weaker GPS signals. Further field tests using the AGPS receivers were carried out and results are discussed next to illustrate the importance of assistance data.

Table 4.2: HSGPS Receiver- Position Results Using LSQ Internal Solution for Roof and Garage Tests

Type of Environment	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
Roof	5.9	8.5	5.5	8.7
Residential Garage	19.9	5.5	18.3	4.3

Table 4.3: HSGPS Receiver- Position Results Using SiRF Internal Solution for Roof and Garage Tests

Type of Environment	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
Roof	5.1	8.1	4.7	8.2
Residential Garage	22.8	4.3	18.9	3.9

4.5 Timing Assistance

This section will discuss results obtained from various aiding scenarios involving different timing accuracies and the AGPS system. Timing assistance which was described in Section 2.4.3 can be used to narrow the acquisition search space. The tests aim to illustrate the effects of accurate timing assistance under different test conditions. Syrjärinne and Kinnari, [2002] have shown using Location Measurement Units (LMUs), that the timing accuracy of a GSM or UMTS network can range from a few microseconds to hundreds of microseconds.

4.5.1 Suburban Environment

The TTFF results for precise and coarse time aiding are shown in Figure 4.12 and Figure 4.13. The position results that were obtained using the C³NAV² and the SiRF internal solution are shown in Table 4.4, through to Table 4.7.

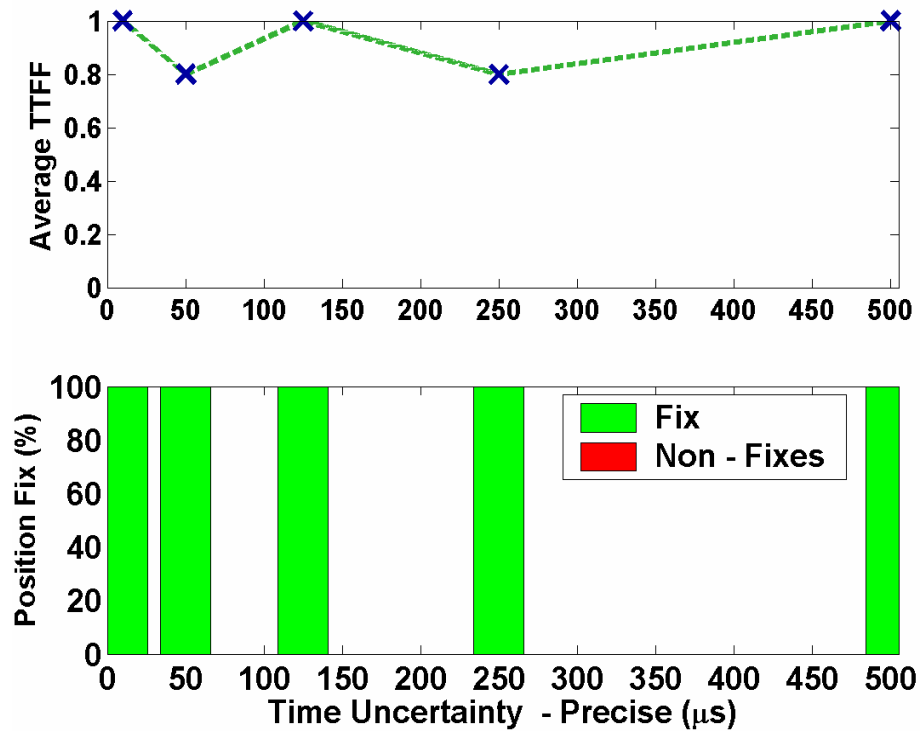


Figure 4.12: Precise Time Aiding - Normalized TTFF and Position Fixes for Suburban Test

Table 4.4: Precise Time- Position Results Using LSQ for Suburban Test

Precise Time (µs)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
10	40.4	5.6	35.4	6.1
50	47.6	4.9	39.3	5.9
125	60.8	5.6	60.1	5.7
250	44.3	6.2	47.3	6.4
500	57.5	5.0	54.5	6.4

Table 4.5: Precise Time Aiding - Position Results Using the SiRF Internal Solution Suburban Test

Precise Time (μ s)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
10	32.3	5.1	28.5	6.3
50	30.3	5.2	27.3	6.2
125	61.3	5.3	69.3	6.1
250	28.5	5.3	25.1	7.0
500	26.4	5.3	24.2	6.5

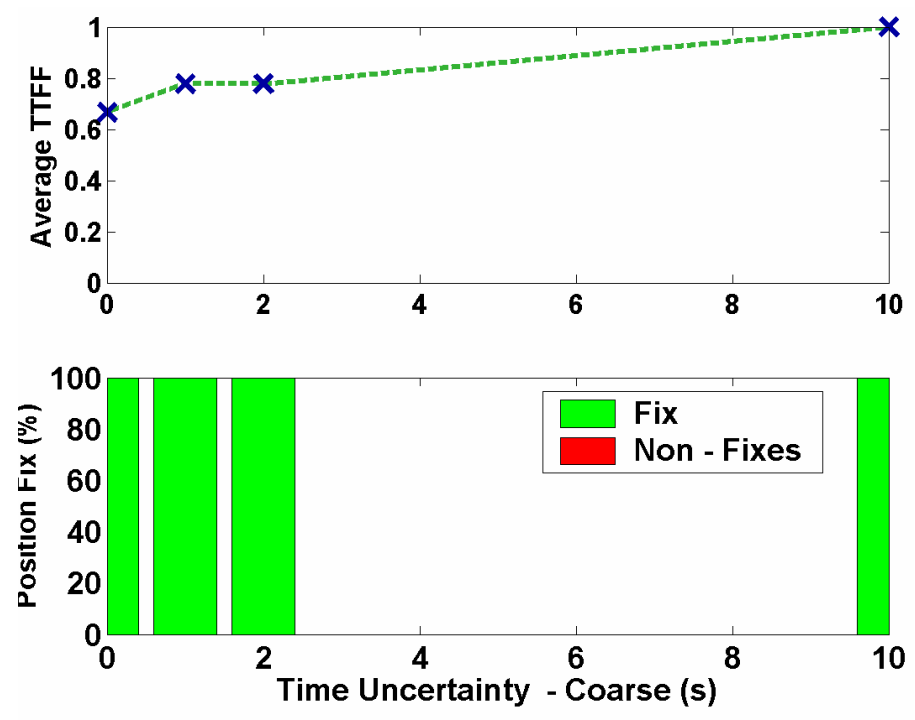


Figure 4.13: Coarse Time Aiding- Normalized TTFF and Position Fixes for Suburban Test

Table 4.6: Coarse Time Aiding- Position Results Using LSQ for Suburban Test

Coarse Time	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites Tracked	2D RMS Error (m)	Mean # of Satellites Tracked
1 ms	25.1	5.3	24.3	6.4
1 s	26.4	5.2	25.4	6.6
2 s	24.4	5.1	25.9	6.9
10 s	29.1	5.2	22.1	6.7

Table 4.7: Coarse Time Aiding- Position Results Using SiRF Internal Solution for Suburban Test

Coarse Time	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites Tracked	2D RMS Error (m)	Mean # of Satellites Tracked
1 ms	24.8	5.4	24.8	7.1
1 s	25.4	5.5	25.2	7.4
2 s	24.8	5.3	24.7	7.6
10 s	28.4	5.3	28.5	6.8

There was no variation or trends in the TTFF with varying timing uncertainty; however, the TTFF was increased by 50% when the timing uncertainty changed to coarse time from precise time aiding. The AGPS receiver was able to obtain a position fix for 100% of the time during all the tests. The AGPS receiver used at least five or more satellites illustrating very good availability. The SiRF internal compared to the LSQ solution showed better position accuracy in terms of position accuracy because it uses Kalman filtering. The position results with coarse time aiding showed similar position accuracies using the two methods. The position results (SiRF solution) for five fixes, with the exception of 125 μ s, became better with increasing timing uncertainty (precise time aiding). The position accuracy with coarse time aiding showed no trend. Generally the accuracy for thirty position fixes was better than five position fixes because more satellite were used. More satellite leads to better geometry or HDOP, which results in the improvement of user-position accuracy. Acquisition is unpredictable in nature; possibilities exist where some trials may have large position errors (satellites possibly tracking reflected signals). This was the case for 125 μ s where there were large errors which degraded the horizontal position accuracy. The position errors were as large as 175 m for 125 μ s, while the maximum position error for 250 μ s was 58 m. The TTFF increased with increasing time uncertainty with coarse time aiding. However, the difference between the maximum and minimum TTFF was only 2 s. This difference becomes significant when TTFF was normalized because the average TTFF for most of the tests was 4 s.

Similar results in terms of trends in TTFF and position results were obtained when simulation tests with varying time uncertainties (precise time) were carried out under nominal signal conditions (-130 dBm) [Karunanayake, 2005b]. Simulation tests had also shown that the position accuracy for the first few fixes was worse and improves as the code tracking loops start getting better estimates of the position solution. The position accuracy was better with increasing timing uncertainty. Garin *et al.*, [2002] had also shown that first fix at different signal levels was worse than the second fix.

The suburban environment, as shown in Figure 4.14, reflects that of nominal signal conditions. The satellites generally had a C/N_0 that was greater than 35 dB-Hz, however there were some weak signals (28 dB-Hz) that may have been attenuated due to signal masking from coniferous trees. The C/N_0 PDF (at different elevations) plot was obtained using all the data from the 125 μ s test. A similar method was used to obtain the C/N_0 plot for the other environments. PRNs 6 and 9 were lower elevation satellites located on the south side and were attenuated by trees. The remaining three satellites were higher elevation and therefore did not suffer from signal blockage or attenuation. Similar results were obtained in Hu, [2006] and MacGougan, [2003].

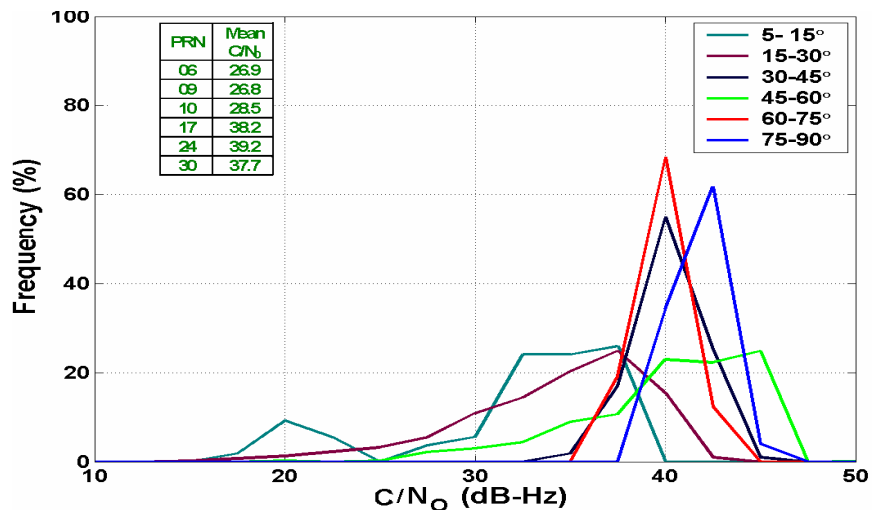


Figure 4.14: C/N₀ PDF for the Suburban Test

4.5.2 Residential Garage

The results for the residential garage for precise and coarse time aiding are shown in Figure 4.15 and Figure 4.16. The position results that were obtained using C³NAV² and the SiRF internal software are shown in Table 4.8 through Table 4.11.

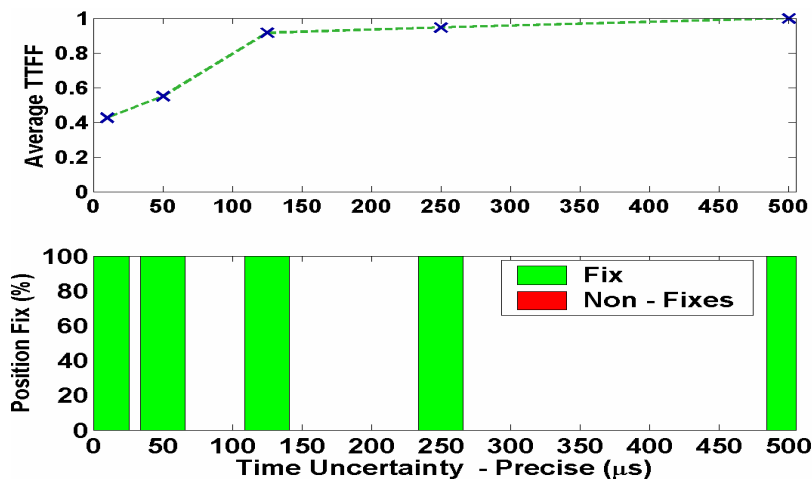


Figure 4.15: Precise Time Aiding – Normalized TTFF and Position Fixes for Garage Test

Table 4.8: Precise Time Aiding - Position Results Using LSQ for the Garage Test

Precise Time (μ s)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites Tracked	2D RMS Error (m)	Mean # of Satellites Tracked
10	16.8	3.7	23.1	4.1
50	27.2	3.8	23.0	3.9
125	19.0	3.5	23.5	4.5
250	15.4	3.6	29.7	5.6
500	16.5	4.2	21.9	4.1

Table 4.9: Precise Time Aiding - Position Results Using SiRF Internal Solution for Garage Test

Precise Time (μ s)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites Tracked	2D RMS Error (m)	Mean # of Satellites Tracked
10	25.4	5.3	19.4	5.2
50	24.9	5.3	25.5	5.2
125	18.8	5.1	20.7	5.0
250	18.3	5.0	23.9	5.4
500	17.7	5.2	26.9	5.3

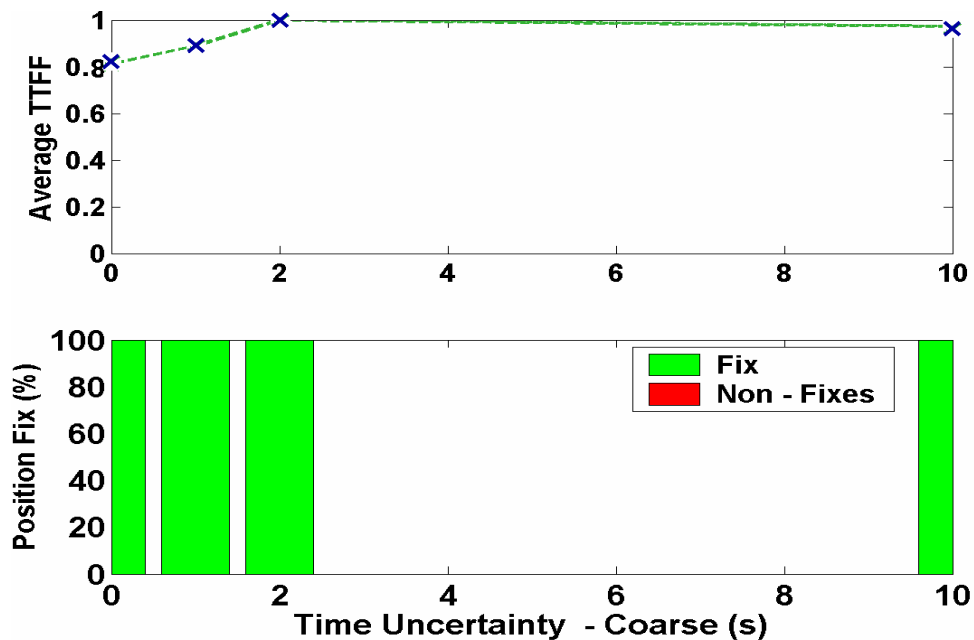


Figure 4.16: Coarse Time Aiding – Normalized TTFF and Position Fixes for Garage Test

Table 4.10: Coarse Time Aiding - Position Results Using LSQ for Garage Test

Coarse Time	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
1 ms	23.5	3.1	20.2	3.1
1 s	31.9	3.8	27.6	4.1
2 s	30.7	4.2	20.2	3.8
10 s	18.4	4.5	24.1	4.4

Table 4.11: Coarse Time Aiding - Position Results Using SiRF Internal Solution for Garage Test

Coarse Time	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
1 ms	33.4	5.3	26.4	5.2
1 s	36.8	5.3	28.7	3.9
2 s	21.2	5.3	21.8	4.0
10 s	40.8	5.3	30.0	4.6

Unlike the suburban test that showed little correlation between precise timing uncertainty and the TTFF, the garage test showed an increase in TTFF when the timing uncertainty or the position uncertainty was increased. When the timing uncertainty was changed from 50 μ s to 125 μ s, the TTFF increased by 50%. The AGPS receiver was able to obtain a position fix 100% of the time during all the tests. Simulation tests by Karunanayake *et al* [2004] under similar conditions (> -136 dBm) showed longer TTFFs with increasing timing uncertainty. Simulation tests also showed 100% position fix at similar signal levels. Better timing accuracy reduces the acquisition search space which is important to acquire signals under weak field signal conditions. This issue of timing accuracy is not the factor in the sub-urban environment because of nominal GPS signals. Similar to the suburban test, when the timing uncertainty was changed from precise time to coarse time, the TTFF increased by 50%. However, the TTFF did not show any trend when the coarse time aiding was changed. The position results from the C³NAV² and SiRF internal had similar trends. The position accuracy of five position fixes was better than the thirty

fixes. This was opposite of what was observed in the suburban environment. During thirty fixes the number satellites used would vary (three to five), which would change the HDOP, and, in fact, could degrade the position accuracy. For example, time aiding of 125 μ s showed a maximum position error for five fixes of 37 m, while it was 176 m for thirty fixes. The position accuracy (SiRF solution for five fixes) improved with decreasing timing uncertainty (precise) but did not show any trends with coarse time aiding. These trends in position results were similar to the suburban environment.

Most of the signals experienced signal attenuation (see Figure 4.17) but some satellites such as PRNs 19 and 27 had strong signals (C/N_0 greater than 30 dB-Hz). The weakest satellite (PRN 29) had a C/N_0 of 21.4 dB-Hz. Higher signal attenuation would be caused by building materials such as concrete walls while some stronger signals may experience moderate attenuation from wooden surfaces. The trends between elevation is not clear unlike the sub-urban test were higher elevation satellites had stronger signals. Other factors such as different surfaces (wood or concrete), that is signal attenuation determines the signal strength.

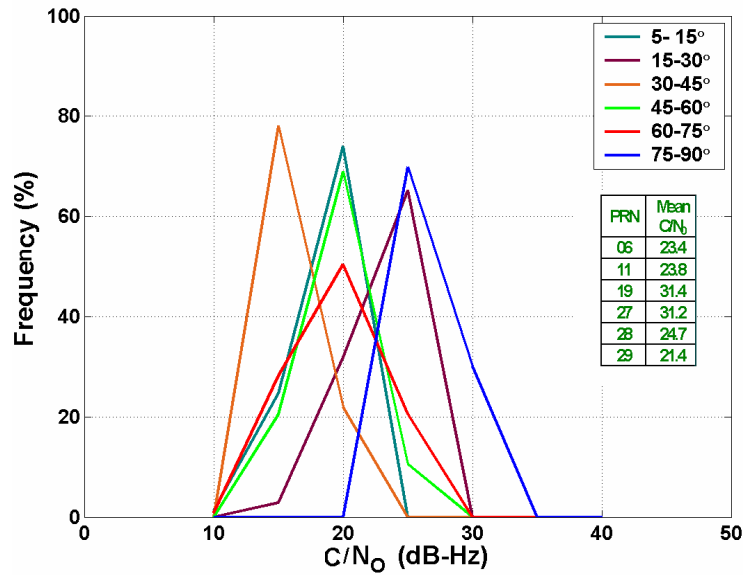


Figure 4.17: C/N_0 PDF for the Residential Garage Test

4.5.3 Speed-skating Track

The speed-skating track results using different timing uncertainties are shown in Figure 4.18. The position results that were obtained using C^3 NAV G^2 and the SiRF internal software are shown in Table 4.12 and Table 4.13. Acquisition tests with coarse aiding were carried out using 1 ms. Tests with other timing levels could not be carried out on that day so they are not shown.

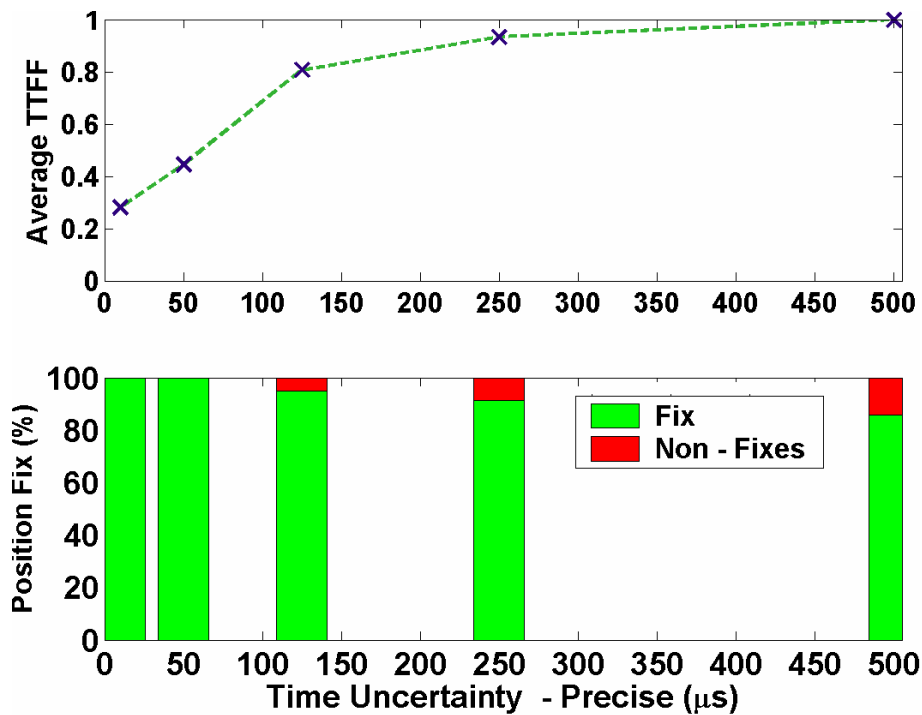


Figure 4.18: Precise Time Aiding – Normalized TTFF and Position Fixes for Speed-skating Test

Table 4.12: Precise Time - Aiding Position Results Using LSQ for Speed-skating Test

Precise Time (μs)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
10	55.1	3.6	46.7	3.8
50	49.0	3.5	45.4	3.4
125	58.4	3.9	59.1	3.9
250	46.6	3.3	43.3	3.2
500	57.5	4.1	47.9	4.0

Table 4.13: Precise Time Aiding - Position Results using SiRF Internal Solution for Speed-skating Test

Precise Time (μs)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
10	54.0	5.3	57.7	4.7
50	57.8	5.3	48.8	4.6
125	45.7	3.7	37.5	3.8
250	45.5	5.8	49.3	3.4
500	60.1	5.3	59.8	3.4

Similar to the residential garage test results, the results here with precise time aiding show a definite trend. For example, when the timing uncertainty was changed from 50 to 125 μs , the TFFF increased by 83%. The results are similar to the residential garage tests illustrating the importance of good timing accuracy. The position fix success rate also decreased with increasing timing uncertainty. The success rate was 100% for 10 μs and 50 μs but decreased from 95% to 85% when the timing uncertainty was changed from 125 μs to 500 μs . These results are different from those of the residential garage which had 100% success rate for all the tests. Similarly, simulations tests carried out at weak power (-152 dBm) had a success rate of 80% [Karunanayake, 2000b]. Acquisition is difficult to carry out with weaker signals especially when the timing uncertainty is increased, since weaker signals prolong the search process (longer integration time). In many instances the receiver is unable to find the GPS signals before the required time of

300 s. When the time aiding was changed from precise to coarse, 500 μ s to 1 ms, the TTFF increased by 150%. The position accuracy using C³NAV² was better for five position fixes than thirty fixes (similar to the suburban environment). The SiRF solution did not show any trend between five and thirty position fixes. Unlike, the other two environments the position accuracy did not show any trend with precise time aiding.

The speed-skating track suffered from significant blockage which reduced satellite availability (usually less than four satellites). Therefore, the solution using C³NAV² was obtained by fixing the height. Height fixing would require three rather than four observations to solve for the three unknowns (horizontal position and time). Generally the AGPS receiver would use anywhere from three to five satellites to obtain a position fix. The speed-skating track had very weak signals (see Figure 4.19). Most of the satellites had signal strength less than 25 dB-Hz. The signals are highly attenuated by materials such as concrete walls. However, there was one satellite with strong signals (possibly entering through a glass window, PRN 5 with C/N₀ of 29.9 dB-Hz).

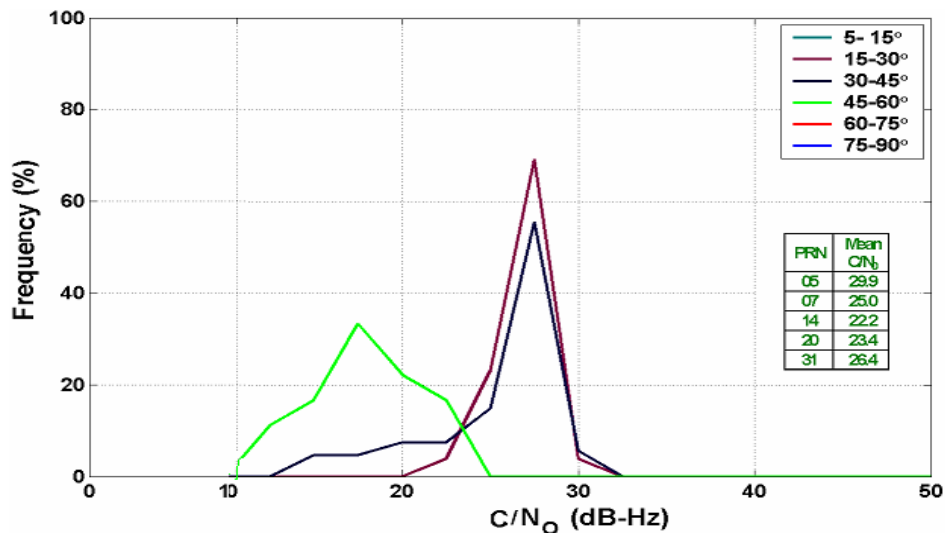


Figure 4.19: C/N₀ PDF for the Speed-skating Track Test

4.5.4 Concrete Basement

The concrete basement results are shown in Figure 4.20 and Figure 4.21 and the position results obtained using the C³NAV^{G2} and SiRF internal solutions are shown in Table 4.14 through Table 4.17.

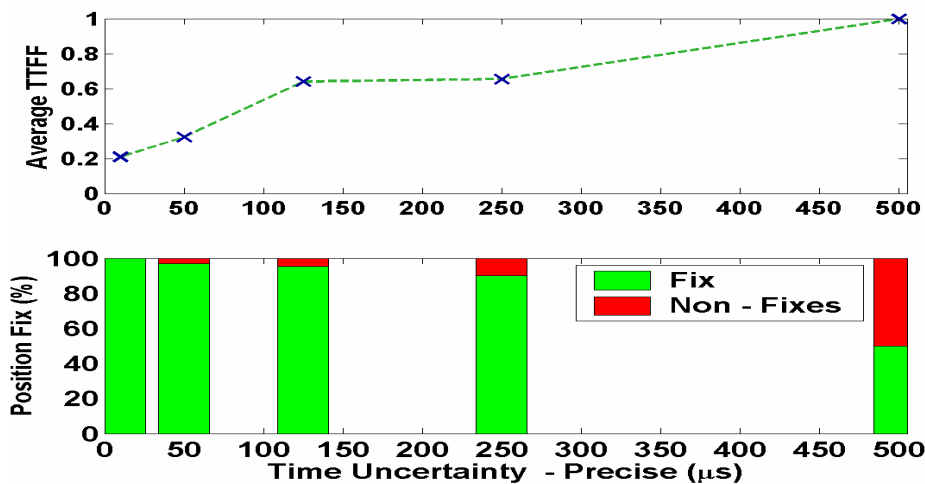


Figure 4.20: Precise Time Aiding – Normalized TTFF and Position Fixes for Concrete Basement Test

Table 4.14: Precise Time Aiding - Position Results using LSQ Concrete Basement Test

Precise Time (μs)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
10	60.1	3.7	49.6	3.7
50	61.0	3.8	78.2	3.9
125	61.7	4.9	68.4	3.5
250	54.5	4.8	55.0	4.1
500	43.0	5.1	77.0	3.0

**Table 4.15: Precise Time Aiding - Position Results using SiRF internal Solution for Concrete
Basement Test**

Precise Time (μs)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
10	74.7	5.2	70.6	4.7
50	59.6	5.1	57.3	4.0
125	58.5	5.3	56.2	3.6
250	55.4	5.4	53.2	3.6
500	58.6	5.3	71.6	3.2

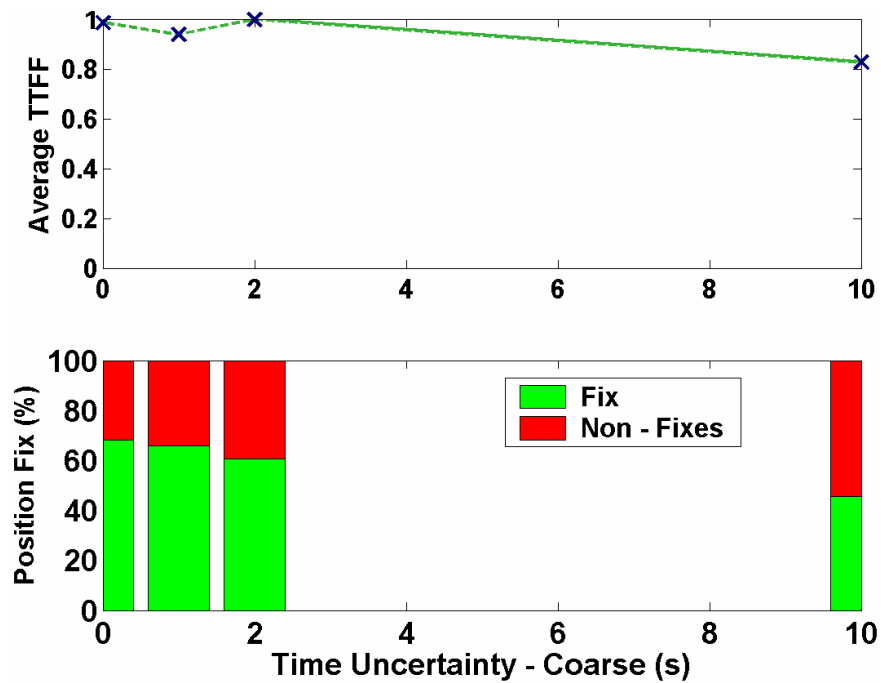


Figure 4.21: Coarse Time Aiding – Normalized TTFF and Position Fixes for Concrete Basement Test

Table 4.16: Coarse Time Aiding - Position Results Using LSQ for Concrete Basement Test

Coarse Time	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
1 ms	44.6	4.4	65.8	4.4
1 s	62.3	4.8	45.0	4.7
2 s	44.6	4.3	65.8	4.1
10 s	57.5	5.1	68.3	4.8

Table 4.17: Coarse Time Aiding - Position Results using SiRF Internal Solution for Concrete Basement Test

Coarse Time	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
1 ms	60.1	5.3	57.4	3.2
1 s	42.1	5.3	37.0	3.6
2 s	43.4	5.3	37.6	3.2
10 s	55.5	5.0	56.8	3.2

Similar to the field tests that were carried out in the residential concrete garage and the speed-skating track, when timing uncertainty (precise time aiding) was increased in the concrete basement, this led to an increase in the TTFF. For example, when the timing uncertainty was changed from 50 μ s to 125 μ s, the TTFF increased by 100%. The trend is similar to the previous two weak signal indoor environments; longer timing certainty results in longer search time or longer TTFF. When the timing uncertainty was changed to coarse time from precise time, the TTFF increased by 10%, however similar to the residential garage test, there was no observable trend between the different timing uncertainties (coarse) and TTFF. The success rate decreased with increasing timing uncertainty. For example, when the timing uncertainty was changed from 50 to 500 μ s, the percentage of successful position fixes decreased from 95% to 50%. It dropped even further, to 45%, when the timing uncertainty was changed to 10 s. The success rate at 500 μ s was better for the speed-skating track (85%) because all the signals in the concrete

basement were highly attenuated. The results illustrate the increased difficulty in obtaining a position fix or finding the GPS signal in an environment which has all weak signals. Similar results were obtained when acquisition tests using the simulator were carried out where all satellites had same power level of -145 dBm [Karunanayake, 2005b].

Similar to the speed-skating track, the receiver in the concrete basement would track or use an average of three satellites. Therefore, the use of height constraints when obtaining the position solution using C³NAV² was required. The position obtained using the two methods had similar results in terms of position accuracy. The results did not show any trends between five and thirty position fixes. Similar to the suburban environment and residential garage, the position accuracy (both methods using five fixes) improved with decreasing timing uncertainty.

The satellites had highly attenuated signals in the concrete basement with signal strengths less than 22 dB-Hz, with the exception of one satellite (see Figure 4.22). The concrete basement had very limited LOS signals (small window). Signals generally entered from concrete walls and were, therefore, highly attenuated.

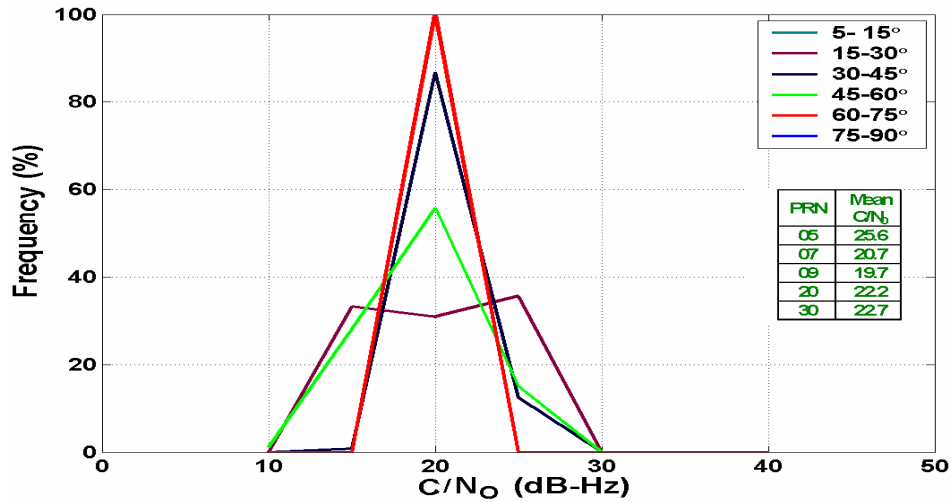


Figure 4.22: C/N_0 PDF for the Concrete Basement Test

Timing assistance is used to predict the code phase and requires a sub-millisecond accuracy which can only be achieved if the timing uncertainty is less than one millisecond. Therefore, when the timing accuracy is worse than one millisecond (coarse time aiding), it has no effect on the TTFF because the receiver is required to search through the entire 0-1022 code phase (see Figure 4.23) [Kinnari, 2002]. Precise time aiding is required to know the approximate location of the C/A code phase and hence speed up the acquisition search process. When the timing uncertainty (precise time) is increased, the result is a wider code phase uncertainty resulting in longer TTFF. The tests carried out under weak signal conditions illustrated the importance of good time aiding (125 μ s).

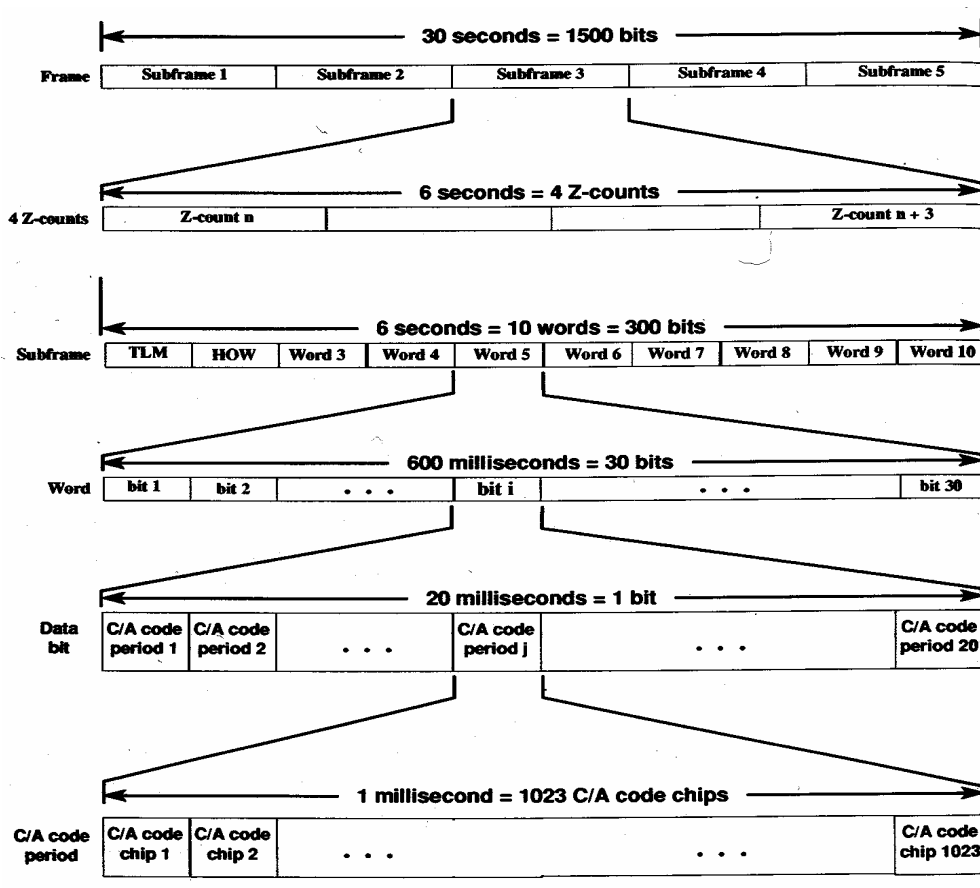


Figure 4.23: Time Relationship for L1 C/A Code [Source: Kaplan and Hegarty, 2006]

4.6 Horizontal Position Assistance

Horizontal position assistance can be sent to the mobile device via a wireless network, such as CDMA or GSM. The GSM network would use the E-OTD method to obtain a position [Syrjärinne and Kinnari, 2002]. The cell size could be as small as 3 km in urban centres and as large as 30 km in rural areas. The user position assistance is used to determine the visible satellites and the combination of timing and position assistance can be used to predict the approximate C/A code phase. The values of horizontal position

uncertainties were chosen to reflect the cell site size. Although a cell site would never be as large as 350 km, a position uncertainty of this magnitude was used because an uncertainty of greater than 300 km with exact timing assistance will force a complete sweep of the entire 1023 code chips [Kinnari, 2002]. The results were quantified using factors such as TTFF, 2-D position accuracy (vertical position accuracy was not considered, the height assistance was kept constant) and number of satellites tracked. The following sections discuss the results obtained in the four field environments.

4.6.1 Suburban Environment

Acquisition tests were carried out using different horizontal position uncertainties and the results with TTFF and position fixes are shown in Figure 4.24. The position results that were obtained from the C³NAV² and SiRF internal solutions are shown in Table 4.18 and Table 4.19.

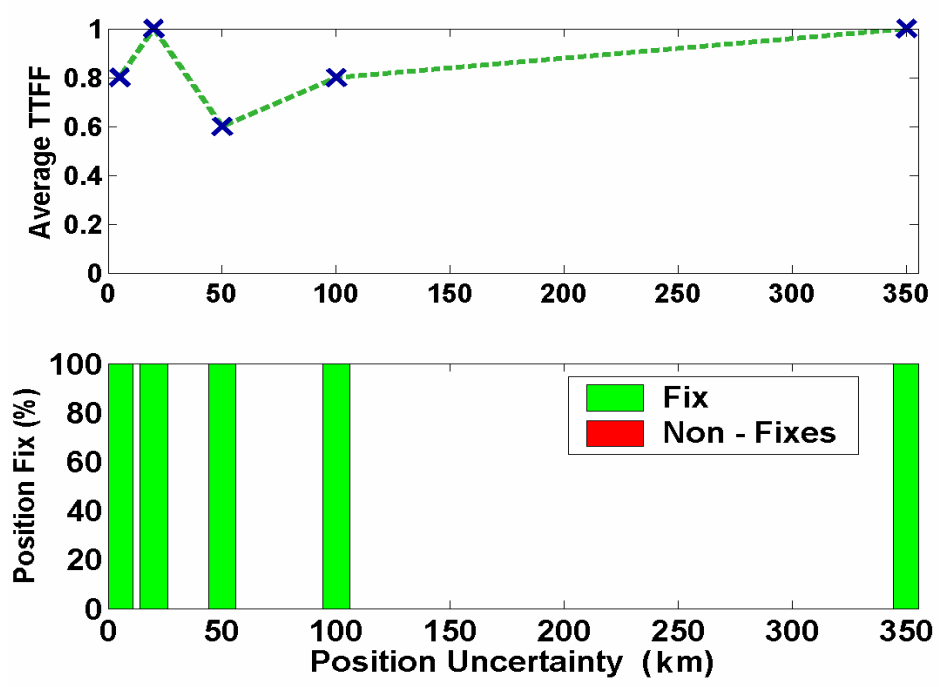


Figure 4.24: Horizontal Position Aiding- Normalized TTFF and Position Fix for Suburban test

Table 4.18: Horizontal Position Aiding- Position Results using LSQ Solution for Suburban test

Horizontal Position (km)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
5	60.8	5.6	60.1	5.7
20	42.6	6.5	38.8	7.2
50	41.2	6.6	31.7	8.1
100	46.4	7.2	34.8	8.6
350	41.2	6.2	36.2	7.7

Table 4.19: Horizontal Position Aiding- Position Results for SiRF Internal Solution for Suburban Test

Horizontal Position (km)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
5	61.3	5.3	69.3	6.1
20	40.7	6.6	40.4	7.6
50	37.3	7.0	30.1	8.1
100	35.3	7.6	39.5	8.8
350	31.0	6.8	33.0	7.9

Similar to the tests that were carried out using timing assistance, the tests with varying horizontal assistance failed to show any trends with respect to TTFF. The AGPS receiver was able to obtain a position fix successfully 100% of the time. The TTFF for 50 km was 40% worse than 20 km, however when using units of time the difference was only 2 s. The receiver used an average of seven satellites to compute the position solution using the two methods that showed similar results in terms of horizontal position accuracy. The position accuracy was better for thirty position fixes. There were more satellites used for thirty fixes. The HDOP or geometry generally improves with more satellites under nominal signal conditions. The position accuracy also improved with larger position uncertainty because more satellites were used to obtain the position solution. Simulation tests had not shown any trend with varying position assistance because the number of satellites used was constant [Karunanayake, 2005b].

4.6.2 Residential Garage

Similar tests using different position uncertainties were carried out in the residential garage. The TTFF for various position uncertainties are shown in Figure 4.25 and the position results are shown in Table 4.20 and Table 4.21.

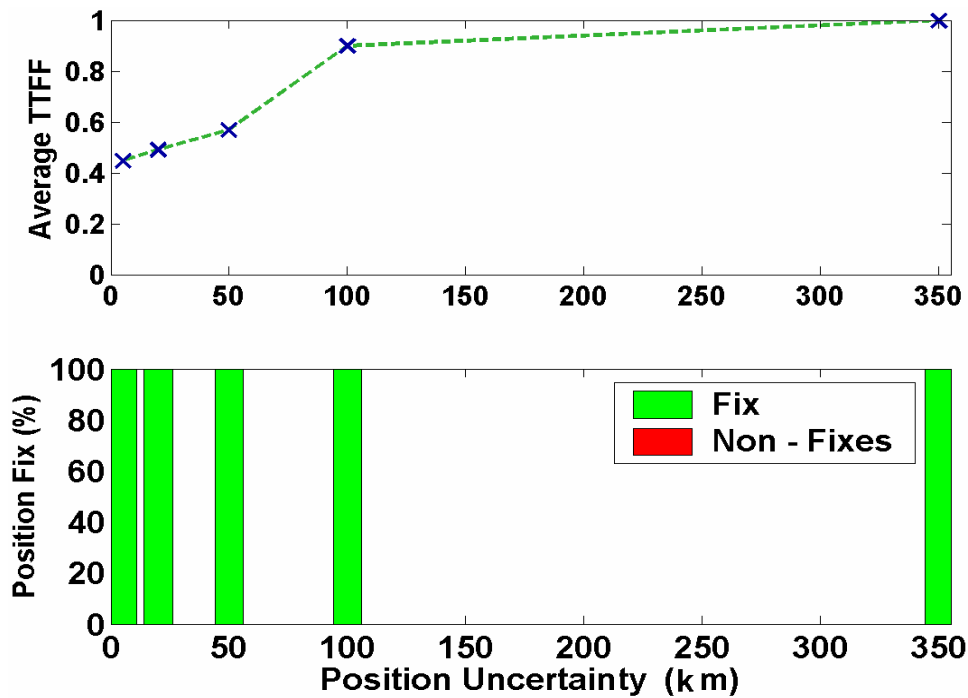


Figure 4.25: Horizontal Position Aiding- Normalized TTFF and Position Fix for Garage Test

Table 4.20: Horizontal Position Aiding- Position Results using LSQ Solution for Garage Test

Horizontal Position (km)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
5	19.0	3.5	23.5	4.5
20	56.3	4.3	37.1	4.7
50	45.1	5.3	56.7	4.9
100	25.2	4.7	40.4	4.5
350	43.1	3.0	50.7	3.9

Table 4.21: Horizontal Position Aiding- Position Results for SiRF Internal Solution for Garage Test

Horizontal Position (km)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
5	18.8	5.1	20.7	5.0
20	44.8	5.1	31.8	5.0
50	61.5	5.4	35.2	4.1
100	61.0	5.4	46.5	4.8
350	52.0	5.2	42.6	3.8

Unlike the suburban test that did not show any trend between the position uncertainty and TTFF, the tests in the garage showed that when the horizontal position uncertainty

increased, so did the TTFF. For example, when the position uncertainty was changed from 20 km to 100 km, the TTFF increased by 100% and the receiver was able to obtain a position fix 100% of the time. Position assistance and accurate timing information can be used to predict the approximate C/A code phase. However, unlike timing assistance, position assistance with ephemeris can be used to predict the approximate satellite Doppler. It has been shown in van Diggelen, [2001] that an uncertainty of one kilometre will result in a Doppler error of one Hertz. Higher uncertainty in position would result in larger search space (C/A code and Doppler), resulting in longer search time or TTFF.

The two position solutions used an average of at least four satellites with similar horizontal position accuracy. The position results did not show any trends between five and thirty positions fixes or with different horizontal position uncertainty. The number of satellites was similar for different tests (three to five). The tests with varying horizontal position aiding using the simulator also did not show any trends in position results [Karunanayake, 2005b]. The TTFF increased with decreasing horizontal position aiding under weak signal conditions (-140 dBm).

4.6.3 Speed-skating Track

The TTFFs for various position uncertainties are shown in Figure 4.26 and the corresponding position results using C³NAV² and SiRF software are shown in Table 4.22 and Table 4.23.

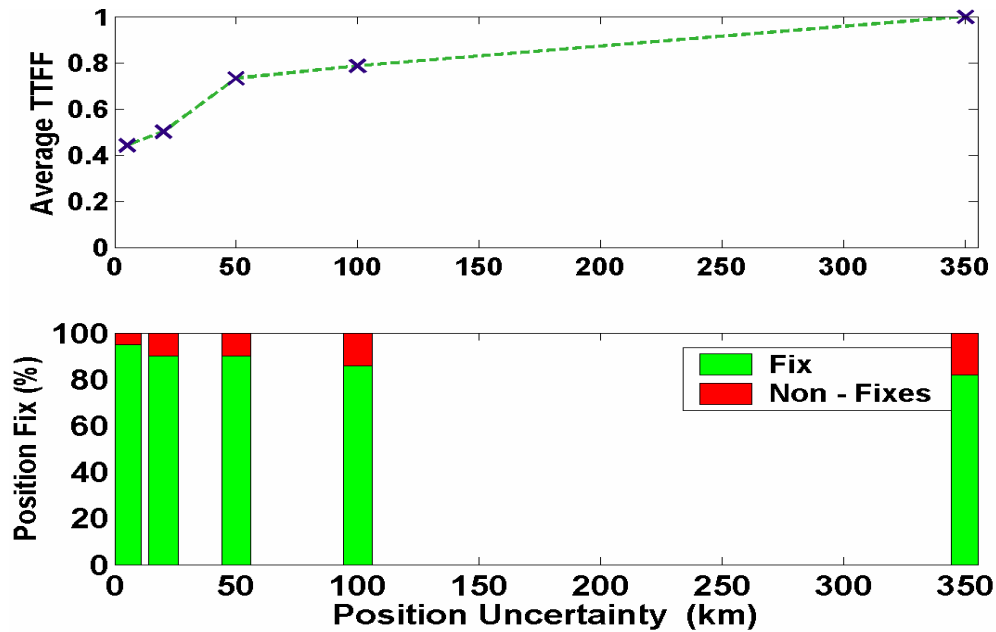


Figure 4.26: Horizontal Position Aiding – Normalized TTFF and Position Fix for Speed-skating Test

Table 4.22: Horizontal Position Aiding- Position Results using LSQ Solution for Speed-skating Test

Horizontal Position (km)	Five Position Fixes		Thirty Position Fixes	
	2D RMS	Mean # of	2D RMS	Mean # of
	Error (m)	Satellites	Error (m)	Satellites
5	58.4	3.9	59.1	3.9
20	43.2	3.3	42.2	3.6
50	59.7	3.6	43.8	3.4
100	57.6	4.0	45.9	4.9
350	51.8	4.5	45.2	5.1

Table 4.23: Horizontal Position Aiding- Position Results for SiRF Internal Solution for Speed-skating Test

Horizontal Position (km)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
5	45.7	3.7	37.5	3.8
20	34.1	5.5	36.1	4.3
50	40.9	5.5	41.0	4.4
100	50.8	5.4	50.6	4.9
350	46.2	5.3	38.2	5.4

Similar to the garage test, acquisition tests in the speed-skating track showed that increasing the horizontal position uncertainty led to a longer TTFF. For instance, when the position uncertainty was changed from 20 to 50 km, the TTFF increased by 50%. However, unlike the garage test, increasing the position uncertainty resulted in a decrease in the success rate for obtaining a position fix. For example, when the position uncertainty was changed from 20 km to 100 km the success rate decreased from 92% to 82%. The results are similar to tests that were carried out using different time aiding. It is difficult to search GPS signals under extremely weak signal conditions. When the position uncertainty is large, a larger Doppler and C/A search space is the result. The receiver could be searching in the wrong search bin and is unable to find the GPS signal before the required time of 300 s.

The AGPS receiver used at least three satellites to obtain the C³NAV² solution for most of the tests. The SiRF internal solution however used an average of four satellites and had better horizontal position accuracy because of better satellite availability. Similar to the residential garage test, there were no observable trends between five and thirty position fixes or different position aiding values.

4.6.4 Concrete Basement

Finally tests using different position uncertainties were carried out in the concrete basement. The TTFFs using different position uncertainties are shown in Figure 4.27, while the position results are shown in Table 4.24 and Table 4.25.

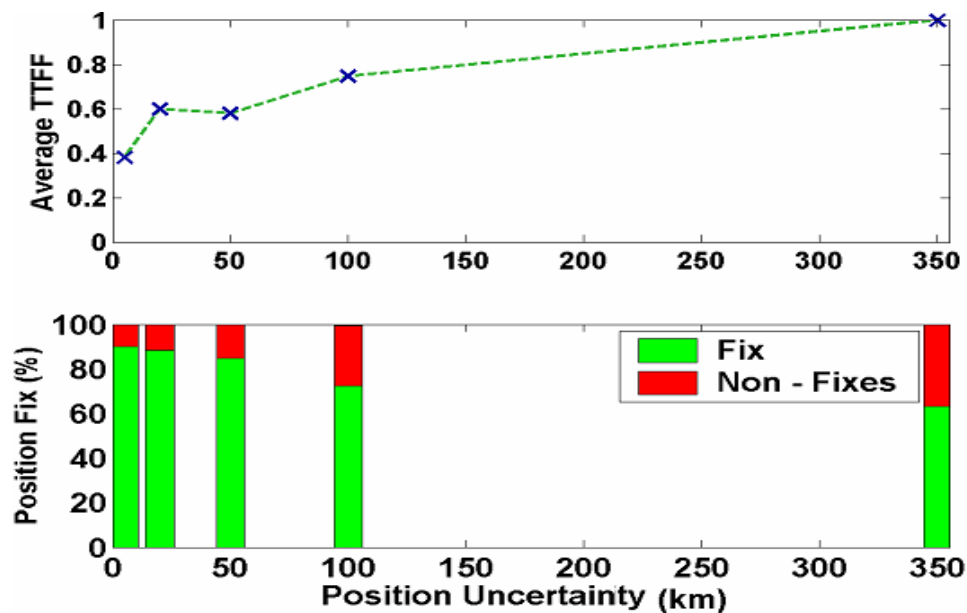


Figure 4.27: Horizontal Position Aiding – Normalized TTFF and Position Fix for Concrete Basement

Table 4.24: Horizontal Position Aiding– Position Results using LSQ Solution for Concrete Basement

Horizontal Position (km)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
5	61.7	5.3	68.4	3.2
20	67.5	4.5	58.5	3.3
50	48.7	4.2	86.4	3.2
100	86.0	4.4	104.1	3.4
350	62.6	4.3	62.3	2.9

**Table 4.25: Horizontal Position Aiding– Position Results for SiRF Internal Solution for Concrete
Basement**

Horizontal Position (km)	Five Position Fixes		Thirty Position Fixes	
	2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
5	58.5	5.3	56.2	3.6
20	45.5	5.1	42.8	3.2
50	69.1	5.4	75.1	3.3
100	54.3	5.2	58.6	3.2
350	46.3	5.3	56.1	2.7

Similar to acquisition tests that were carried out in the residential garage and the speed-skating track, increasing the position uncertainty led to longer TTFFs here as well. For example, when the position uncertainty was changed from 50 km to 100 km, the TTFF increased by 40%. When the position aiding was changed from 20 km to 50 km the TTFF dropped by about four seconds. Since the TTFF was in hundreds of seconds, this small drop is not particularly significant. Similar to the speed-skating test, increasing the position uncertainty led to a decrease in the success rate. For instance, when the position uncertainty was changed from 20 km to 100 km the success of position fixes decreased from 85% to 72%. The success rate was better for the speed-skating track (82% for 100 km) when compared to the concrete basement test. The speed-skating track has one strong signal which is initially acquired and later used to acquire the remaining weaker satellites. In the concrete-basement all the signals are weak therefore takes a longer time to acquire. In many instances it is not possible to acquire or obtain a position fix during the required time. The concrete-basement is in fact a very harsh environment. This can be illustrated by the fewer number of satellites tracked or used by the two methods (position solutions) and the poor position accuracy. The position results obtained with both methods was similar, with SiRF solution providing slightly better position accuracy. Similar to the speed-skating track and residential garage the position results did show any trends for position aiding scenarios.

4.7 Ephemeris and Almanac Assistance

When the mobile station knows the approximate user location and is aware of the current GPS time, visible satellites can be determined if the satellite ephemeris is sent to the mobile via a wireless network. In all of the above tests, the satellite ephemeris and almanac were provided to the AGPS receiver. In the following tests, the satellite ephemeris or almanac will be withheld from the AGPS receiver. Simulation tests have already demonstrated the importance of satellite ephemeris in terms of factors such as lower TTFF and better acquisition sensitivity.

Acquisition tests were carried out using scenarios three and four (see Table 4.1) in the four different field test environments. In scenario three the receiver does not get any ephemeris while in scenario four the receiver does not get almanac. The timing and horizontal uncertainty were kept at 125 μ s and 5 km for both scenarios. The position results using C³NAV² and the SiRF internal solution are shown in Table 4.26 and Table 4.27.

Table 4.26: Ephemeris or Almanac Aiding- Position Results Aiding Using LSQ for all the Test Sites

		Five Position Fixes		Thirty Position Fixes	
		2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
Suburban	Almanac	41.4	6.7	35.9	7.2
	Ephemeris	49.4	7.1	40.4	6.7
Residential	Almanac	25.9	5.3	25.2	6.2
Garage	Ephemeris	21.3	5.2	18.5	6.4
Speed- skating Track	Almanac	57.9	3.4	55.4	3.3
	Ephemeris	X	X	X	X
Concrete	Almanac	63.2	4.1	56.1	3.4
Basement	Ephemeris	X	X	X	X

Table 4.27: Ephemeris or Almanac Aiding - Position Results Using SiRF Internal Solution for all the Test Sites

		Five Position Fixes		Thirty Position Fixes	
		2D RMS Error (m)	Mean # of Satellites	2D RMS Error (m)	Mean # of Satellites
Suburban	Almanac	46.7	6.8	41.6	7.3
	Ephemeris	21.0	7.3	35.4	6.8
Residential Garage	Almanac	27.0	5.3	23.0	6.4
	Ephemeris	31.2	5.3	27.2	6.6
Speed-skating Track	Almanac	62.9	3.6	61.9	3.6
	Ephemeris	X	X	X	X
Concrete Basement	Almanac	45.8	4.2	37.5	4.5
	Ephemeris	X	X	X	X

X – Acquisition or position fix could not be obtained

The acquisition test results show that the AGPS receiver was able to perform signal acquisition without ephemeris in the suburban and residential garage but not in the speed-skating track or concrete basement. The TTFF was ten times more for the suburban test and six times more for the residential garage when compared with time (125 μ s) and horizontal position aiding (5 km). The position results were similar to the ones that were obtained using different time and position aiding.

Acquisition tests could not be carried out in the other two environments (speed-skating track and the concrete basement). This further confirms the simulation tests, where the AGPS receiver (without ephemeris) was unable to acquire signals below -142 dBm.

The acquisition tests without almanac assistance had the same TTFF when compared with the time and position assistance of 125 μ s and 5 km for all four test environments. If the receiver does not have ephemeris, it will download satellite ephemeris and this can take up to 30 seconds. This prolongs the search process resulting in longer TTFFs. The satellite almanac, unlike satellite ephemeris, provides coarse information for such things as satellite orbits and, therefore, is not required for signal acquisition.

4.8 Comparison of Different Environments

This section compares the acquisition performance of the AGPS receiver in the four different environments. The roof-top receiver is again used as a reference since there are strong LOS signals (43 to 45 dB-Hz) present on the roof-top. All the other tests can be referenced to the roof test. The results for different precise time aiding and horizontal position assistance are shown in Figure 4.28 and Figure 4.29. The position results obtained from C³NAV^G² are given in Table 4.28 where the AGPS receiver received ephemeris, almanac; time aiding of 125 μ s uncertainty and position assistance of 5 km uncertainty.

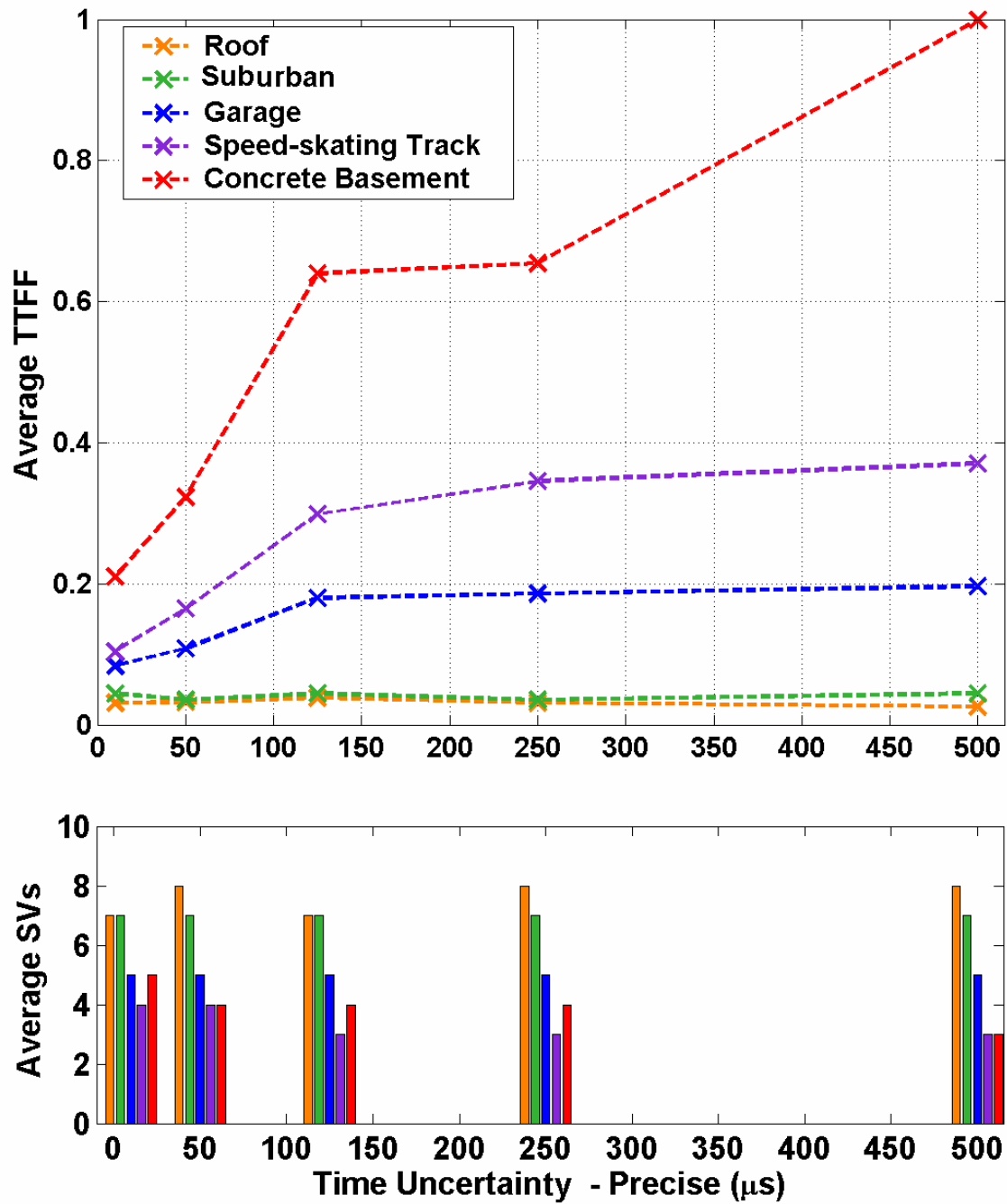


Figure 4.28: Precise Time Aiding- Normalized TTF and Number of Satellites Tracked for Different Test Sites

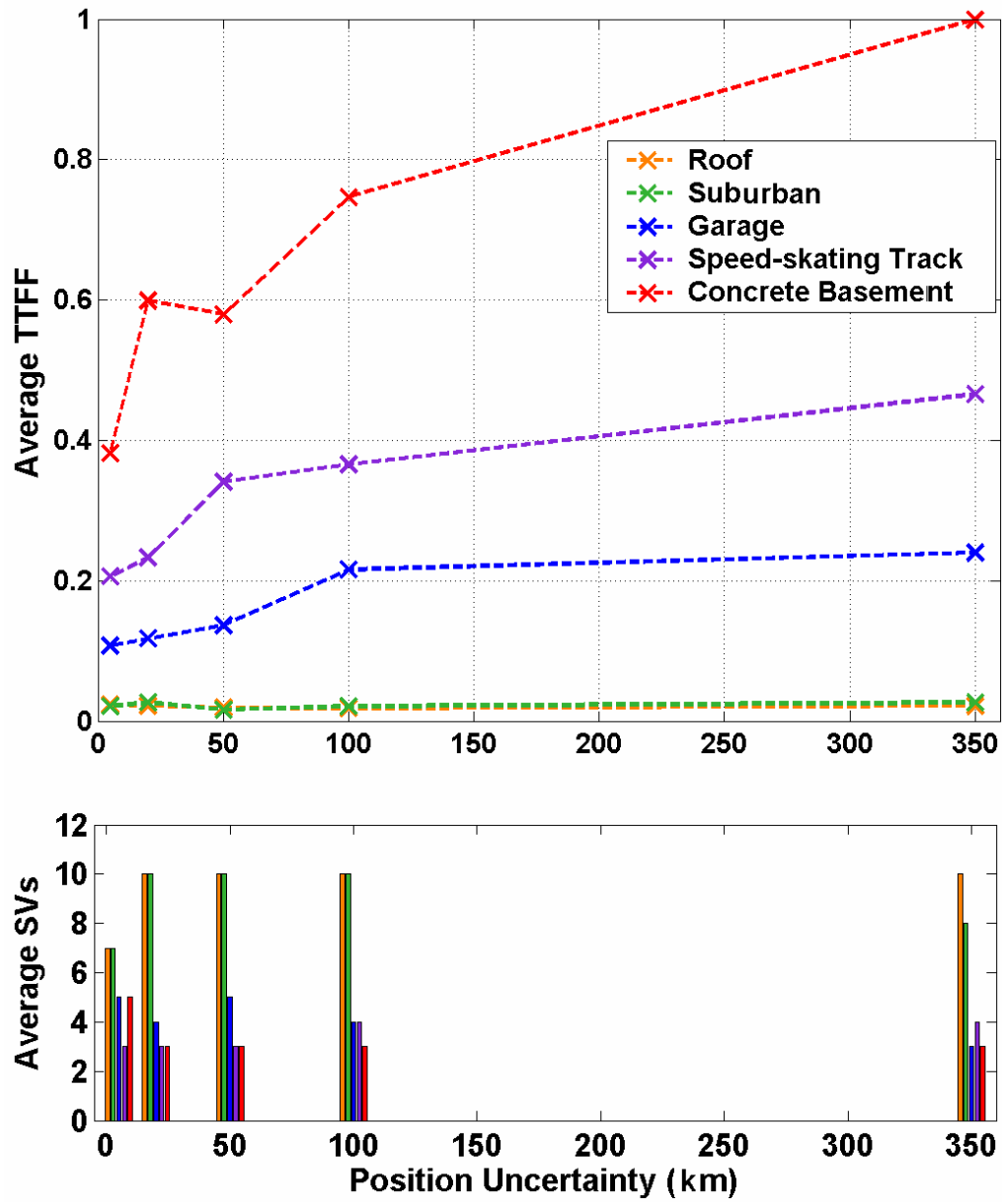


Figure 4.29: Horizontal Position Aiding- Normalized TTFF and Number of Satellites for Different Test Sites

Table 4.28: Position Results Using LSQ Solution Under Different Field Test Conditions

Different Field Test Sites	Five Position Fixes			Thirty Position Fixes		
	2D RMS Error (m)	Mean # of Satellites	Mean HDOP	2D RMS Error (m)	Mean # of Satellites	Mean HDOP
Roof	3.6	8.2	1.1	3.8	7.9	1.2
Suburban Environment	60.8	5.6	3.5	60.1	5.7	3.2
Residential Garage	18.8	5.1	1.3	23.5	5.0	1.6
Speed-skating Track	45.7	3.7	4.2	37.5	3.8	3.9
Concrete Basement	61.7	5.3	2.7	68.4	3.2	3.5

The trends between the timing and position assistance showed similar results with the number of satellites tracked and the TTF. The TTF between two different environments are compared using time aiding of 125 μ s and position uncertainty of 5 km. The TTFs of the roof test showed a similar trend when compared to the suburban environment. Both test sites had nominal signal conditions (39 to 45 dB-Hz). Position accuracy was worse in the suburban environment, which could be due to a lower number

of satellites used. A lower number of satellites results in poorer geometry or multipath effects (glass building would cause strong specular reflections). The residential garage had attenuated signals (30 dB-Hz) with TTFFs which were five times longer than the suburban environment. However, the AGPS receiver had better position accuracy due to factors such as better geometry (HDOP was 1.6, less than suburban which had HDOP of 3.2), and multipath from diffuse sources may not be as severe as that due to specular reflective sources. Similarly, the speed-skating environment with signal strength of 25 dB-Hz had a TTFF that was twice as long as that of the residential garage. The speed-skating environment also had poor position accuracy that could be due to factors such as poor geometry and strong specular reflections due to the corrugated roof for example. Although the concrete basement and the speed-skating track had similar signal conditions however, the signal in the concrete basement was 5 dB lower (see Figure 4.19 and Figure 4.22), the TTFF was twice as long for the concrete basement. This was the case because the test in the speed-skating track was carried out near a window where there was one strong satellite (PRN 5). Simulation tests and Karunanayake [2005b] have shown the importance of initially acquiring a strong satellite that can then be used to aid in acquiring the remaining weaker satellites. Once the first strong satellite is acquired, it can be used to provide such things as accurate GPS timing and clock bias, which can be used to acquire the remaining weaker satellites. The results have shown that it takes a longer period of time to acquire a signal in weaker signal conditions. Weaker signals imply a longer search time or integration time and, thus, a longer TTFF [Shewfelt *et al.*, 2001]. Typically coherent integration is kept constant while non-coherent integration is changed to acquire signal at different power levels.

Figure 4.30 and Figure 4.31 show the time series of the horizontal error and number of satellites for the residential garage test and the concrete basement test. The aiding data for both tests included ephemeris, almanac, time aiding with uncertainty of $125 \mu\text{s}$ and position assistance with uncertainty of 5 km. The Concrete basement showed lower satellite availability and lower position accuracy. Significant signal blockage resulted in a lower number of satellites tracked, while weaker signals (20 – 25 dB-Hz) would lead to frequent loss of lock, larger code errors which would degrade the position accuracy.

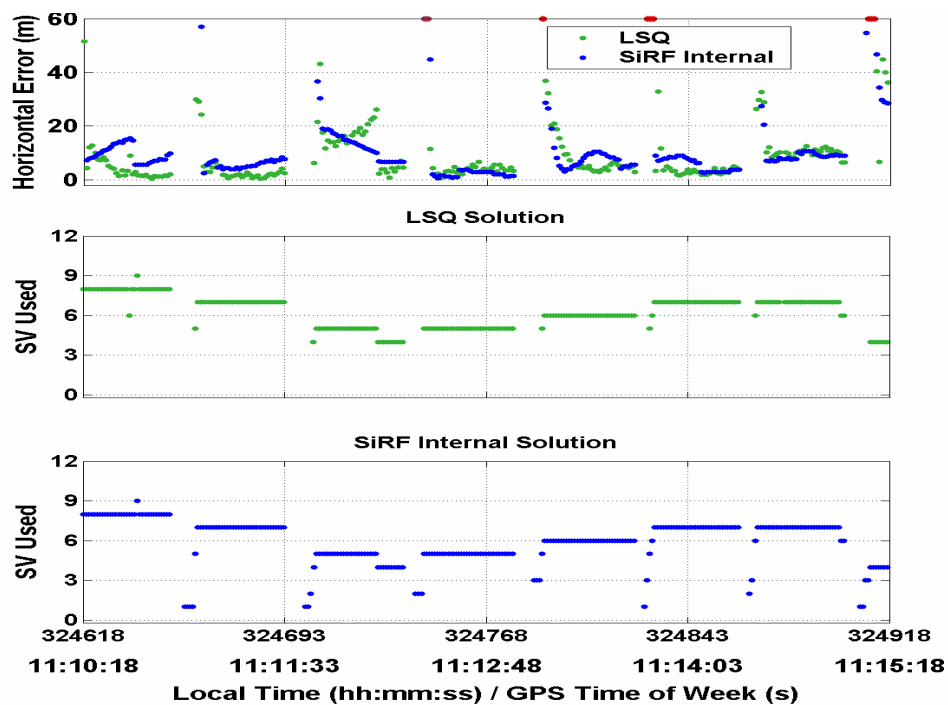


Figure 4.30: Horizontal Error and the Number of Satellites Tracked for Residential Garage Test

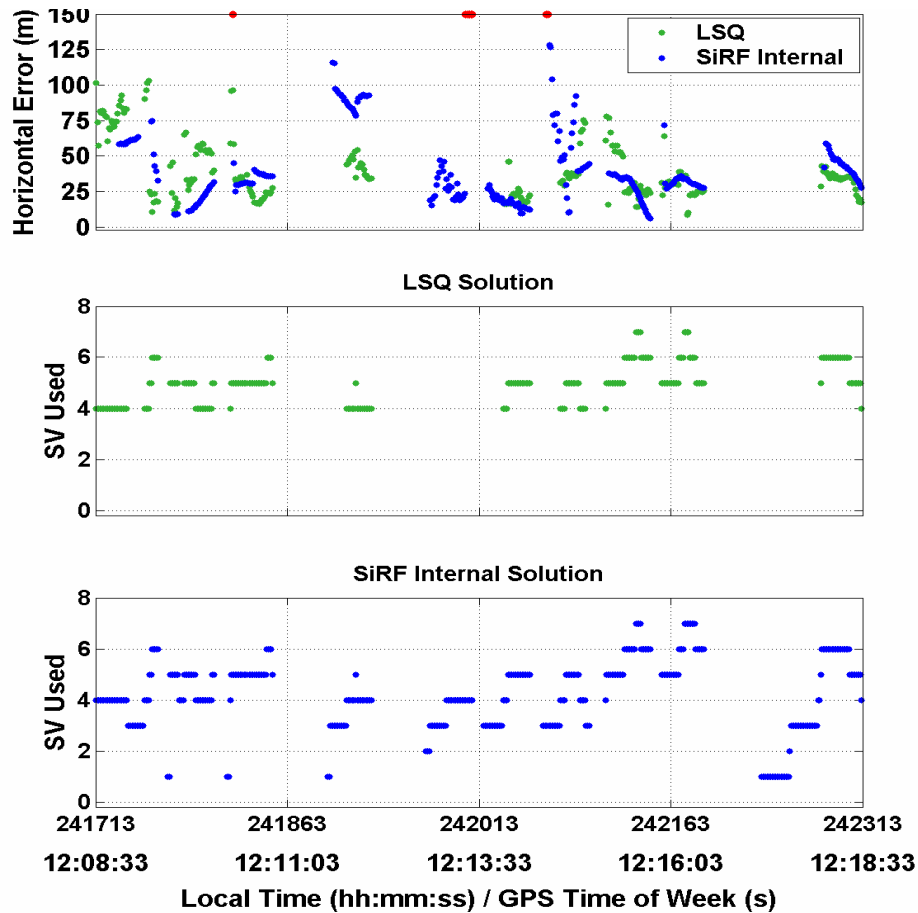


Figure 4.31: Horizontal Error and the Number of Satellites Tracked for Concrete Basement Test

4.9 Comparison of Simulation & Field Tests

Simulation tests were carried out in a controlled environment where every satellite had the same power unlike field test conditions where the signal power varied due to various factors such as signal attenuation, and signal blockage or reflected signals coming from different types of surfaces. No measurement errors such as multipath were simulated. Field tests however were subject to multipath effects which would affect the position accuracy. Signal blockage reduces satellite visibility which would lead to poor geometry

or HDOP which would also degrade the position accuracy. During simulation tests the number of satellites is kept constant and, therefore, signal blockage is not a problem. The results using the two methodologies showed similar trends in TTFF and acquisition sensitivity. The trends in the TTFF depend on the signal strength and were, therefore, easier to observe. Research by El-Natour *et al.*, [2005] has shown that multipath does not affect signal acquisition in terms of TTFF.

The trends in position were difficult to observe because there are other factors such as multipath which affect the position accuracy during various field tests. The only factor in the simulator tests that would affect the position accuracy was different signal power level. Simulation tests with different time aiding showed that position accuracy became better with increased time uncertainty [Karunanayake, 2005b]. Field tests have shown that trend for the first two environments (suburban environment and residential garage), however the last two environments which had very weak signals did not show any trends. Simulation and field tests however did not show any trend between varying horizontal position uncertainty and the position results.

4.10 Chapter Summary

Acquisition field tests using a HSGPS receiver and AGPS receiver were carried out at various test sites. Conclusions drawn from the field tests are discussed below.

- The acquisition field tests showed the limitations of the HSGPS receiver: its inability to acquire in weak signal conditions and long TTFF (ten times longer) when compared to the AGPS receiver.
- The acquisition field tests showed that lack of satellite ephemeris resulted in longer TTFF, while that the almanac is not required for signal acquisition
- Increased timing uncertainty (precise time aiding) results in longer TTFF under weak signal conditions. The tests also showed that sub-millisecond level accuracy for the time aiding is required. In other words, coarse time aiding has no effect on TTFF.
- The tests under different field test conditions showed the increasing level of difficulty in acquiring satellite signals or obtain a position, especially in environments such as the speed-skating track and the concrete basement. These two environments had highly attenuated signals (20 to 25 dB-Hz). The position accuracy using the two methods (SiRF internal and least squares) was less than 50 m for the first two environments (suburban and residential garage)but was worse for the last two environments (speed-skating track and concrete basement, less than 70 m). The position accuracy was better than E-OTD (100 m) and within the maximum requirement of 150 m for the FCC-E911 mandate.

CHAPTER 5: FIELD TESTS: TRACKING

This chapter discusses the test objectives, methodologies and results that were obtained for tracking tests at different test sites. The test sites were the same ones used in Chapter Four to conduct acquisition tests. The chapter also provides analysis under different test conditions and provides some conclusions that can be drawn from various tracking tests.

5.1 Test Objectives

Tracking tests were conducted in different field test conditions to meet the following test objectives.

- Compare the performance of HSGPS and AGPS under weak or degraded field test conditions
- Determine the effects of aiding data on tracking performance of the AGPS receiver

5.2 Field Test Methodology

Tracking tests (see Figure 5.1) that were carried out inside a building required an initialization of twenty minutes under open sky conditions to ensure that the receivers had the complete satellite almanac and ephemeris [MacGougan, 2003]. The AGPS receiver is able to acquire signals indoors; however, initialization was carried with both the HSGPS

and AGPS receivers to keep the methodology consistent. The reference receiver provided aiding data (timing (125 μ s), position uncertainty (5 km), and satellite ephemeris and almanac) to the AGPS receiver. The tests were carried out to illustrate the effects of multipath, signal blockage, and signal attenuation on solution accuracy and availability under different field test conditions. The tracking tests were analyzed through the horizontal and vertical errors, number of satellites used in the position solution, C/N_0 and measurement residuals. The horizontal and vertical position accuracies were measured using Root Mean Square (RMS). Field tests were carried out on different days at various test sites (see Table 5.1).

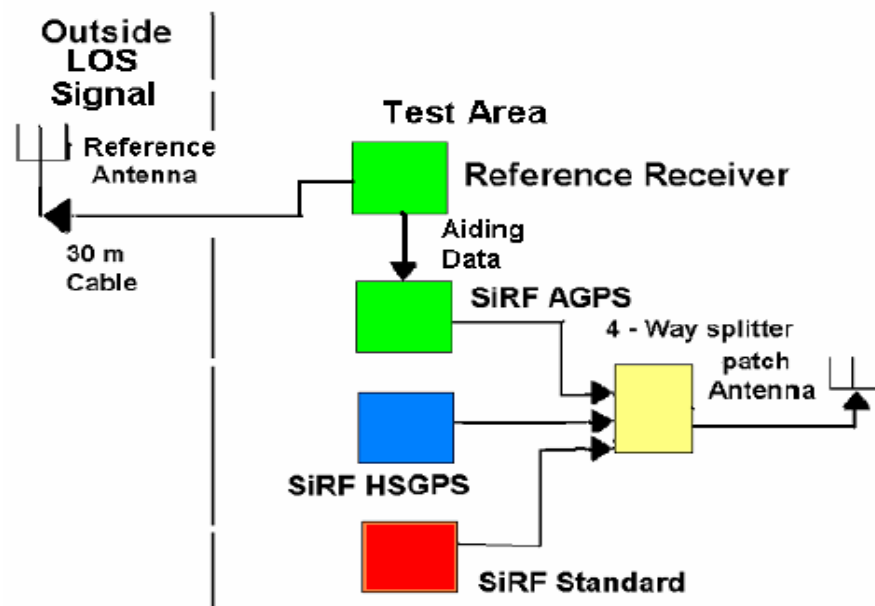


Figure 5.1: Field Set-up for Tracking Tests

Table 5.1: Dates of Tracking Field Tests

Environment	Date of Test
Suburban Environment	October 9 th 2004
Residential Garage	December 9 th 2004
Speed-skating Track	December 3 rd 2004
Concrete Basement	December 6 th 2004

5.3 Suburban Environment

Field tests in a suburban environment were carried out using the AGPS, HSGPS and standard receivers. Figure 5.2 shows the azimuth/elevation of the satellites tracked, while the position results for the three receivers using C³NAV² and the SiRF internal solution are shown in Figure 5.3, Figure 5.4 and Figure 5.5. The residual errors and C/N₀ for the three receivers are shown in Figure 5.6 through Figure 5.11, while Figure 5.12 shows the mean C/N₀ for different elevation and azimuth.

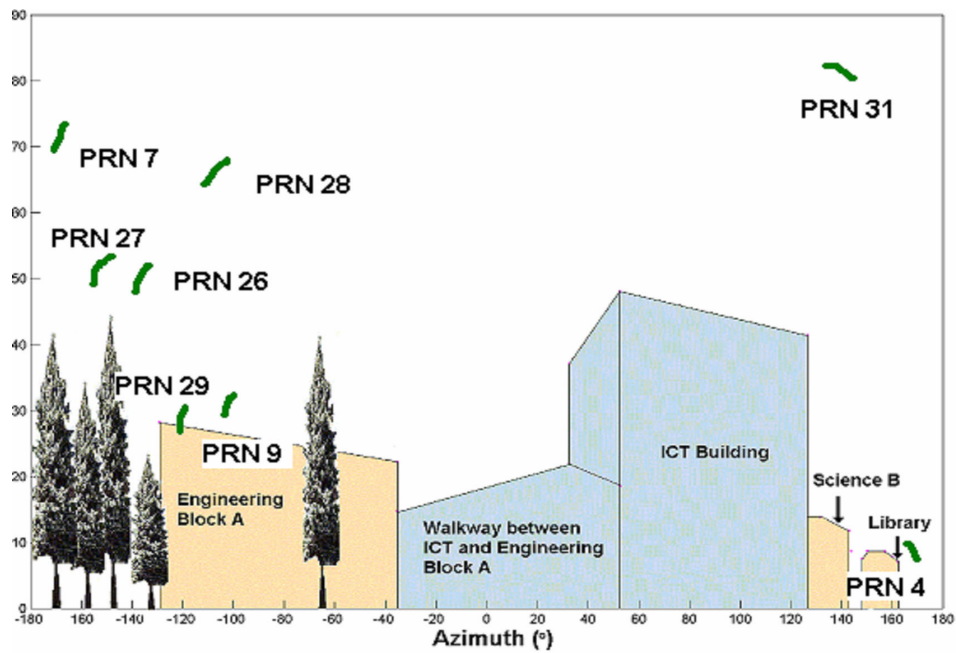


Figure 5.2: Azimuth and Elevation for the Satellites Tracked in the Suburban Test

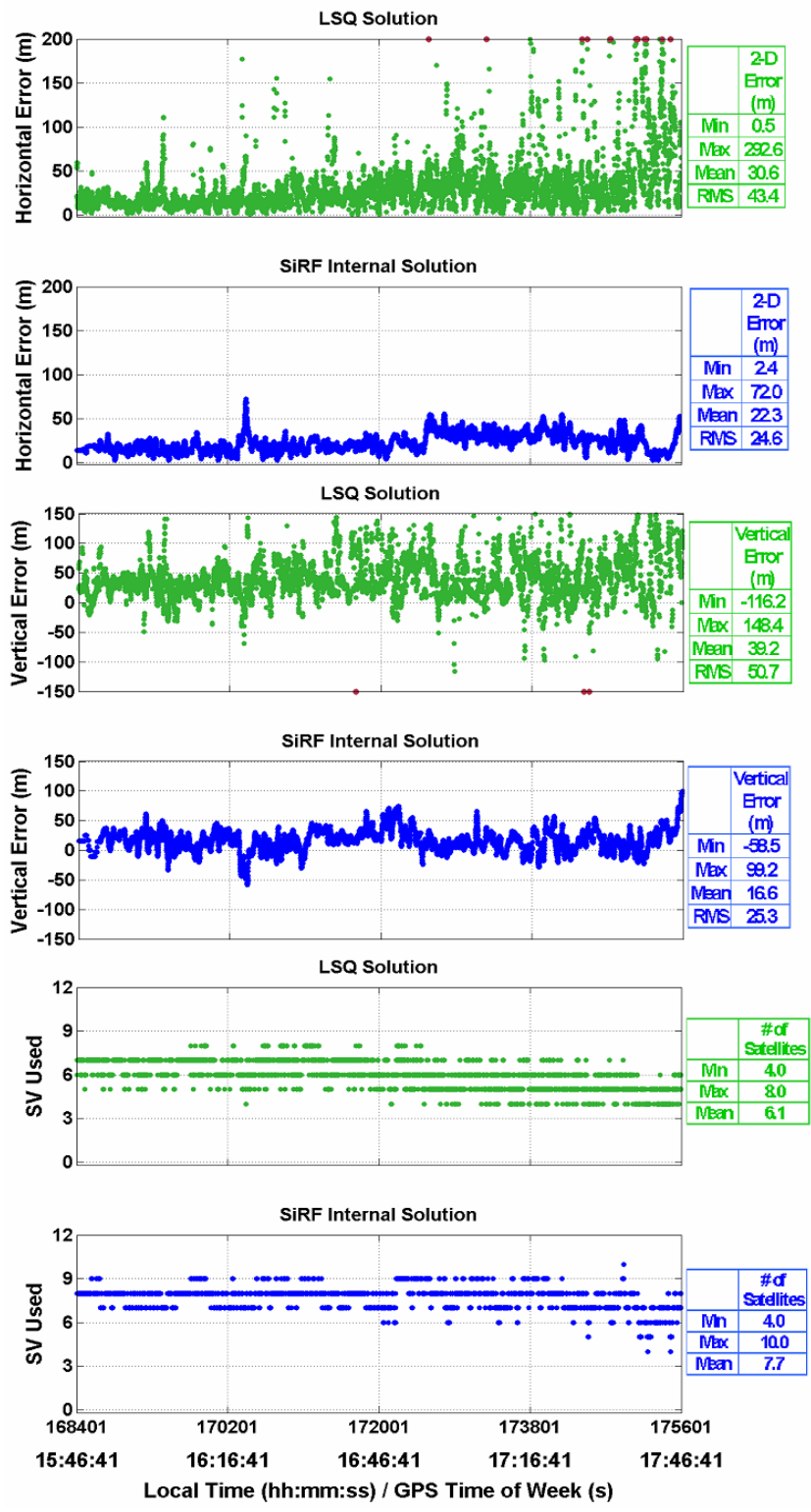


Figure 5.3: AGPS Receiver Position Results for the Suburban Test

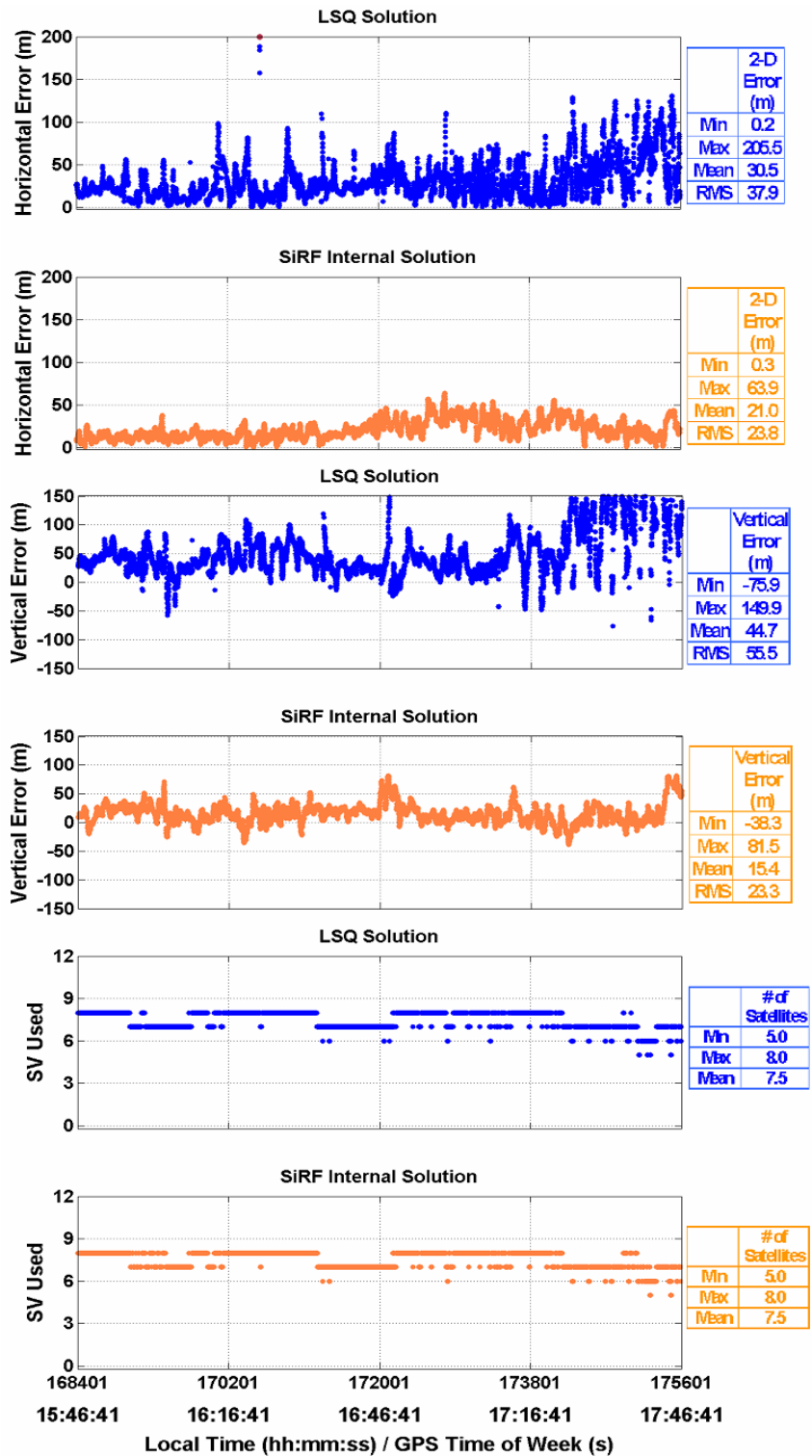


Figure 5.4: HSGPS Receiver Position Results for the Suburban Test

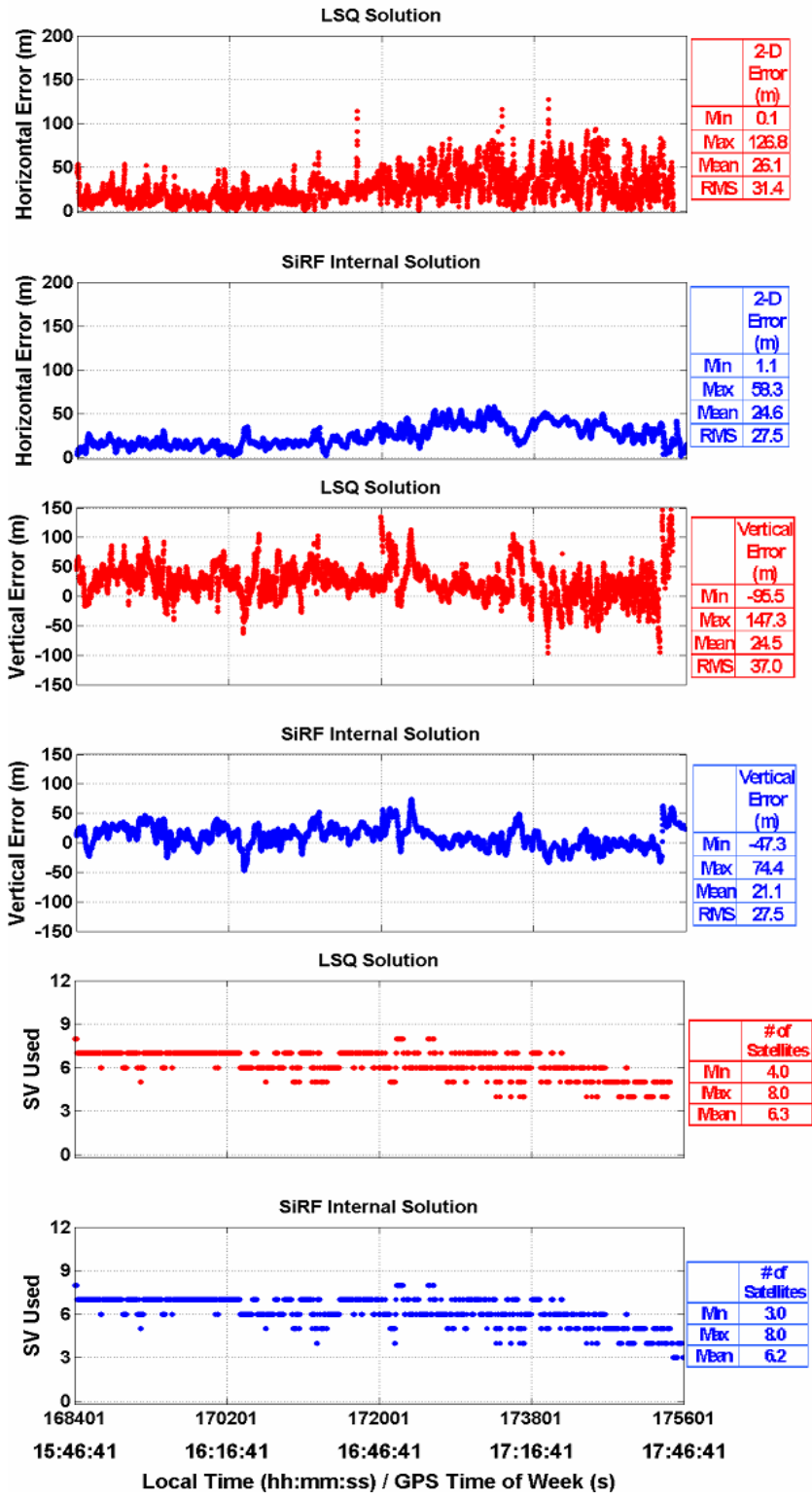


Figure 5.5: Standard Receiver Position Results for the Suburban Test

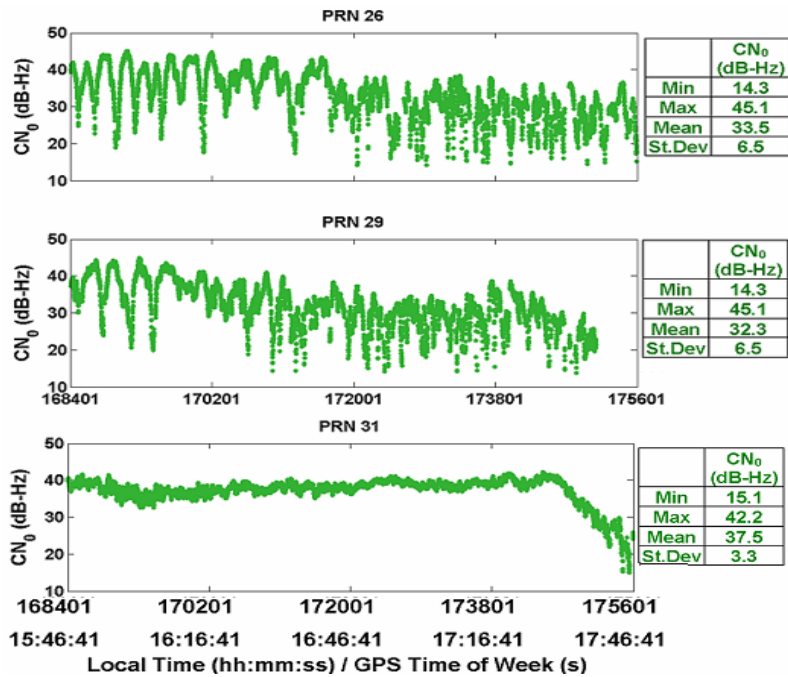


Figure 5.6: Time Series for the C/N₀ for AGPS Receiver for the Suburban Test

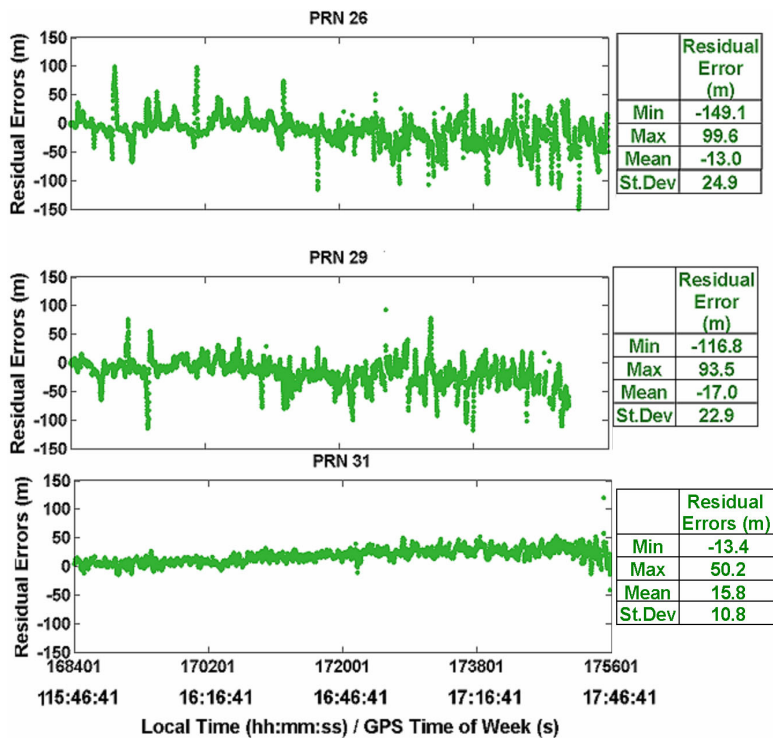


Figure 5.7: Time Series of Residual Errors for the AGPS Receiver the Suburban Test

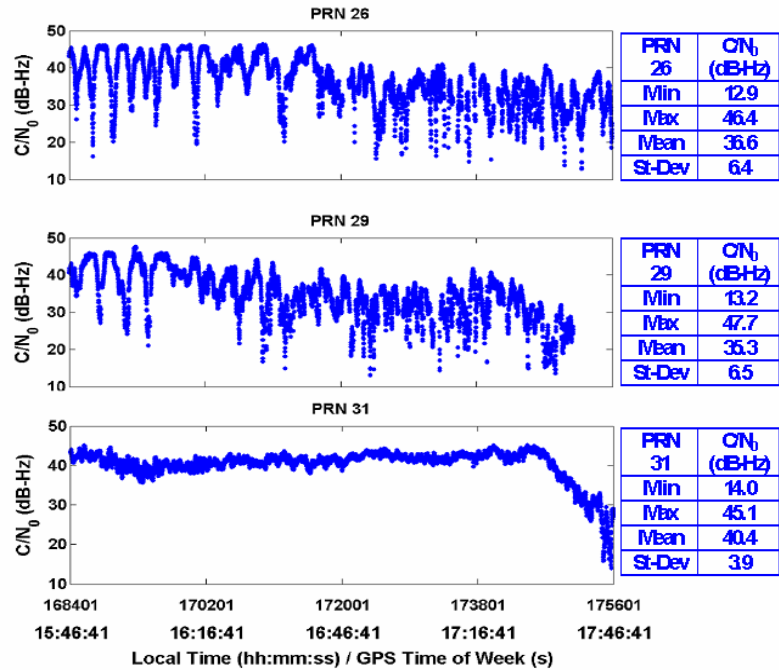


Figure 5.8: Time Series for the C/N₀ for HSGPS Receiver the Suburban Test

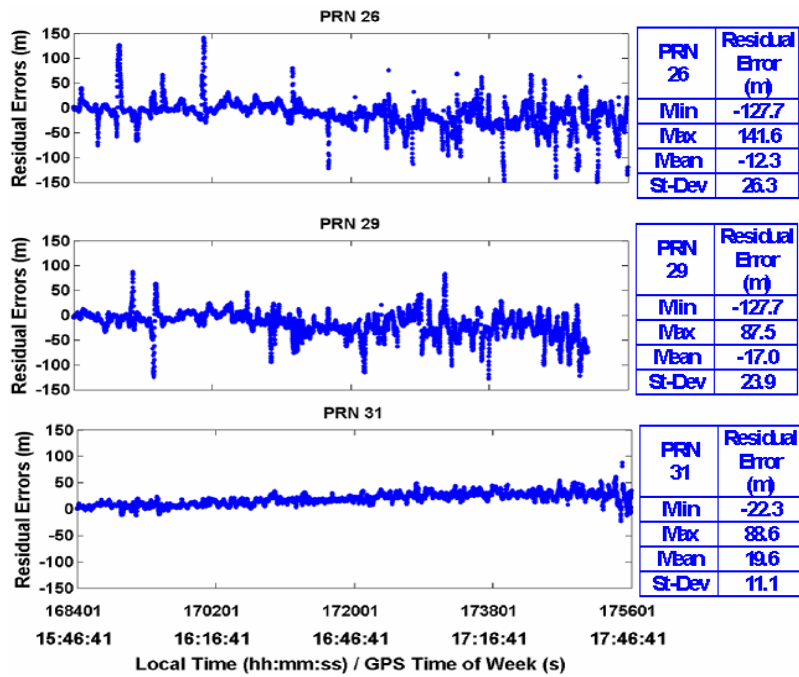


Figure 5.9: Time Series of Residual Errors for the HSGPS Receiver for the Suburban Test

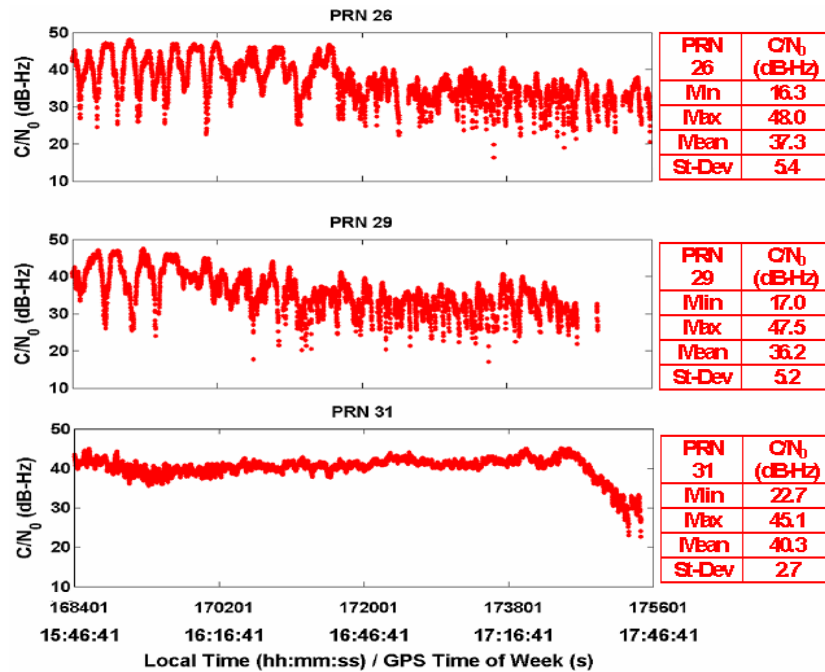


Figure 5.10: Time Series for the C/N₀ for Standard Receiver for the Suburban Test

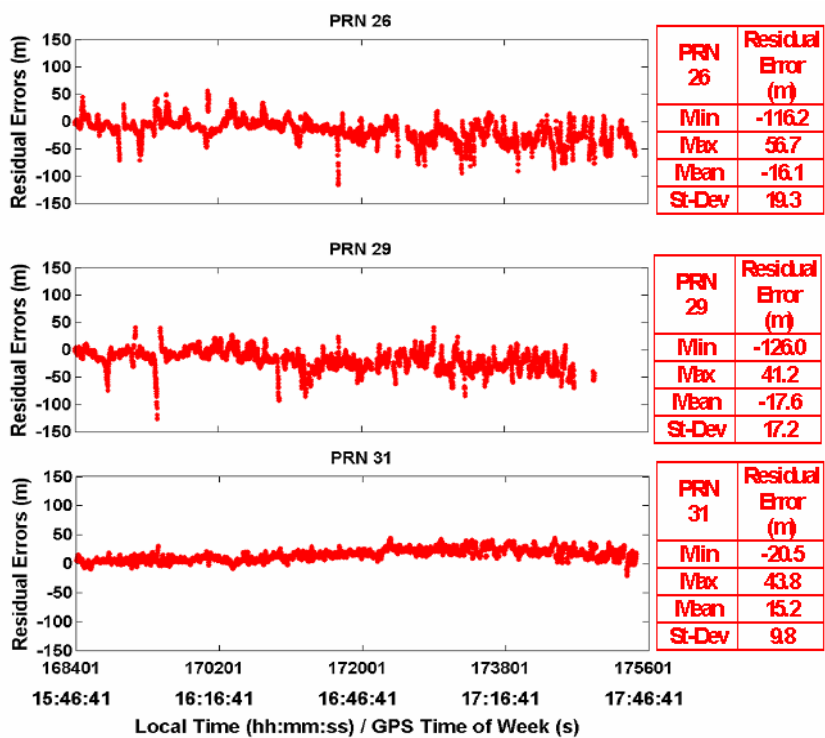


Figure 5.11: Time Series of Residual Errors for the Standard Receiver for the Suburban Test

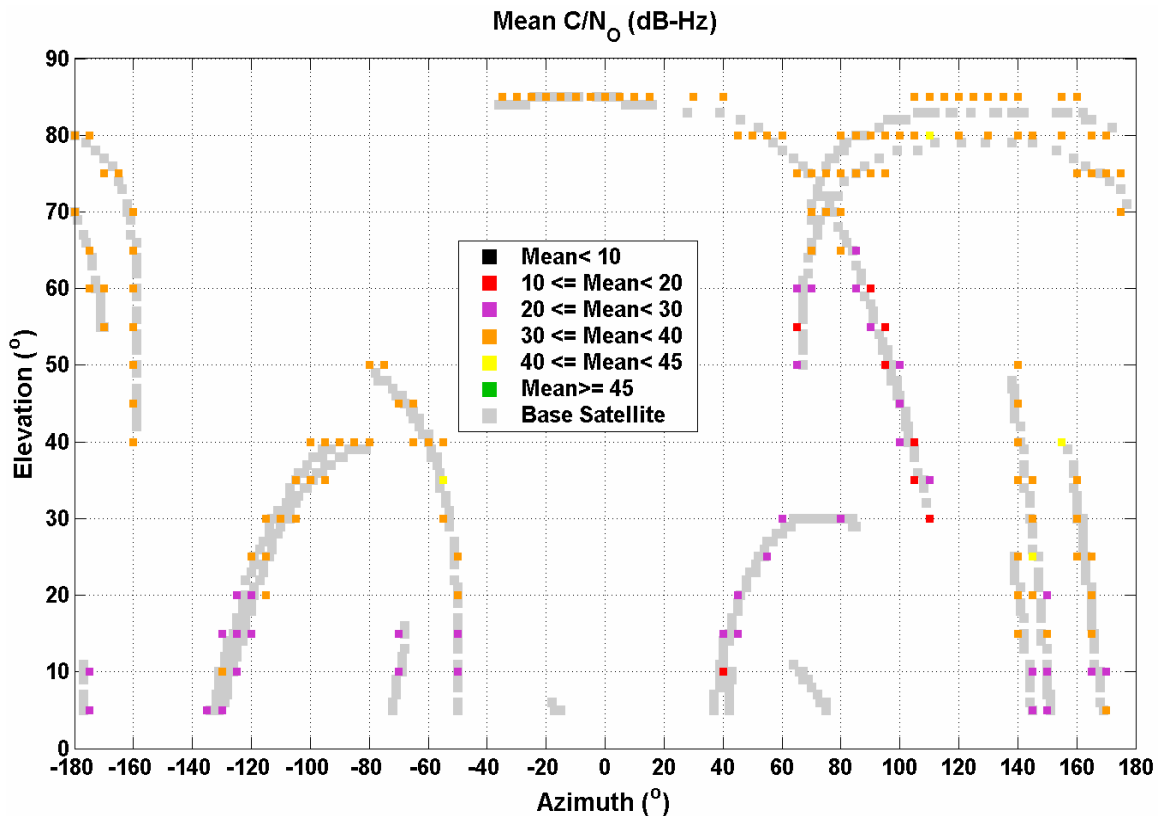


Figure 5.12: Azimuth/Elevation Profile of Average C/N_0 for the Suburban Test

The tracking test in the suburban environment showed that the AGPS and HSGPS receivers had similar performance in terms of horizontal and vertical accuracy, number of satellites used and solution availability. The AGPS used an average of six satellites with 99% availability and using C^3 NAV G^2 had horizontal and vertical accuracies of 43.4 and 50.7 m respectively, while the HSGPS used an average of seven satellites with 99% availability and had horizontal and vertical accuracies of 37.9 m and 55.5 m. The standard receiver used an average of six satellites with 97% availability and had horizontal and vertical accuracies of 31.4 m and 37.0 m. Similar tests by Karunanayake, [2005b] also showed that a standard receiver had better position accuracy than the other two receivers (HSGPS and AGPS). The AGPS and HSGPS receiver carries out longer

integration (80 to 100 ms). The correlation peaks of the reflected signals have similar magnitudes when compared to the direct signal. The receiver tracks these reflected signals which results in large pseudorange errors. The SiRF internal solution for the three receivers had better horizontal and vertical position accuracies compared to the least squares solution. The SiRF internal software performs Kalman filtering to obtain the position solution therefore has better position accuracy.

The C/N_0 and residual plots of the three receivers show similar results. Weaker signals or lower signal power results in larger residual errors which would degrade the user position accuracy. Some high elevation satellites had average C/N_0 values between 40 - 45 dB-Hz (see Figure 5.12), with the majority between 30 – 40 dB-Hz. Lower elevation satellites located on the south side had signal strength between 30 – 40 dB-Hz (signal attenuation due to coniferous trees). In general higher elevation satellites had stronger signals, however signals from lower elevations are attenuated by trees or are echo-only signals (that is lack of satellite visibility due signal blockage from buildings) and therefore have weaker signal strength. Tracking tests conducted in the suburban environment have shown similar results [MacGougan. 2003].

The position accuracy of HSGPS was better than the AGPS receiver. Similar results have been found in the remaining three field test sites. Further analysis illustrated in Table 5.2 showed that results for the three receivers were similar when large position errors were excluded. Larger errors are caused by factors such as multipath or weak signals i.e.

AGPS is more susceptible to multipath and or weaker signals compared to HSGPS. Tracking threshold tests using the simulator with HSGPS and AGPS receivers has shown similar trends in the C/N_0 [Karunanayake, 2005b]. The position accuracy degraded with decreasing simulator power level. The 2D position accuracy was also better for the HSGPS receiver (-150 to -155 dBm) than the AGPS receiver.

Table 5.2: Horizontal Position Results using 67% and 95% of the Best Results for the Three Receivers in the Suburban environment

Receivers	All the Data		95% of the Best		67% of the Best	
	RMS (m)		Results RMS (m)		Results RMS (m)	
	LSQ	SiRF	LSQ	SiRF	LSQ	SiRF
AGPS	43.4	24.6	40.1	23.4	23.0	18.8
HSGPS	37.9	23.8	36.7	22.1	21.8	20.9
Standard	31.4	27.5	29.8	21.2	21.0	17.1

For the remaining three tests, the AGPS receiver is used to obtain the C/N_0 and residual plots. The close proximity of the test site to glass buildings made the receivers vulnerable to multipath effects. Further residual and C/N_0 analyses were carried out to determine the effects of multipath. The location of PRNs 26 and 29 (west side) perhaps caused some reflected signals from the tall glass building located on the east side. After PRN 26 was rejected, the AGPS receiver had a horizontal accuracy of 35.4 m compared to 43.5 m

without any satellite rejection. Similar results were observed when PRN 29 was rejected. However, when PRN 26 or PRN 29 was rejected for the standard receiver, the horizontal position accuracy did not improve suggesting that the AGPS receiver may have been tracking multipath signals for satellites 26 and 29. AGPS and HSGPS, unlike standard receivers, have a different architecture enabling them to track multipath signals.

In a relatively benign environment such as the suburban test site here, AGPS and HSGPS offer no advantage compared to a standard receiver. In fact, an AGPS receiver may be disadvantageous due to its ability to receive and process weak signals created by multipath.

5.4 Residential Garage

Field tests in a residential garage were carried out using the AGPS and HSGPS receivers. The standard receiver was unable to track inside the garage. Figure 5.13 shows the azimuth/elevation of satellites in view, while the position results for the two receivers using C³NAV² and the SiRF internal solution are shown in Figure 5.14 and Figure 5.15. The residual errors and C/N₀ for the AGPS receiver are shown in Figure 5.16 and Figure 5.17. Figure 5.18 shows the mean C/N₀ at different azimuth and elevation for satellites used.

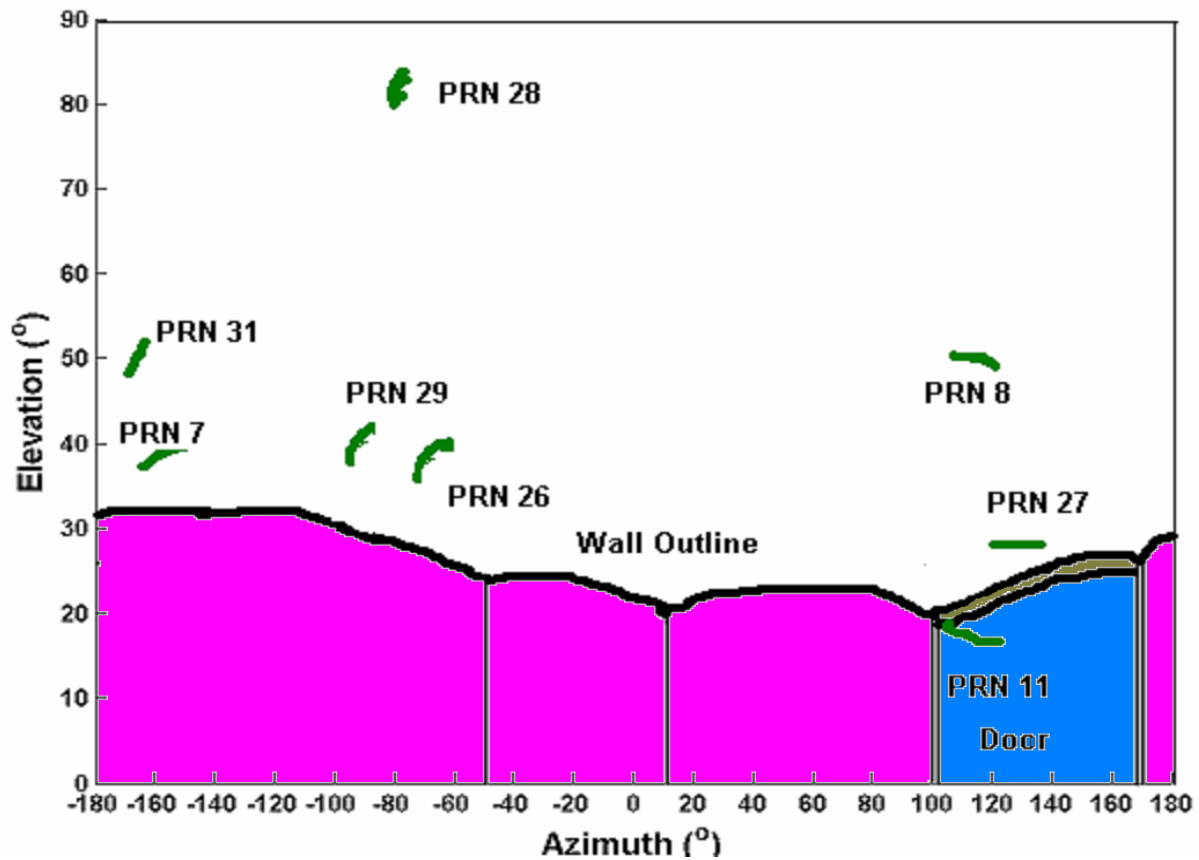


Figure 5.13: Azimuth and Elevation for the Satellites in the Residential Garage Test

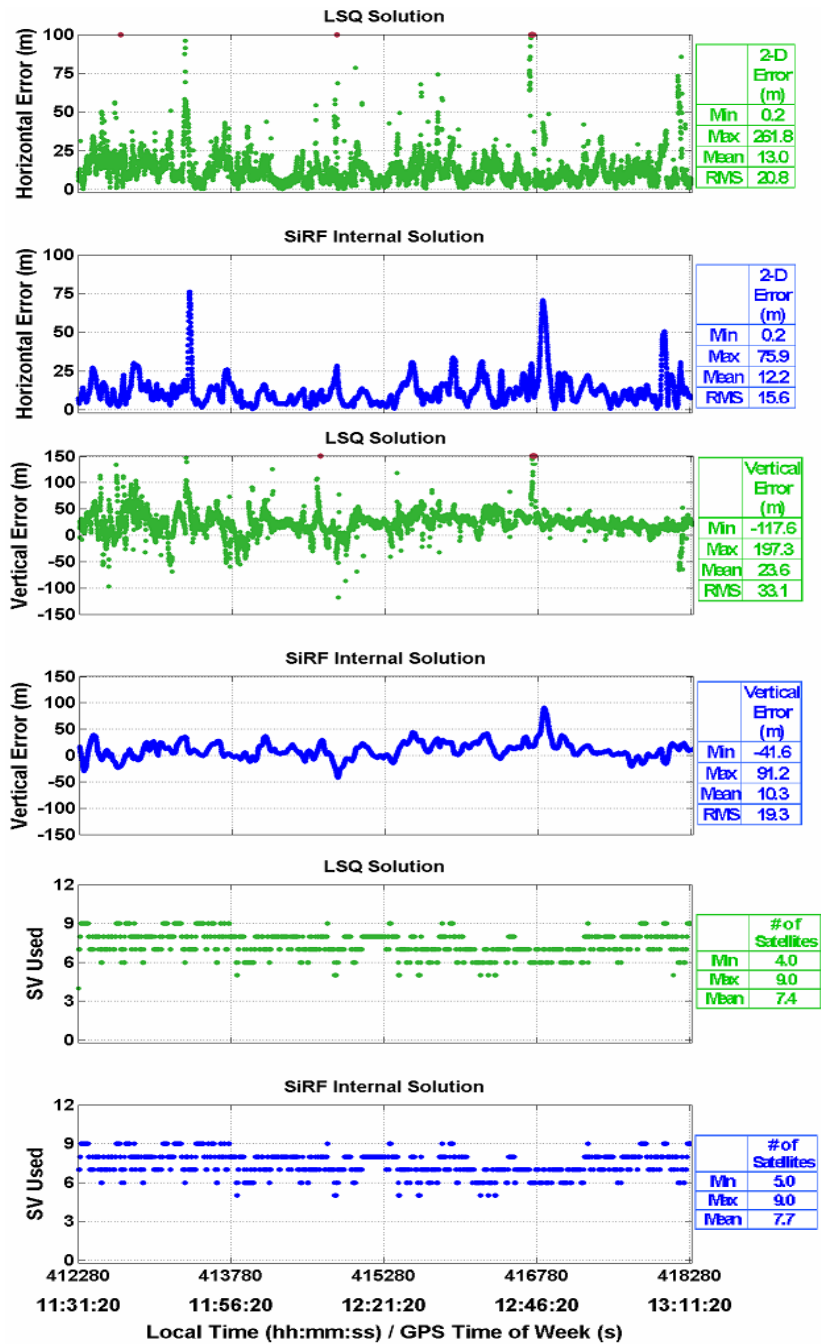


Figure 5.14: AGPS Receiver Position Results for the Residential Garage Test

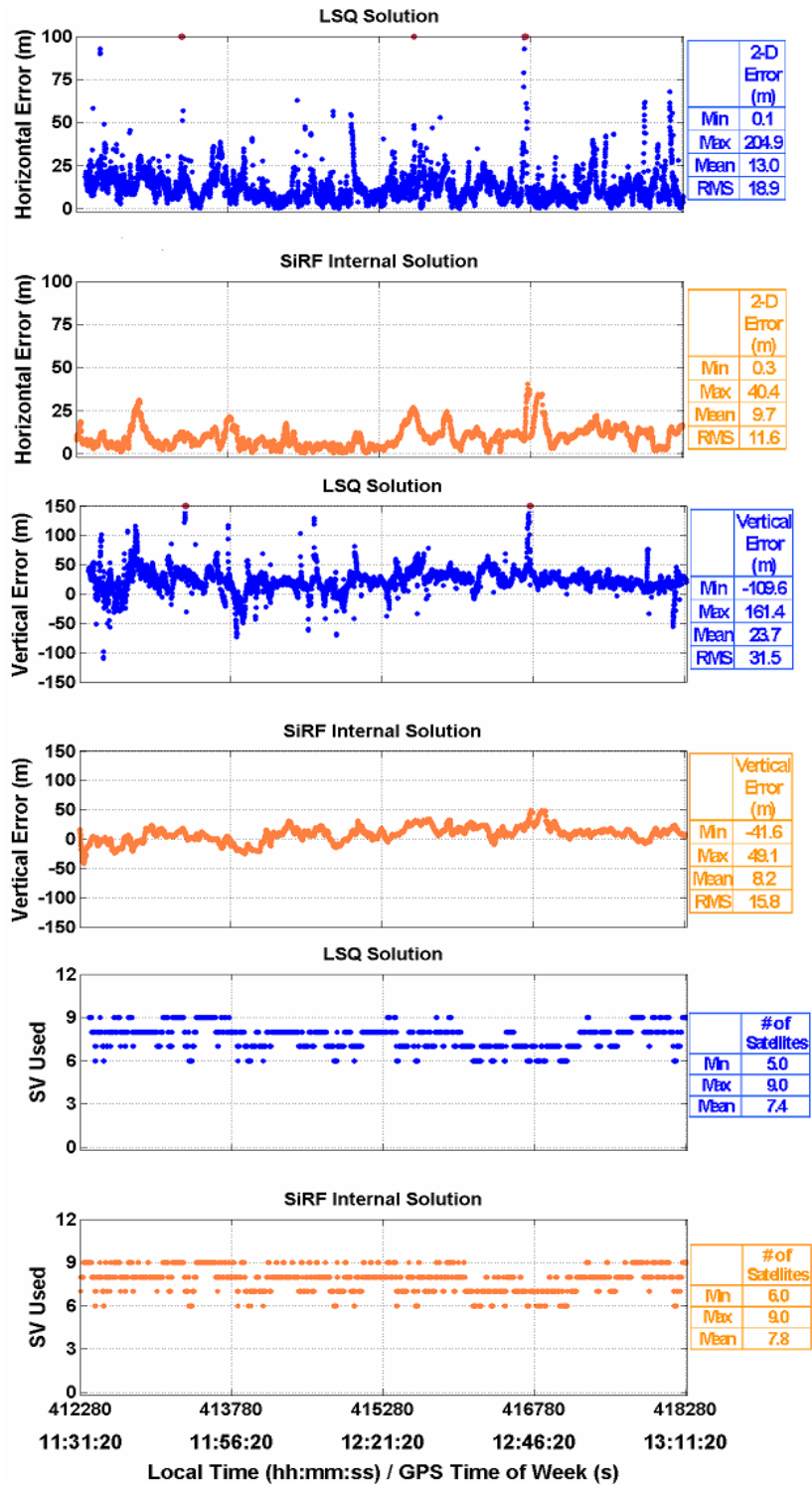


Figure 5.15: HSGPS Receiver Position Results for the Residential Garage Test

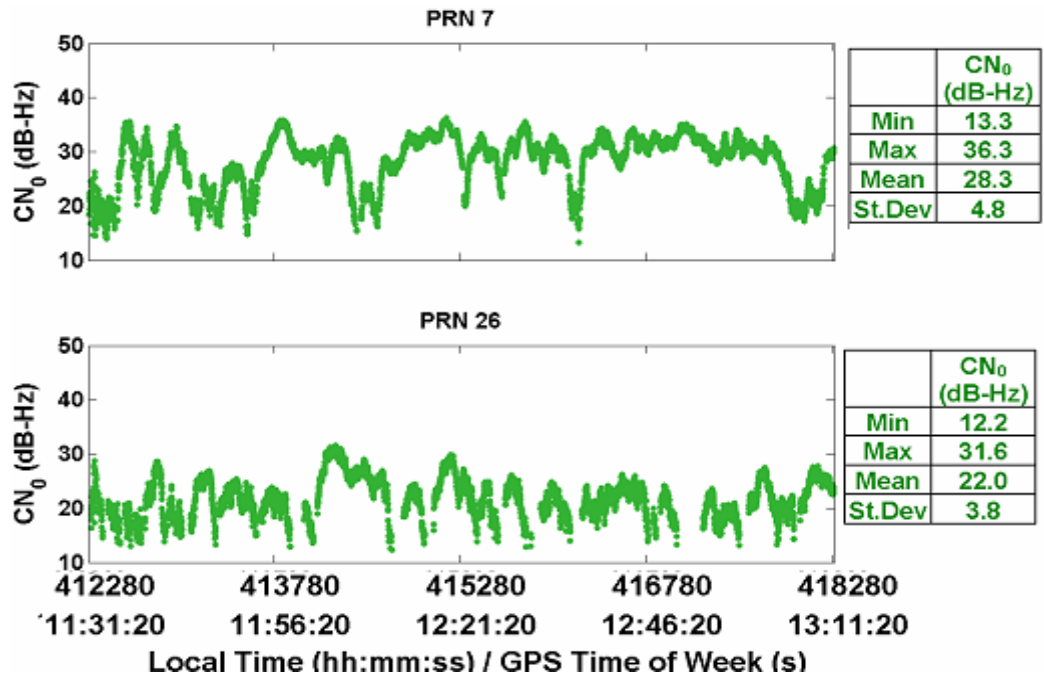


Figure 5.16: Time Series for the C/N₀ for AGPS Receiver for the Residential Garage Test

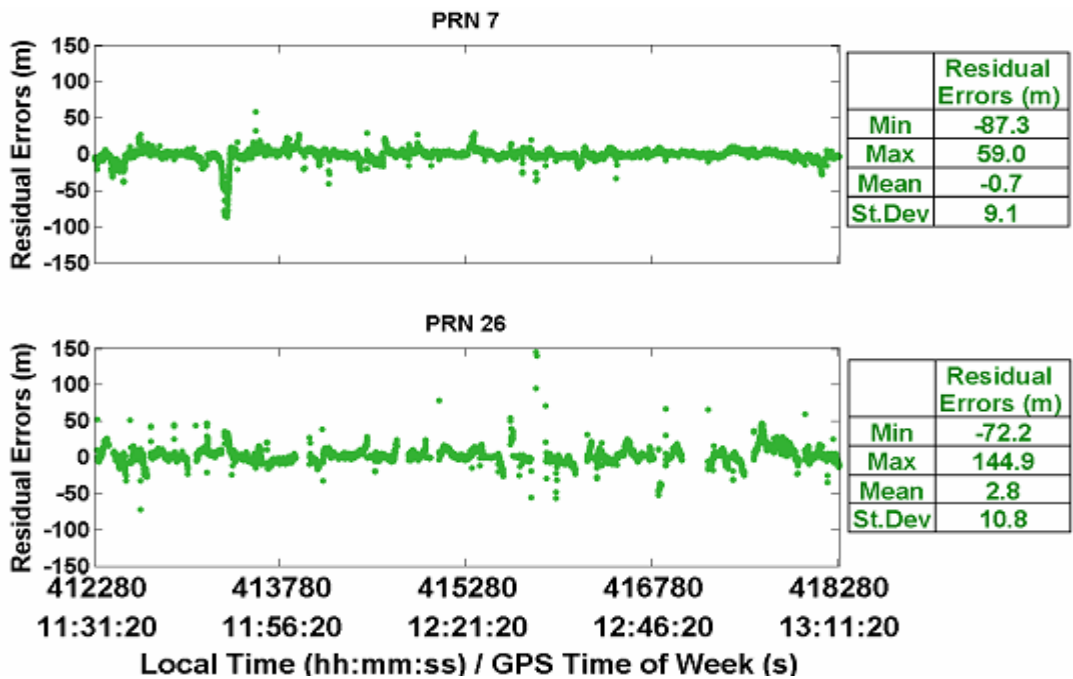


Figure 5.17: Time Series of Residual Errors for the AGPS Receiver for the Residential Garage Test

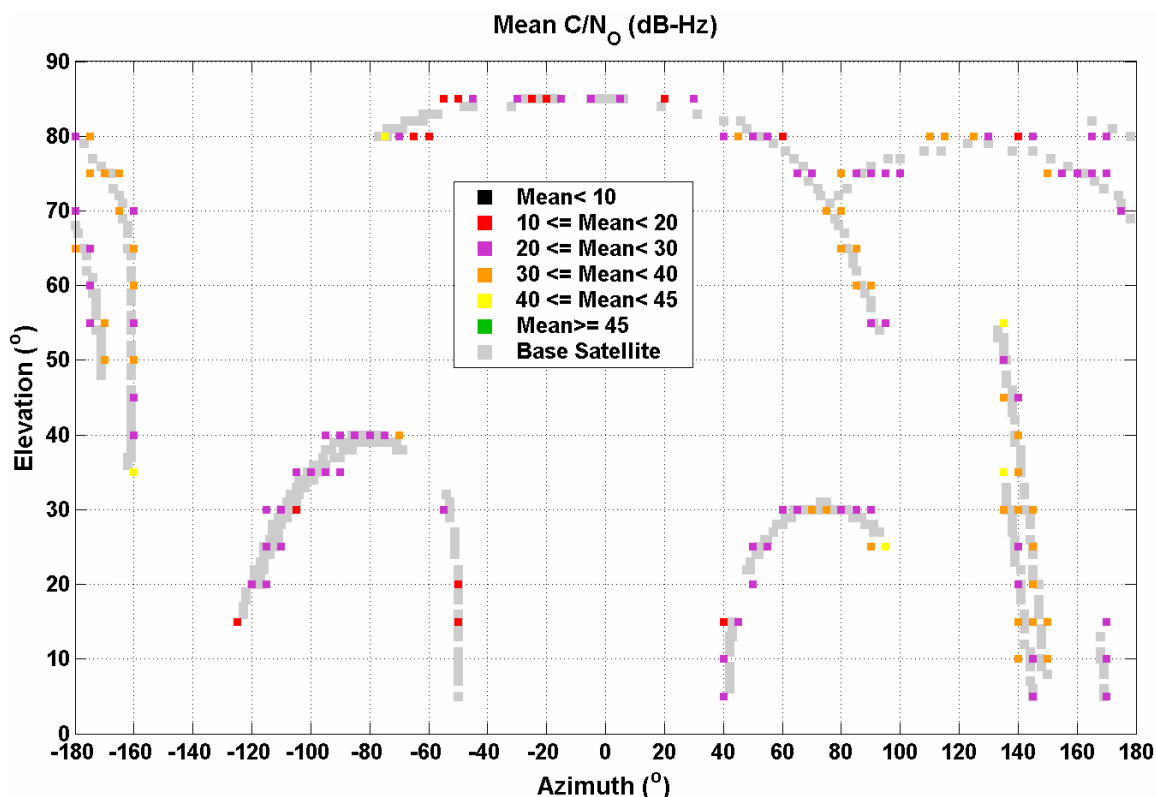


Figure 5.18: Azimuth/Elevation Profile of Average C/N_0 for the Garage Test

The AGPS receiver used an average of seven satellites with 99% availability and for the least squares solution had horizontal and vertical position accuracies of 20.8 m and 33.1 m, while the HSGPS receiver used seven satellites with 98% availability, and had horizontal and vertical accuracies of 18.9 m and 31.5 m. The HSGPS showed better position accuracy using the SiRF solution when compared to the AGPS receiver. The maximum horizontal position error was 40.4 m compared to 75.9 m using the AGPS receiver.

Similar to the suburban test, the two receivers showed better position accuracy (horizontal and vertical) with the SiRF internal solution. For example, AGPS had a

horizontal accuracy of 15.6 m, which was less than 20.8 m that was obtained using C^3 NAVG². The Kalman filtering employed by the internal SiRF software is again the reason.

The garage illustrates varying degrees of signal attenuation. PRN 7 for example, had stronger signals compared to PRN 26 (see Figure 5.16) because its signals entered through the wooden wall on the south side, while the PRN 26 signals entered through concrete walls (front of the wall was made of wood while the rear was constructed of concrete). Figure 5.18 shows that most of the satellites signal strength were between 20 - 30 dB-Hz. Some strong signals entered from the wooden door with mean C/N_0 between 40 – 45 dB-Hz while some very weak signals with mean C/N_0 between 10 – 20 dB-Hz (entered via the concrete walls). The results did not show any correlation with elevation, unlike the sub-urban test where higher elevation satellite had stronger signals. PRN 26 had weaker signals which resulted in larger residual errors (see Figure 5.17) when compared to PRN 7. Weaker signals would result in larger code tracking errors which can be seen in the larger residual errors. Similar tests carried out by Hu, [2006] showed larger pseudorange errors with highly attenuated signals (mean of greater than 15 m).

The residential garage suffered varying degrees of attenuation coming from various different sources such as the concrete and wooden material. The diffuse reflective sources are not as severe which is shown by the good position accuracy for the two receivers (less than 20 m). The standard receiver was unable to track in the residential garage thus

illustrating the importance of higher sensitivity in weaker field conditions. Simulation tests by MacGougan, [2003] had shown that HSGPS has better sensitivity than the standard receiver (16 dB). Field tests using the HSGPS by Hu, [2006] showed similar results; pseudorange error was generally less than 5 m which showed that the effect of multipath was not severe.

5.5 Speed-skating Track

Similar to the previous tests that were carried out in the residential garage, the AGPS and HSGPS receivers were used to carry out tracking tests in the speed-skating track. The azimuth/elevation for various satellites are shown in Table 5.3, while position results for the two receivers using C³NAV² and the SiRF internal solution are shown in Figure 5.19 and Figure 5.20. The residual errors and C/N₀ for the AGPS receiver are shown in Figure 5.21 and Figure 5.22. Figure 5.23 shows the mean C/N₀ at different azimuths and elevations for the satellites used during the tracking tests.

Table 5.3: Azimuth/Elevation of Satellites in the Speed-skating Track

PRN	Azimuth	Elevation
5	-78	6
7	86	79
9	-62	42
11	42	14
26	-125	11
28	-130	6
29	101	43
31	69	70

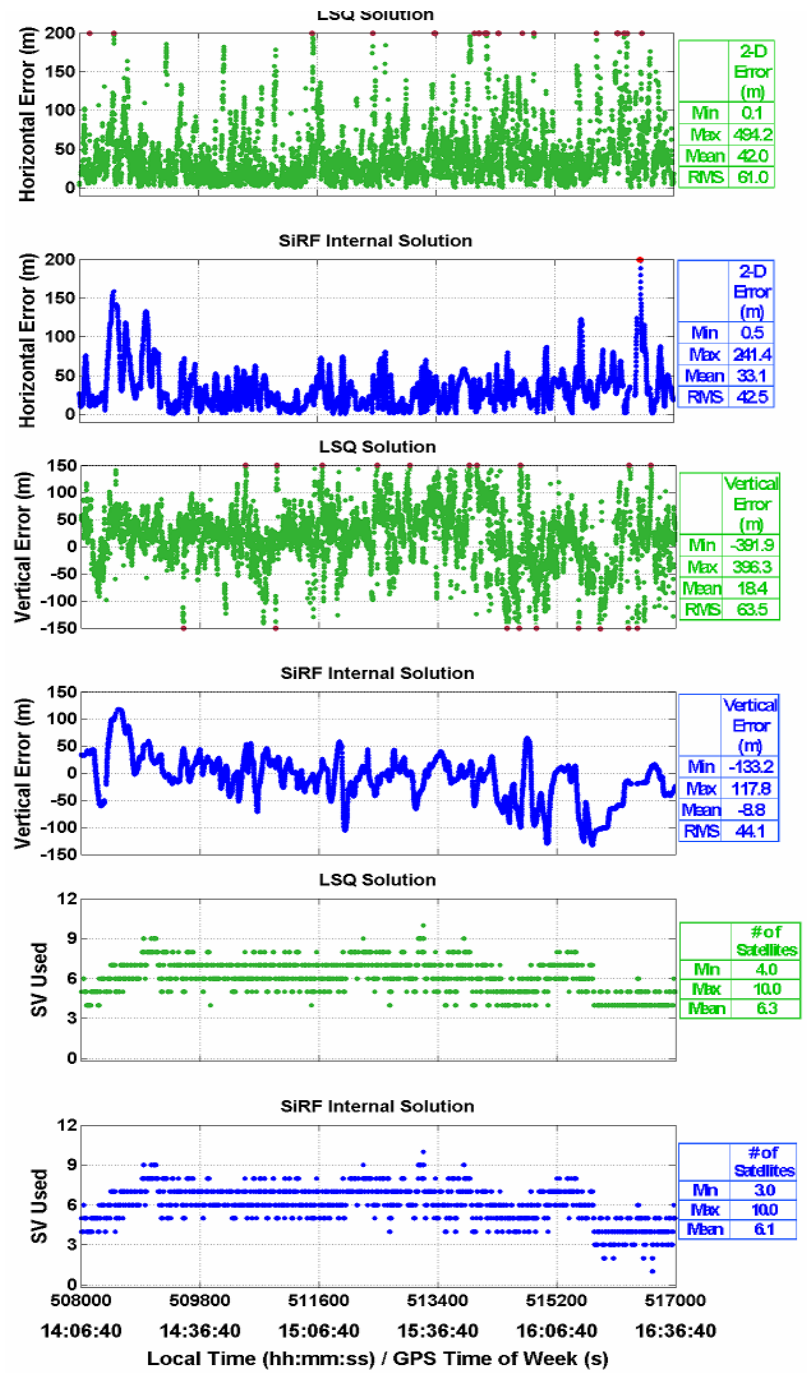


Figure 5.19: AGPS Receiver Position Results for the Speed-skating Track

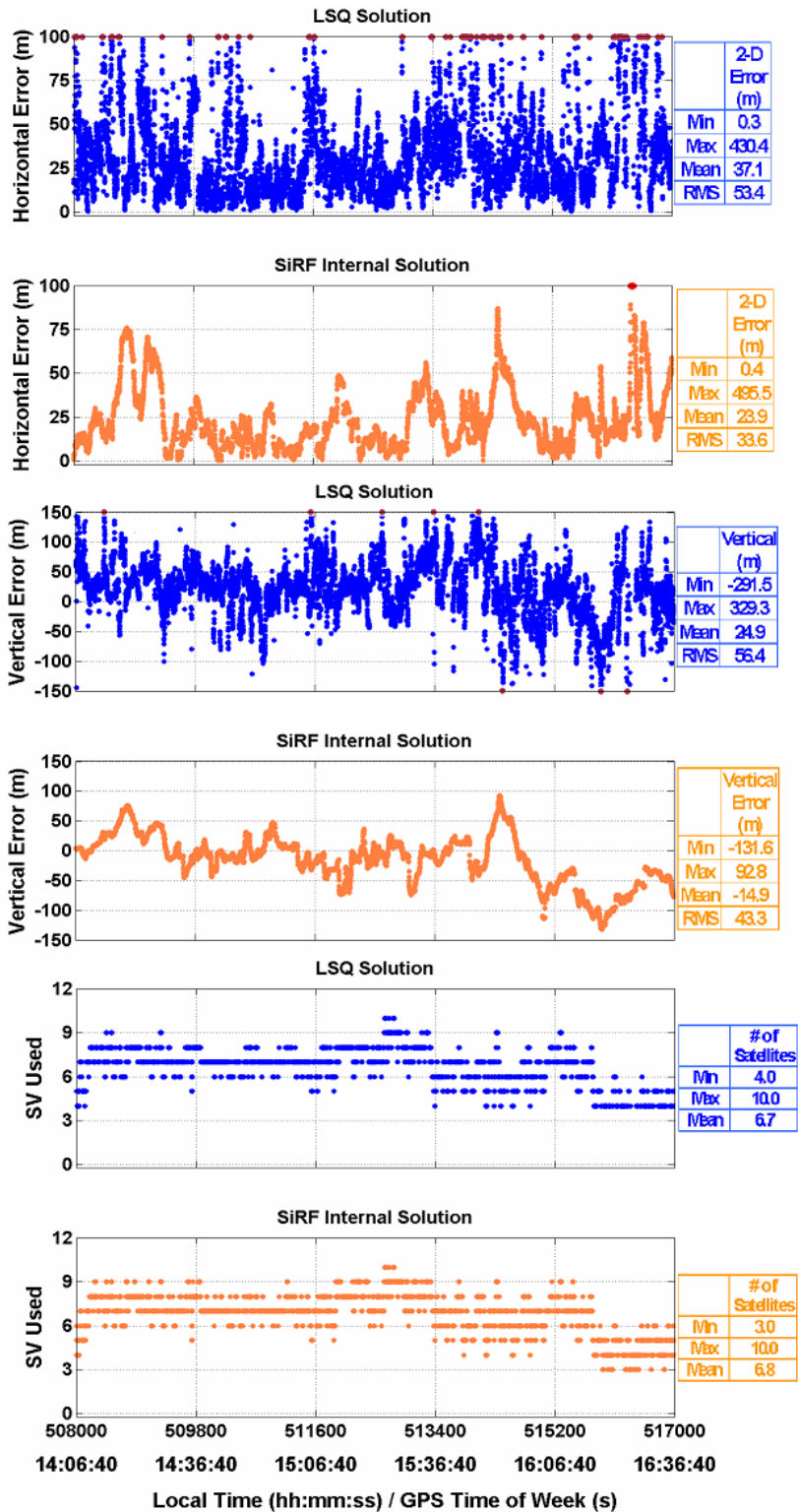


Figure 5.20: HSGPS Receiver Position Results for the Speed-skating Track

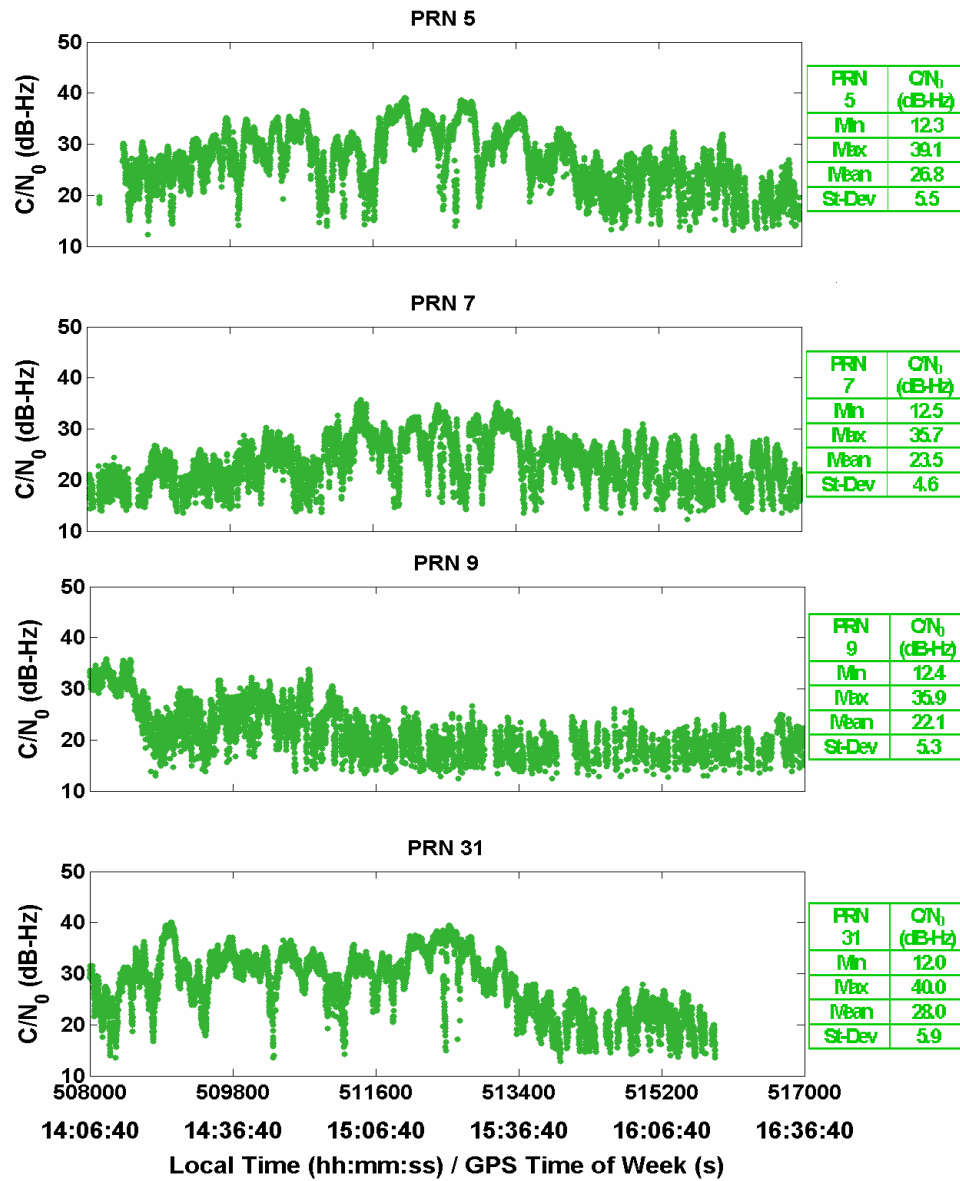


Figure 5.21: Time Series for the C/N_0 for AGPS Receiver for the Speed-skating Track

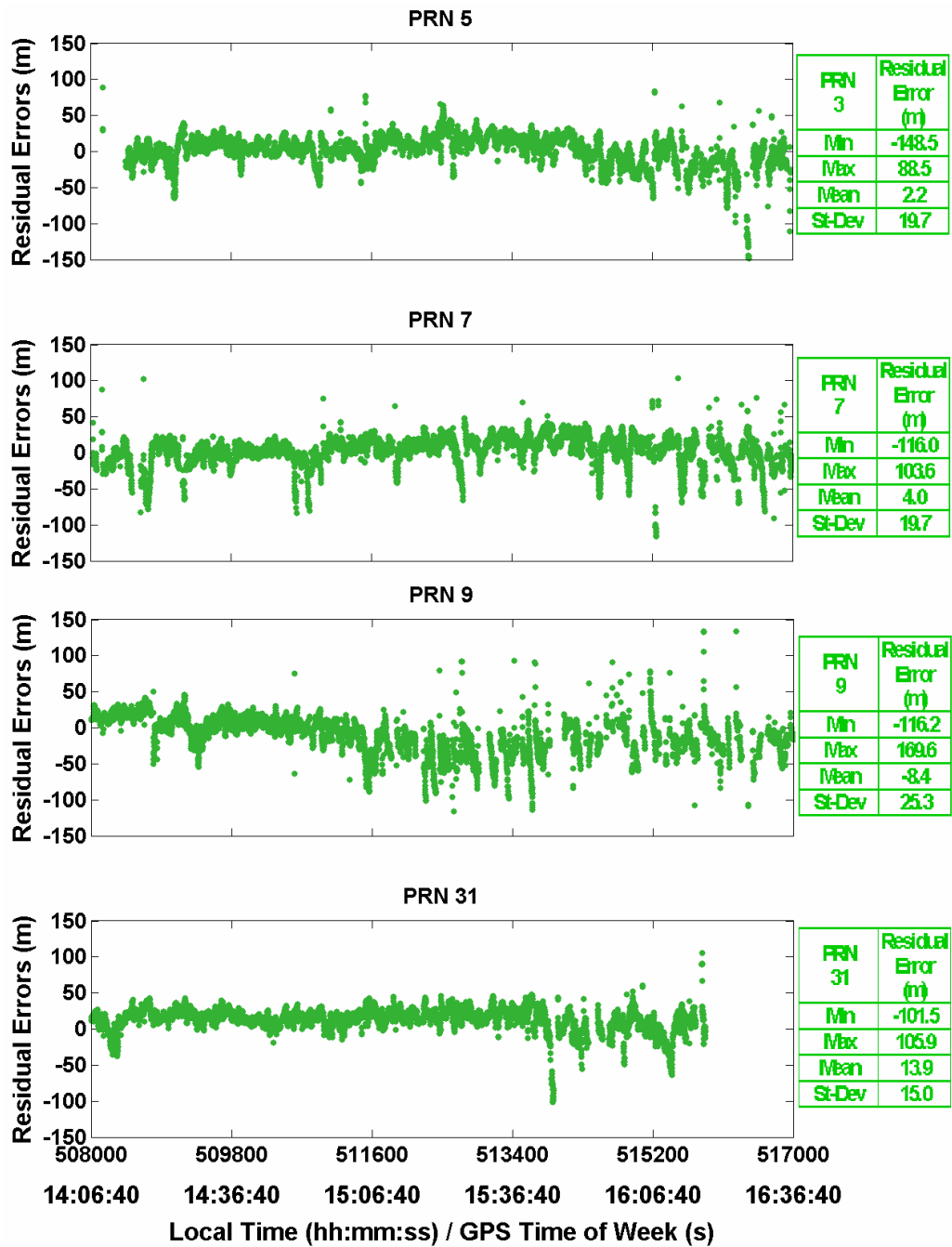


Figure 5.22: Time Series of Residual Errors for the AGPS Receiver for the Speed-skating Track

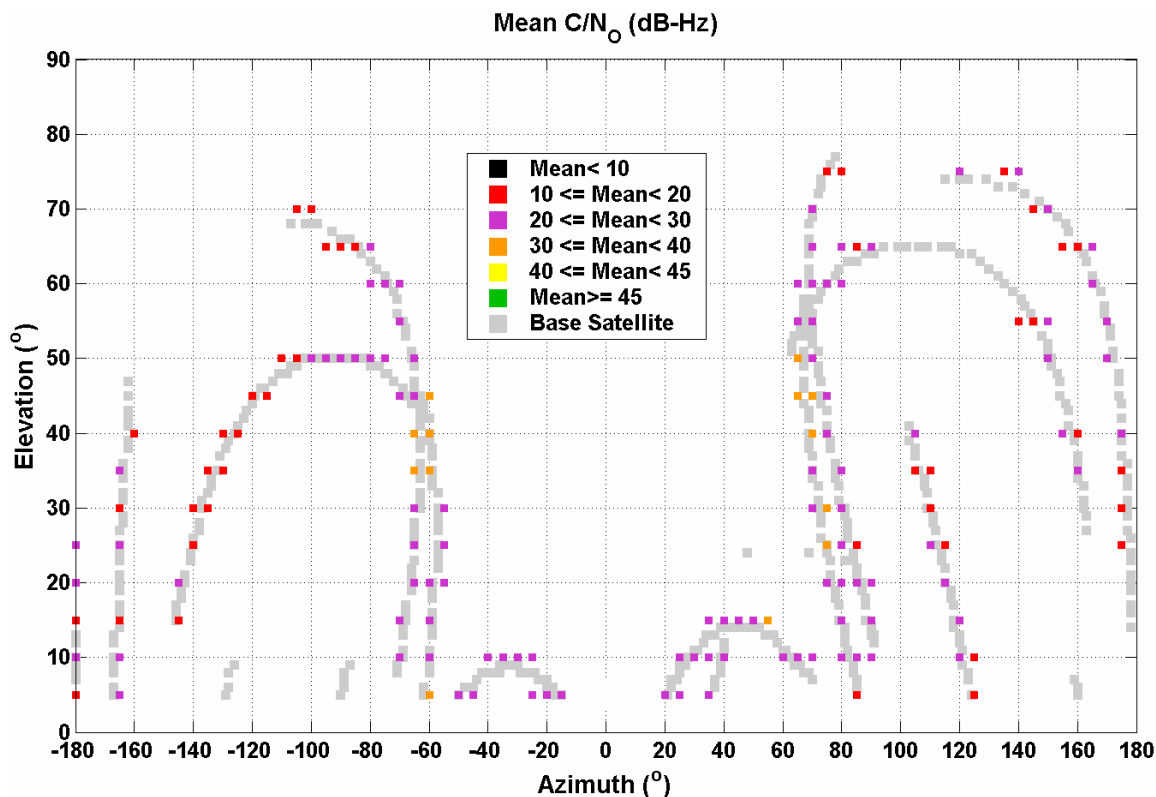


Figure 5.23: Azimuth/Elevation Profile of Average C/N_0 for the Speed-skating Test

The two receivers, AGPS and HSGPS, were able to track signals inside the speed-skating track with similar position accuracy. The AGPS receiver used an average of six satellites with 94% availability and horizontal and vertical position accuracies of 61.0 m and 63.5 m, while the HSGPS used six satellites with 97% availability, and horizontal and vertical accuracies of 53.4 m and 56.4 m. Similar to the previous two field tests, the SiRF internal solution had better position accuracy when compared to the LSQ solution. For example, the AGPS solution had a horizontal position accuracy of 42.5 m, which is better than the $C^3\text{NAV}G^2$ accuracy of 61.0 m.

The tracking test in the speed-skating track was carried out near a window. As a result, signals may have reflected off the roof or attenuated signals could have entered through concrete surfaces. This is illustrated in Figure 5.21 which shows two stronger satellites, PRNs 5 and 31, and two more highly attenuated satellites, PRNs 7 and 9. The satellites generally had weak signals between 20 – 30 dB-Hz, however there were some stronger signals (30 – 40 dB-Hz) entering from the window either from the East or West side and satellites from the North or South side had highly attenuated signals (10 – 20 dB-Hz) due to attenuation by the concrete walls or the porcelain roof. The results are further reflected in the residual plot (see Figure 5.22) where weak signals result in larger code tracking errors. At lower C/N_0 values, the tracking loop may be tracking close to the tracking threshold, resulting in larger pseudorange errors. Field tests conducted by Hu, [2006] showed larger pseudorange errors (beyond 50 m) when compared to the residential garage (less than 5m).

The speed-skating track has bigger dimensions compared to the residential garage, and specular reflective sources would result in larger errors due to multipath. These materials would also highly attenuate the GPS signals thus further degrading the position accuracy. The AGPS and HSGPS had position accuracy of 60 m better than the requirement of 150 m by FCC-E911 or 100 m which can be obtained by cellular positioning methods such as E-OTD.

5.6 Concrete Basement Test

Figure 5.24 shows the azimuth/elevation of the satellites tracked in the concrete basement while the position results for the AGPS and HSGPS receivers using C³NAV² and the SiRF internal solution are shown in Figure 5.25 and Figure 5.26. The residual errors and C/N₀ obtained using the AGPS receiver is shown in Figure 5.27 and Figure 5.28. Figure 5.29 shows the average C/N₀ at different azimuth and elevation for satellites used during the tracking test.

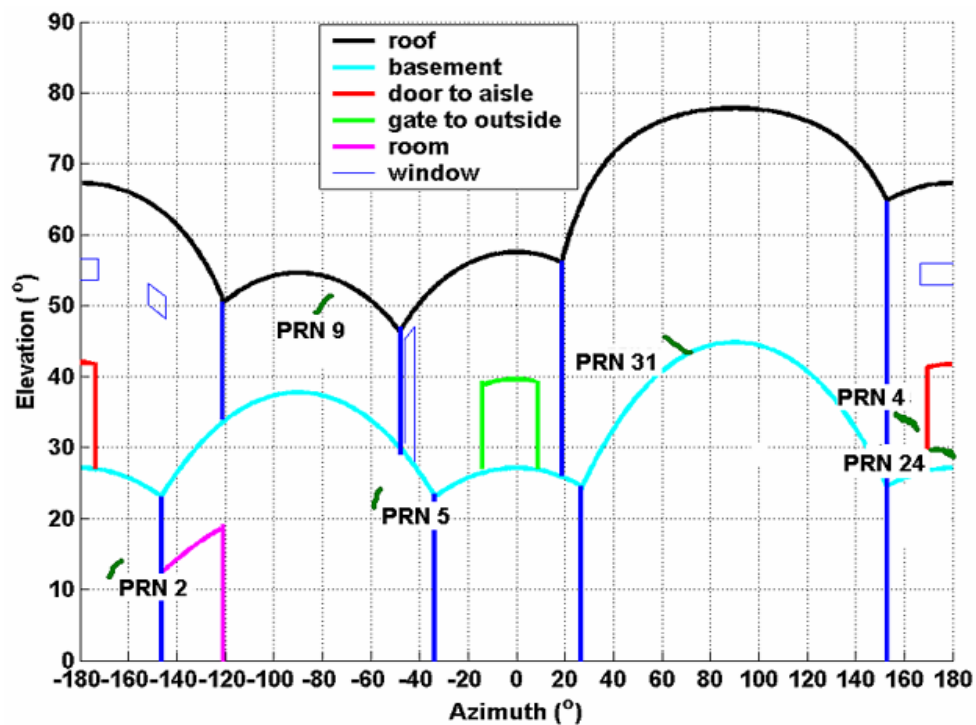


Figure 5.24: Azimuth Elevation for the Satellites in the Concrete Basement Test

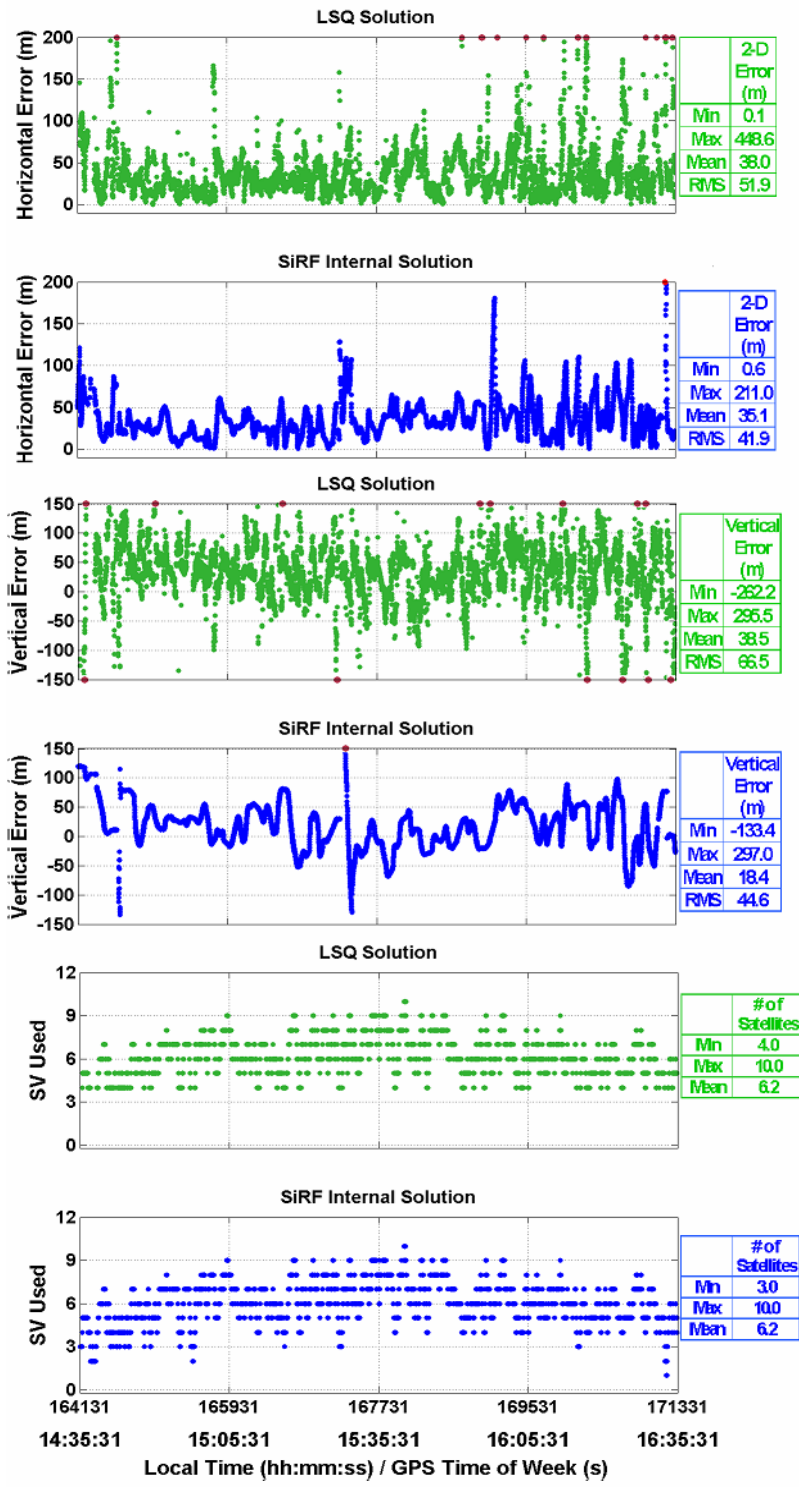


Figure 5.25: AGPS Receiver Position Results for the Concrete Basement Test

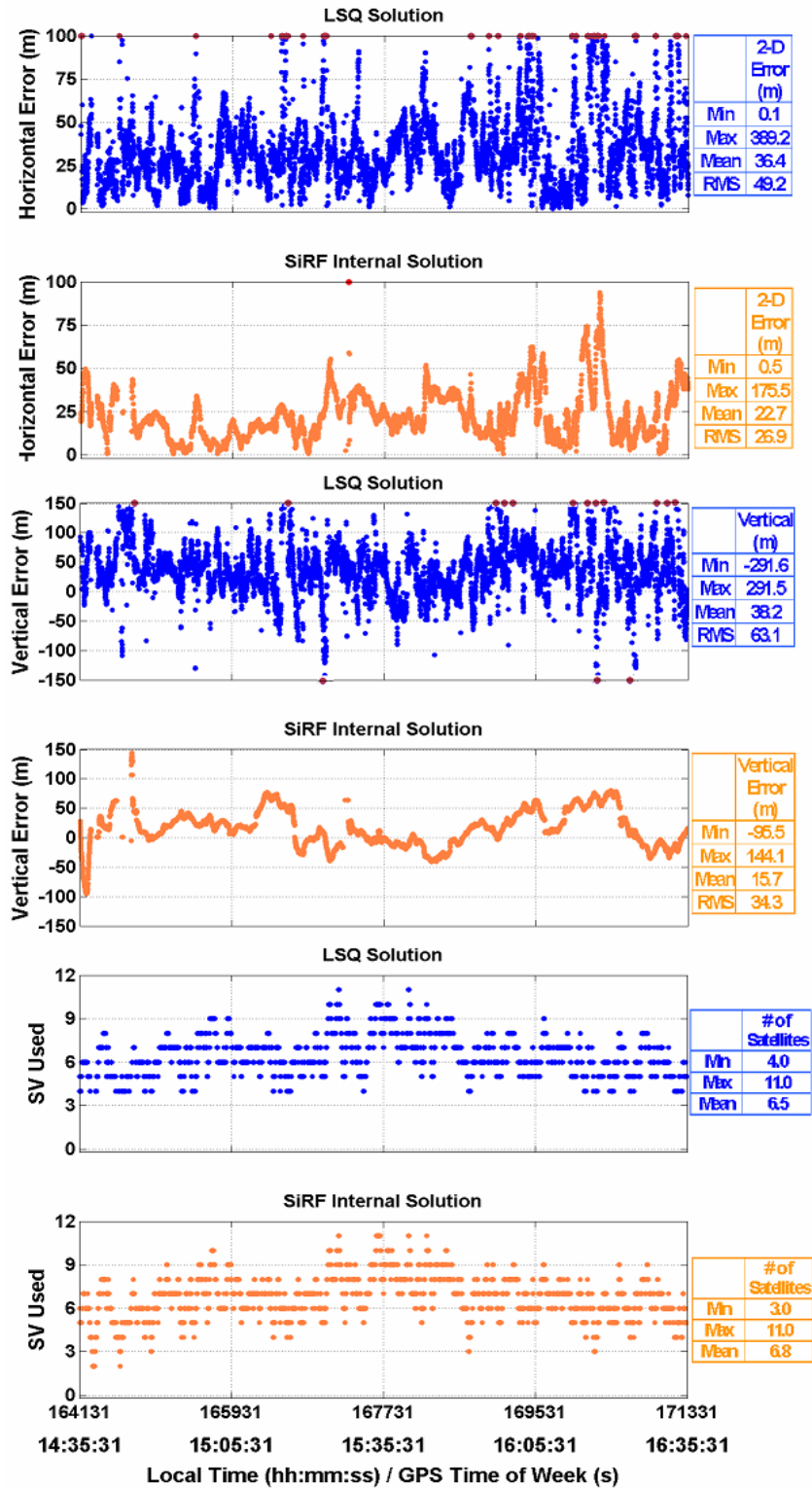


Figure 5.26: HSGPS Receiver Position Results for the Concrete Basement Test

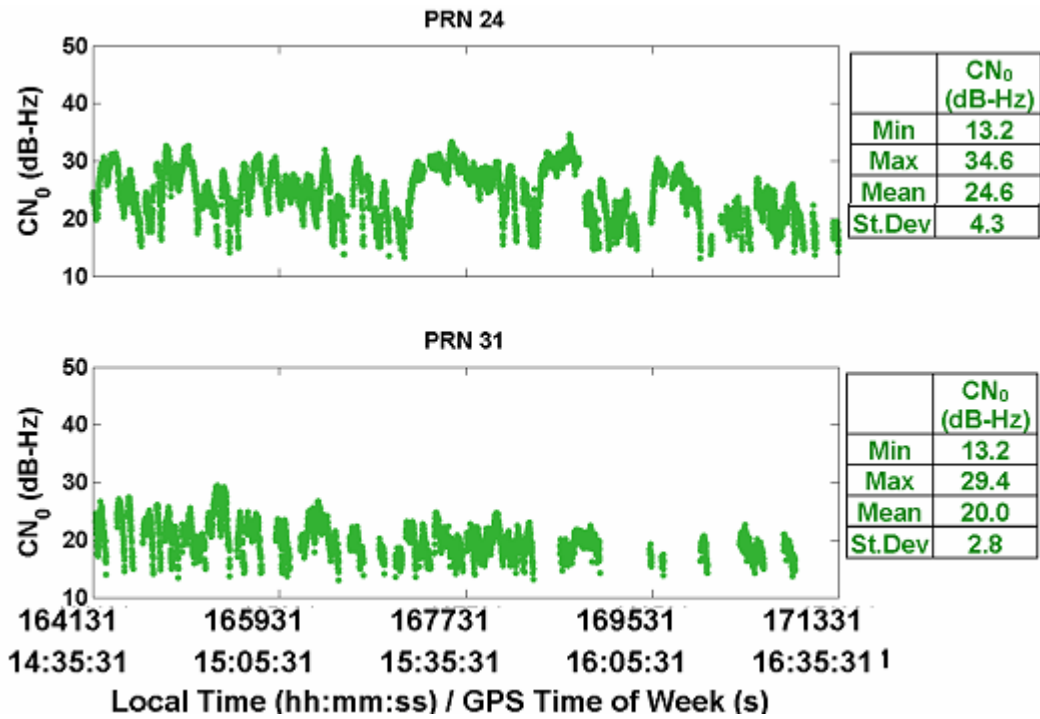


Figure 5.27: Time Series for the C/N₀ for AGPS Receiver for the Concrete Basement Test

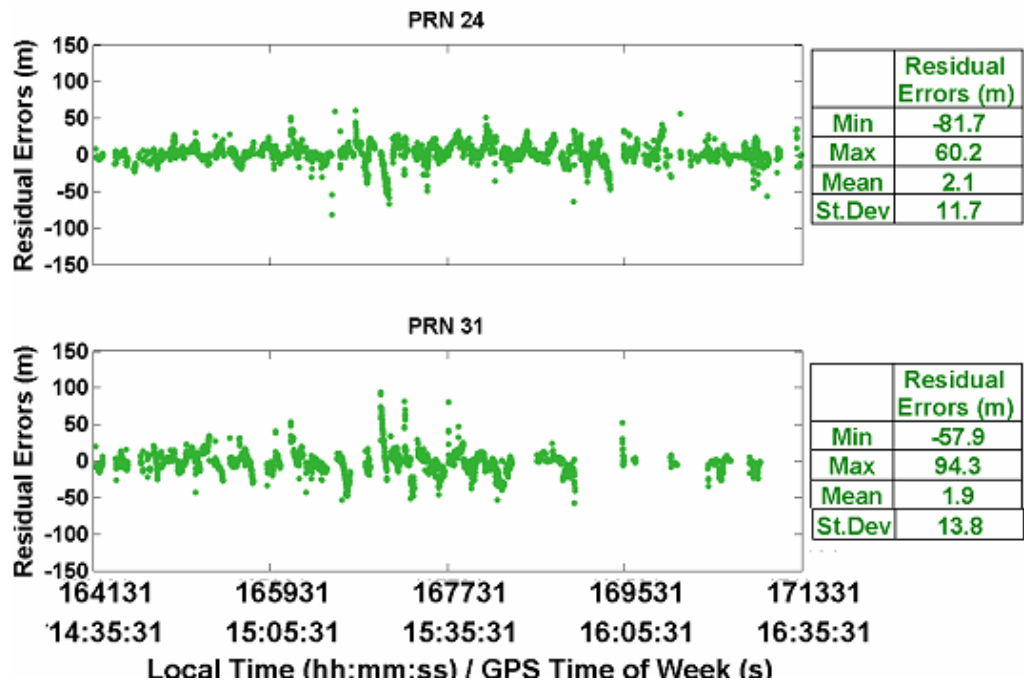


Figure 5.28: Time Series of Residual Errors for the AGPS Receiver for the Concrete Basement Test

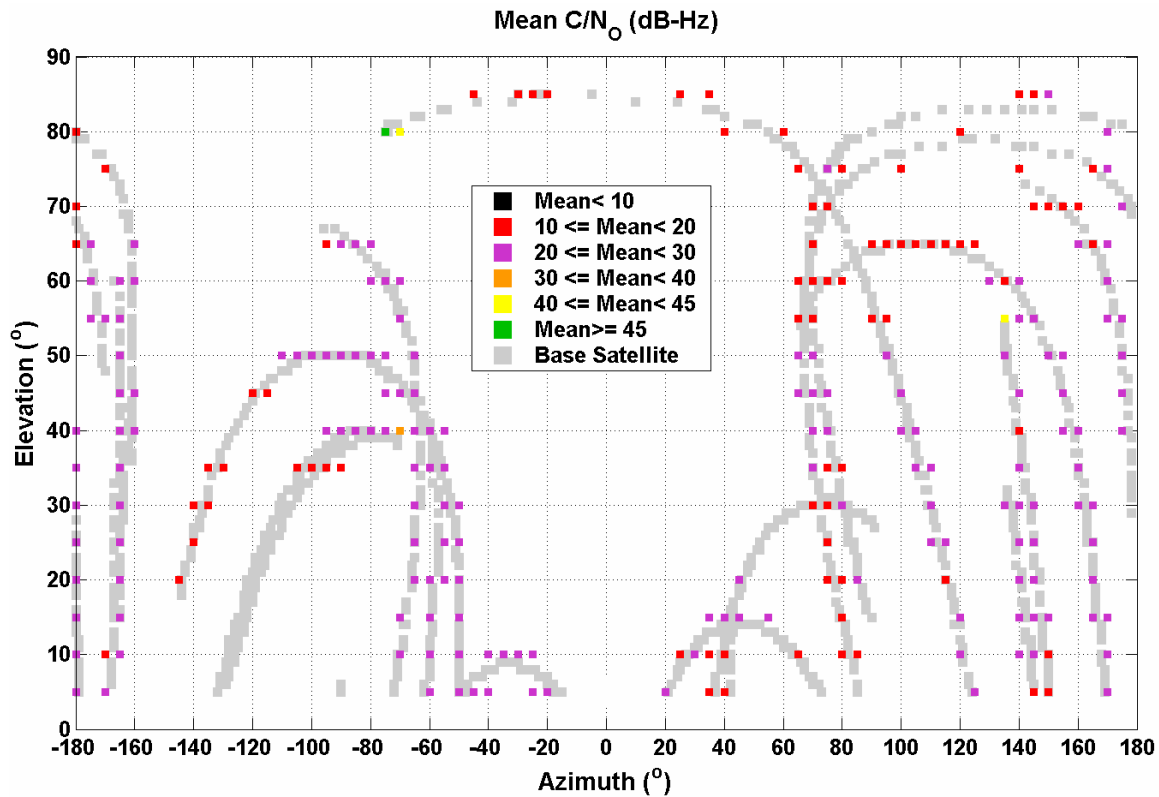


Figure 5.29: Azimuth/Elevation Profile of Average C/N_0 for the Concrete Basement Test

The AGPS and HSGPS receivers gave similar results in terms of position accuracy, number of satellites used and availability. The AGPS receiver used an average of six satellites, with 95% availability, and yielded horizontal and vertical position accuracies of 51.9 m and 66.5 m, while the HSGPS receiver tracked an average of six satellites, with 97% availability, and yielded horizontal vertical position accuracies of 49.2 m and 63.1 m.

The GPS signals were highly attenuated as shown in the C/N_0 plot, where PRN 24 had a mean C/N_0 of 24.6 dB-Hz and PRN 31 had mean C/N_0 value of 20.0 dB-Hz. The signals in the concrete basement were generally weak anywhere between 10 – 30 dB-Hz. (see

Figure 5.29) The majority of the signals were attenuated by the concrete walls and had signal strengths between 10 - 20 dB-Hz. These results are different when compared to the speed-skating track where some strong signals entered from the glass windows. The receivers were tracking weak signals (in some cases less than 20.0 dB-Hz) that were close to the tracking threshold. Therefore, the receiver would frequently lose lock on the satellites and this resulted in large code tracking errors, as seen in the residual plots.

The SiRF internal solution had better position accuracy when compared to the C³NAV² solution. For example, the AGPS solution had a horizontal position accuracy of 41.9 m, which was smaller than 51.9 m as provided by C³NAV². The SiRF receiver which uses Kalman filtering is able to identify and recover from large position errors, thus obtaining better position accuracy when compared to the least squares solution.

The concrete basement has smaller dimensions than the speed-skating track. The concrete walls highly attenuate the GPS signals (mostly Non Line of Sight signals). The above factors result in a better position accuracy because of smaller multipath delays and reflections from diffuse sources. Similar tests in the field showed that GPS signals were attenuated by as much as 30 dB in the concrete basement [Hu, 2006]. The tests had also shown that the pseudorange errors were worse in the speed-skating track (greater than 50 m) when compared to the concrete basement (less than 50 m).

5.7 Comparison of Different Environments

This section compares the position results from the four test environments. The AGPS position results from C³NAV² and SiRF internal solution for the four environments are shown in Figure 5.30 through Figure 5.33. The results are compared and discussed in terms of factors such as signal blockage, multipath and signal attenuation. The AGPS and HSGPS receivers had similar position accuracy; therefore, only the AGPS results are used to compare the different environments.

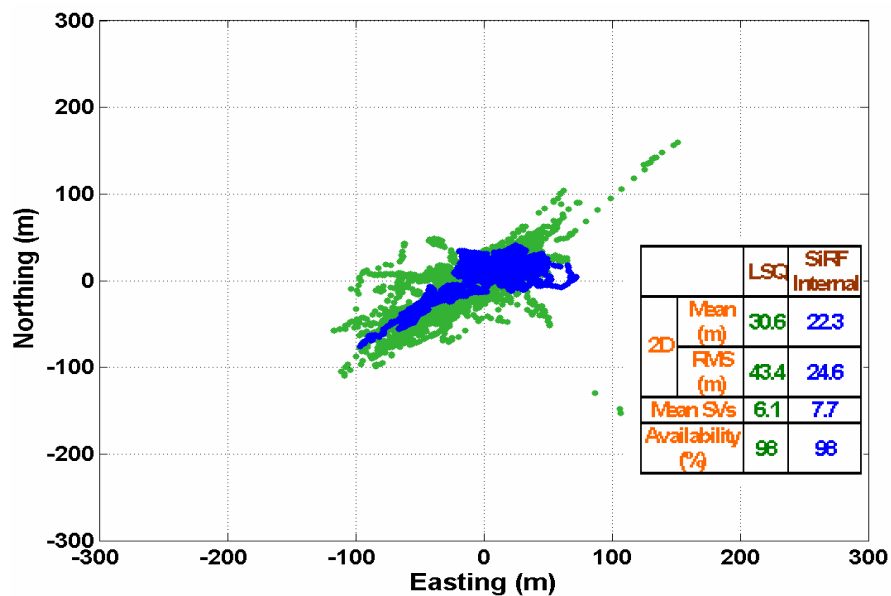


Figure 5.30: Position Solution for Suburban Test using the AGPS Receiver

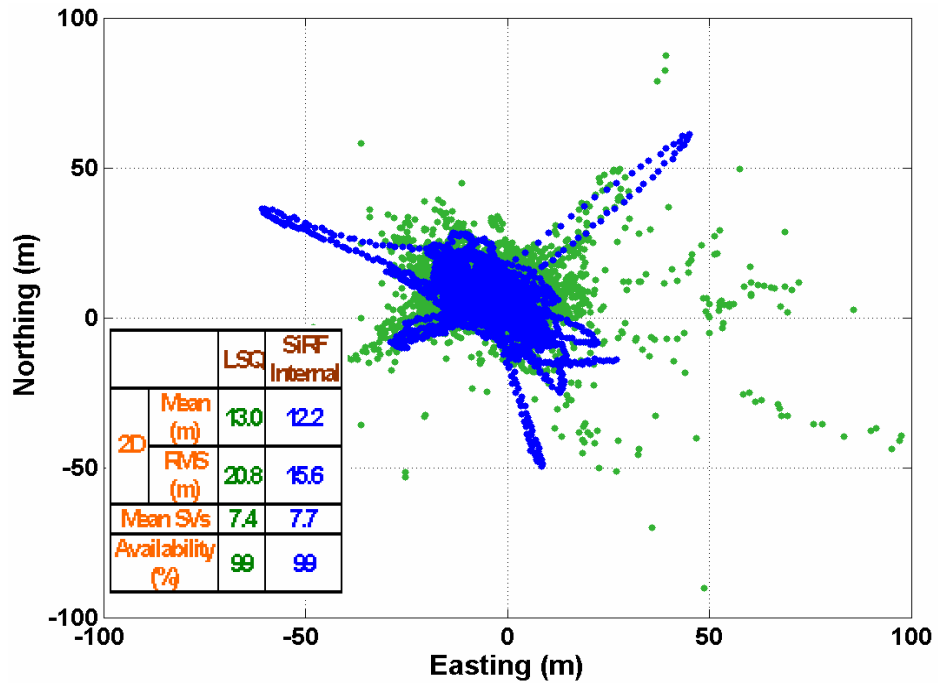


Figure 5.31: Position Solution for Residential Garage Test using the AGPS Receiver

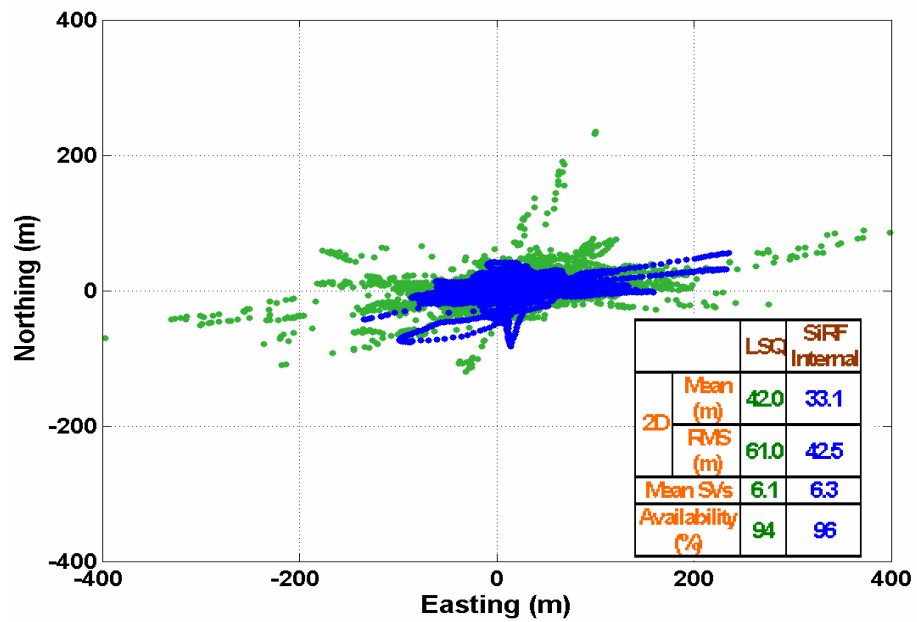


Figure 5.32: Position Solution for Speed-skating Track Test using the AGPS Receiver

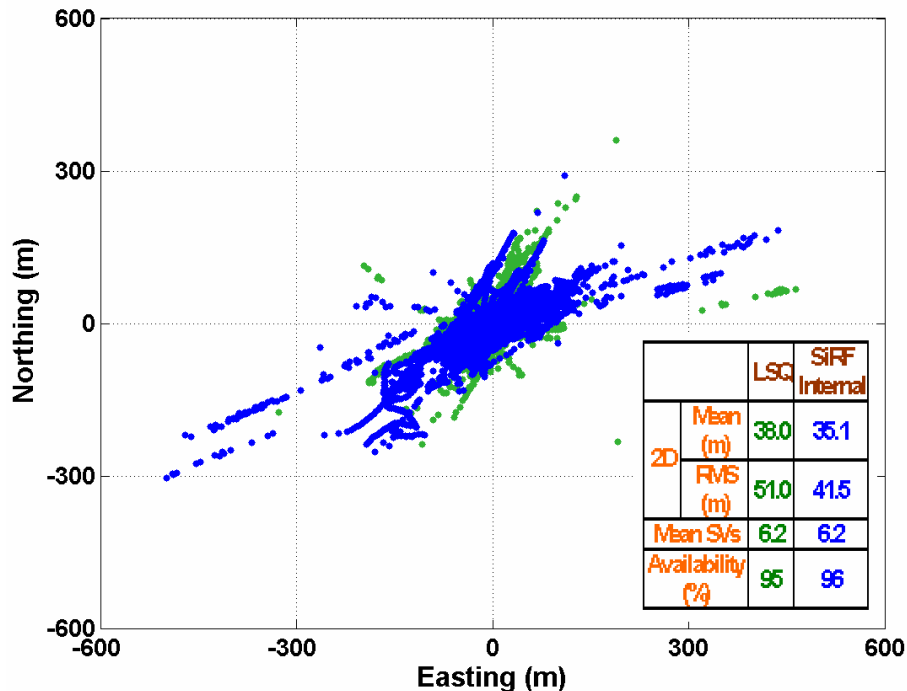


Figure 5.33: Position Solution for Concrete Basement Test using the AGPS Receiver

The field tracking tests in different environments illustrated different challenges that were reflected in the position results, number of satellites tracked, availability and C/N₀. The AGPS receiver had better positioning accuracy in the residential garage test than in the suburban test, although signal strengths were weaker in the garage (greater than 35 dB-Hz in the suburban environment compared to less than 30 dB-Hz in the garage). The suburban test site was located close to a tall glass building that caused strong specular reflections. These reflections cause strong multipath signals which when received introduce pseudorange errors and thus degrade the position accuracy. The residential garage, on the other hand, was subject to diffuse reflections (due to material such as wood or concrete). Diffuse reflections are scattered in many directions and are, therefore,

not as strong as specular reflections. As a result, they may have little, if any effect on the line of sight signal received by the receiver.

The signals inside the speed-skating track and the concrete basement were very weak (20 dB-Hz) and this led to larger code tracking errors, and poorer positioning accuracy when compared to the other two environments. The position accuracy was worse for the speed-skating track because of specular reflections from structures such as the corrugate roof and longer multipath delays due to the larger internal dimensions of the track compared to the concrete basement. The LSQ had larger position errors than the SiRF solution for all the four environments and the reason is Kalman filtering.

5.8 Chapter Summary

Tracking tests were carried out in different environments to investigate the performance of AGPS, HSGPS and standard receivers. The tests conducted have led to the following conclusions.

- The tests in indoor environments (for e.g. the residential garage) required the use of HSGPS or AGPS because the standard receiver was not able to track in these types of low signal strength conditions
- Similar position accuracy results for the HSGPS and AGPS receivers in the four environments suggest that aiding data does not enhance the tracking performance. Aiding data is used to shorten the acquisition process. The Doppler and C/A code

estimates from aiding data is coarse information for the code and carrier tracking loops and thus not useful to them. The HSGPS generally had slightly better accuracy because it was not as susceptible to multipath effects or highly attenuated signals.

- The HSGPS and AGPS receivers had very good solution availability (greater than 95%) and very good horizontal position accuracy (less than 50 m using the SiRF internal solution and less than 70 m using the least squares solution). This is within the maximum requirements of FCC-E911 mandate (150 m for 95% of the time), and is better than E-OTD position method (greater than 100 m).

CHAPTER 6: CONCLUSIONS & RECOMMENDATIONS

Wireless devices should be able to work anywhere all the time. The FCC-E911 mandate and LBS have been driving forces for accurate wireless positioning with shorter TTFF. Acquisition tests in four different environments were carried out to illustrate the ability of AGPS to obtain an accurate position) solution (better than 50 m for most of the tests with short TTFF (< 30 s) in a typical residential environment such as the residential garage.

6.1 Conclusions

The primary objective of this work was to investigate and compare the acquisition and tracking performance of AGPS and HSGPS receivers. First, numerous simulation tests were carried out and the results analyzed in terms of factors such as TTFF, signal power and position accuracy. Furthermore results obtained from the acquisition tests carried out in the field confirmed earlier findings from simulation tests (Chapter Three) and Karunanayake, [2005b]. Finally tracking tests were carried out in to compare the two receivers. The conclusions are discussed below.

- Simulations tests showed that AGPS higher acquisition sensitivity as compared to the other two receivers, 13 dB better than HSGPS and 20 dB better than the standard receiver making it suitable for many applications where GPS signals can be as low as -150 dBm.
- Simulations tests also showed that satellite ephemeris data are important for signal acquisition in terms of sensitivity (11 dB better with ephemeris aiding) and

- a shorter TTFF (ten times in the sub-urban environment), while almanac data are not required for signal acquisition.
- The acquisition field tests showed the limitations of the HSGPS receiver: its inability to acquire in weak signal conditions and long TTFF (ten times longer) when compared to the AGPS receiver.
 - The acquisition tests in the field showed that lack of satellite ephemeris resulted in longer TTFF, while that the almanac is not required for signal acquisition. The tests confirmed the results that were obtained from simulation tests in Chapter 3.
 - Increased timing uncertainty (precise time aiding) results in longer TTFF under weak field signal conditions. The tests also showed that sub-millisecond level accuracy for the time aiding is required. In other words, coarse time aiding has no effect on TTFF.
 - The tests under different field test conditions showed the increasing level of difficulty in acquiring satellite signals to obtain a position fix, especially in environments such as the speed-skating track and the concrete basement. These two environments had highly attenuated signals (20 to 25 dB-Hz). The position accuracy was better than 50 m for the first two environments (Sub-urban and Residential Garage) but was worse for the last two environments up to 70 m.
 - The tests in the residential garage required the usage of HSGPS or AGPS because standard receiver was not able to track in these types of challenging conditions.
 - Similar position accuracy results for the HSGPS and AGPS receivers in the four environments suggest that aiding data does not enhance the tracking performance. In fact aiding data is used to shorten the acquisition process. The HSGPS

- generally had slightly better accuracy because it was not as susceptible to multipath effects or highly attenuated signals.
- The HSGPS and AGPS receivers had very good solution availability (greater than 95%) and very good horizontal position accuracy (less than 50 m using the SiRF internal solution and less than 70 m using the least squares solution). This is within the maximum requirements of FCC-E911 mandate (150 m for 95% of the time), and is better than E-OTD position method (greater than 100 m).

6.2 Recommendations

Field tests were carried out under static conditions, but further tests are required under kinematic conditions, which could not be carried out with the current setup. Kinematic tests using the hardware simulator have been carried out by Karunanayake et al., [2005].

The simulations tests that were conducted did not account for multipath effects. It is recommended that the effects of multipath on an AGPS receiver be investigated using a simulator. Indoor or outdoor replications using the hardware simulator of the field tests sites have already been carried out using the SiRF HSGPS receivers. [Hu, 2006]. Similar research needs to be carried out using the AGPS receiver. This would be valuable for AGPS product developers where time can be saved by conducting tests with the simulator that could reflect field test conditions.

Wireless technologies are evolving to the Third Generation (3G) network where different networks such as Wireless Local Area Network (WLAN) and cellular networks (CMDA and GSM) will be integrated into one network [Fapojuwo, 2003]. Field tests in extremely weak signal conditions like the concrete basement have shown that AGPS has extremely long TTFF. However, AGPS augmented with WLAN will increase the availability of a position solution in these types of challenging conditions. This in fact is nothing new. The concept is similar to QUALCOMM's use of a hybrid AGPS/CDMA network to obtain position solutions in urban canyon environments.

The future of GNSS looks promising for indoor LBS applications with the arrival of Galileo and GPS modernization such as the L2C signal, which has better cross-correlation properties of 45 dB as compared to the L1 C/A code with a cross-correlation of 21 dB. Galileo will transmit two signals for civilian users. The signals will be transmitted at higher power (5 dB) than GPS signals and one of the signals will be dataless facilitating longer coherent integration. The augmentation of GPS/Galileo will increase satellite availability, which is particularly attractive in challenging areas such as indoors or urban canyon environments. Heinrichs et al, [2006] discuss the possibility of GPS/Galileo receivers receiving aiding data from the UMTS network. Further studies are required to investigate the performance of such receivers once they become available.

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